

1 **Inverse design of indoor environment using an adjoint RNG k- ϵ turbulence model**

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10
11 **Abstract**

12 The adjoint method can determine design variables of an indoor environment according to the
13 optimal design objective, such as minimal predicted mean vote (PMV) for thermal comfort.
14 The method calculates the gradient of the objective function over the design variables so that
15 the objective function can be minimized along the fastest direction using an optimization
16 algorithm. Since the objective function is controlled by the Reynolds-averaged Navier-Stokes
17 (RANS) equations with the RNG k- ϵ model during the optimization process, all the
18 corresponding adjoint equations should be solved, rather than the "frozen turbulence"
19 assumption used in previous studies. This investigation developed adjoint equations for the
20 RNG k- ϵ turbulence model and applied it to a two-dimensional ventilated cavity and a
21 three-dimensional, two-person office. Design processes with the adjoint RNG k- ϵ turbulence
22 model led to a near-zero design function for the two cases, while those with the "frozen
23 turbulence" assumption did not. This investigation has successfully used the new method to
24 design a two-person office with optimal thermal comfort level around the two occupants.

25
26 **KEYWORDS**

27 Inverse design, Indoor environment, Adjoint method, Numerical algorithm, Validation,
28 Turbulence model

29
30 **Practical Implications**

31
32 This investigation developed an adjoint RNG k- ϵ turbulence model for optimal design of the
33 indoor environment. The adjoint method with the adjoint RNG k- ϵ turbulence model can be
34 used to design the optimal indoor environment for buildings. Design objective could include
35 air velocity distribution, temperature distribution, and the combination of variables calculated
36 by CFD, such as thermal comfort, indoor air quality, etc.

37
38 **Nomenclature**

39 **g** gravity vector
40 **J** objective function
41 **k** turbulent kinetic energy
42 **k_a** adjoint turbulent kinetic energy
43 **L** augmented objective function
44 **N** incompressible Navier-Stokes equations in vector form

45	p	air pressure
46	p_a	adjoint pressure
47	\mathbf{R}	residual form of the RNG k- ϵ turbulence model equations
48	T	air temperature
49	T_a	adjoint temperature
50	T_{op}	operating air temperature
51	\mathbf{V}	air velocity
52	\mathbf{V}_a	adjoint velocity
53		
54	<i>Subscripts</i>	
55	a	adjoint method
56	op	operate
57		
58	<i>Greek symbols</i>	
59	γ	thermal expansion coefficient of air
60	δ	small constant set by a designer
61	ϵ	turbulent energy dissipation rate
62	ϵ_a	adjoint turbulent energy dissipation rate
63	Θ	design domain
64	κ	effective thermal conductivity
65	ν	kinematic (laminar) viscosity
66	ν_{eff}	effective viscosity
67	ν_k	effective diffusivity for k
68	ν_t	turbulent viscosity
69	ν_ϵ	effective diffusivity for ϵ
70	ξ	design variables

71

72 **1 INTRODUCTION**

73 To create a thermally comfortable and healthy indoor environment, conventional designs use a
74 trial-and-error process¹ that assumes certain thermo-fluid boundary conditions (B.C.), such as
75 air supply inlet size, number, and locations, air supply velocity and temperature, etc. An
76 appropriate method is then used to estimate the resulting distributions of air temperature,
77 velocity, relative humidity, and contaminant concentrations. The trial-and-error process is
78 very time consuming because the assumed thermo-fluid B.C. may not be desirable. Recently,
79 inverse or optimal design processes² have emerged, such as the computational fluid dynamics
80 (CFD)-based³ genetic algorithm (GA) method⁴, CFD-based proper orthogonal decomposition
81 (POD) method⁵, CFD-based artificial neural network (ANN) method⁶, and adjoint method⁷.

82

83 To design a desirable indoor environment, the CFD-based GA method must calculate a large
84 number of samples, and the number of calculations increases exponentially with the number
85 of design variables. To reduce computing effort, Wei et al.⁵ developed a CFD-based POD
86 method that can transform the nonlinear problem into a linear one and build a cause-effect
87 mapping relationship between the objective function and design variables. Since it is a
88 reduced-order method, the accuracy of this method is greatly reduced. The CFD-based ANN

89 method can also build the mapping relationship between the objective function and design
90 variables, in this case by selecting a certain number of samples to train the ANN model. With
91 a well-trained ANN model, the design objective can be predicted without CFD calculations.
92 However, the accuracy of this method depends on the number of samples. In addition, the
93 mapping relationship established by either the CFD-based POD method or the CFD-based
94 ANN method is applicable only to a specific case. For a different case, the relationship would
95 change. The adjoint method can quickly find the optimal design of an indoor environment
96 using an optimization algorithm without building a mapping relationship for each new
97 problem, although it may become trapped in local optima⁸. The adjoint method is the most
98 efficient and suitable method for the inverse design of a thermally comfortable and healthy
99 indoor environment.

100
101 The adjoint method was developed recently⁹ for inversely designing an indoor environment
102 by solving a set of Navier-Stokes equations and adjoint equations, alternatively. The adjoint
103 equations are derived from the continuous Navier-Stokes equations¹⁰. Liu and Chen⁹ used the
104 adjoint method to inversely identify the thermo-fluid B.C. required to achieve the optimal
105 design of ventilation for an enclosed environment. Liu et al.¹¹ then adopted this method to
106 improve the thermal comfort level for an airline cabin. Zhao et al.¹² also attempted to use the
107 adjoint method combined with area-constrained topology and cluster analysis to design a
108 thermally comfortable indoor environment. However, the results of these studies indicated
109 that this method cannot make the objective functions reach the ideal values. One reason may
110 be that the objective functions become trapped in local optima. Another reason may be that
111 the adjoint method, which was used in previous studies^{9, 11-14}, derives only the adjoint
112 equations of the RANS equations with the “frozen turbulence” assumption. With the
113 assumption, the k and ε will not feel changes of the design variables and use the solution (k
114 and ε) of the flow equations when the adjoint equations are solved. The assumption can
115 reduce deriving manually effort^{14, 15} by neglecting variations in the turbulent variables.
116 However, this method can provide only approximate gradients, or even incorrect gradients,
117 which may not lead to global optima for the design.

118
119 A variant of the method is the discrete adjoint method¹⁶, which first discretizes the
120 Navier-Stokes equations and then derives the discrete adjoint equations from the discrete
121 Navier-Stokes equations. Since the discrete adjoint method derives the complete adjoint
122 equations that include the adjoint equations for calculating turbulent viscosity, it can provide
123 accurate gradients¹⁷. However, it requires a large amount of computational memory, which
124 may not be affordable in practice. To obtain accurate gradients and reduce computational
125 memory, one option is to develop a continuous adjoint method without using the “frozen
126 turbulence” assumption¹⁸.

127
128 Since the turbulent viscosity ν_t in the momentum equation can be solved by introducing an
129 appropriate turbulence model, some studies have derived the complete continuous adjoint
130 equations with an appropriate adjoint turbulence model. Zymaris et al. proposed a
131 Spalart-Allmaras one-equation adjoint turbulence model¹⁹ and a standard k - ε adjoint
132 turbulence model²⁰ to minimize duct pressure losses. Papoutsis-Kiachagias²¹ developed a

133 low-Reynolds-number Launder-Sharma k-ε adjoint turbulence model to optimize duct shape,
 134 with the aim of minimizing viscous losses. All these applications have proved that an
 135 appropriate adjoint turbulence model can provide an accurate gradient of the objective
 136 function over the design variables. However, none of the above turbulence models is suitable
 137 for inversely designing an indoor environment. Chen²², Conceição et al.²³, Zhang et al.²⁴ and
 138 Zhang et al.²⁵ compared different turbulence models and found that the RNG k-ε turbulence
 139 model shows the best overall performance for solving indoor airflow among various RANS
 140 models. In addition, there do not have the corresponding adjoint equation. Therefore, this
 141 study aimed to develop an adjoint RNG k-ε turbulence model for inversely designing an
 142 indoor environment.

143

144 2 METHODS

145 2.1 Adjoint method with adjoint RNG k-ε turbulence model

146 To design an indoor environment using the adjoint method, we must first construct a suitable
 147 objective function. For example, the objective function J is a desirable distribution of air
 148 velocity \mathbf{V} and temperature T in the design domain Θ :

149

$$150 \quad J(\xi) = \int_{\Theta} f(\mathbf{V}, T) d\Theta \quad (1)$$

151

152 where ξ is a vector that represents the design variables, such as air supply inlet size, number,
 153 and locations; air supply velocity, $\mathbf{V}_{\text{inlet}}$; air supply temperature, T_{inlet} , etc., that could lead to
 154 the desirable distribution. In this study, air velocity \mathbf{V} and temperature T in the design domain
 155 Θ , as shown in Eq. (1), are controlled by the incompressible, steady-state RANS equations, as
 156 shown in Eqs. (2), (3), and (4), closed with the RNG k-ε turbulence model²⁶, as expressed by
 157 Eqs. (5) and (6).

158

$$159 \quad N_1 = -\nabla \cdot \mathbf{V} = 0 \quad (2)$$

$$160 \quad (N_2, N_3, N_4)^T = (\mathbf{V} \cdot \nabla) \mathbf{V} + \nabla p - \nabla \cdot (2\nu_{\text{eff}} \mathbf{D}(\mathbf{V})) - \gamma \mathbf{g}(T - T_{\text{op}}) = 0 \quad (3)$$

$$161 \quad N_5 = \nabla \cdot (\mathbf{V}T) - \nabla \cdot (\kappa \nabla T) = 0 \quad (4)$$

$$162 \quad R_1 = \nabla \cdot (\mathbf{V}k) - \nabla \cdot (\nu_k \nabla k) - \nu_t P_k - G_b + \varepsilon = 0 \quad (5)$$

$$163 \quad R_2 = \nabla \cdot (\mathbf{V}\varepsilon) - \nabla \cdot (\nu_\varepsilon \nabla \varepsilon) - [C_{\varepsilon 1} (\nu_t P_k + C_{\varepsilon 3} G_b) - C_{\varepsilon 2} \varepsilon] \frac{\varepsilon}{k} \\ + \frac{C_\mu \eta^3 (1 - \eta / \eta_0)}{1 + \beta \eta^3} \frac{\varepsilon^2}{k} = 0 \quad (6)$$

164 where

$$v_{eff} = \nu + \nu_t; \nu_t = c_\mu \frac{k^2}{\varepsilon}; P_k = \left(\frac{\partial V_i}{\partial x_j} + \frac{\partial V_j}{\partial x_i} \right) \frac{\partial V_i}{\partial x_j}; i, j = x, y, z; \quad (7)$$

$$\eta = Sk / \varepsilon; S = \sqrt{(2S_{ij}S_{ij})}; S_{ij} = \frac{1}{2} \left(\frac{\partial V_i}{\partial x_j} + \frac{\partial V_j}{\partial x_i} \right); G_b = \frac{\nu_t}{Pr_t} \gamma \mathbf{g} \nabla T$$

165

166

167 In these equations, p represents the air pressure; $D(\mathbf{V}) = (\nabla \mathbf{V} + (\nabla \mathbf{V})^T)/2$ is the rate of strain
 168 tensor; k is the turbulent kinetic energy; ε is the turbulent energy dissipation; $c_\mu = 0.0845$; $\eta_0 =$
 169 4.38 ; $\beta = 0.012$; $C_{\varepsilon 1} = 1.42$; $C_{\varepsilon 2} = 1.68$; $C_{\varepsilon 3} = 1$; G_b is the buoyancy production rates of
 170 turbulent kinetic energy; P_k is the shear production rates of the turbulence kinetic energy; Pr_t
 171 turbulence Prandtl number for temperature; S is the modulus of the mean rate-of-strain tensor;
 172 \mathbf{N} represents the incompressible, steady-state RANS equations in residual form; and \mathbf{R}
 173 represents the residual form of the RNG k- ε turbulence model equations.

174

175 To minimize the objective function as shown in Eq. (1), this investigation used the steepest
 176 decent method²⁷ as shown in Eq. (8) to update the design variables. Therefore, the adjoint
 177 method was used to calculate the gradients used in Eq. (8).

178

$$\xi_{n+1} = \xi_n - \lambda_n \frac{dJ}{d\xi_n} \quad (8)$$

179

180

181 where ξ_n, ξ_{n+1} are design variables at current and succeeding design cycles, respectively; n
 182 represents the design cycle and λ is the constant step size.

183

184 The adjoint method introduces an augmented objective function L as shown in Eq. (9) and
 185 transforms the constrained design problem into an unconstrained optimization problem.

186

$$L = J + \int_{\Omega} (p_a, \mathbf{V}_a, T_a, k_a, \varepsilon_a) \cdot (\mathbf{N}, \mathbf{R}) d\Omega \quad (9)$$

187

188

189 where Ω represents the computational domain and $p_a, \mathbf{V}_a, T_a, k_a,$ and ε_a are the adjoint
 190 pressure, adjoint velocity, adjoint temperature, adjoint turbulence kinetic energy, and adjoint
 191 rate of dissipation of turbulent energy, respectively.

192

193 From Eq. (9) we can then derive the following adjoint equations of the RANS equations
 194 closed with the RNG k- ε turbulence model. Detailed intermediate derivation process of the
 195 adjoint equations can be found in¹².

196

- Adjoint continuity equation:

197

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$$-\nabla \cdot \mathbf{V}_a + \frac{\partial J_{\Omega}}{\partial p} = 0 \quad (10)$$

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- Adjoint momentum equation:

$$\begin{aligned}
& -\nabla \mathbf{V}_a \cdot \mathbf{V} - (\mathbf{V} \cdot \nabla) \mathbf{V}_a + \nabla p_a - \nabla \cdot (2\nu_{eff} \mathbf{D}(\mathbf{V}_a)) - T \nabla T_a - k \nabla k_a \\
& - \varepsilon \nabla \varepsilon_a + 2 \frac{\partial}{\partial x_j} \left[\left(k_a + \varepsilon_a C_{\varepsilon 1} \frac{\varepsilon}{k} \right) \nu_t \left(\frac{\partial V_i}{\partial x_j} + \frac{\partial V_j}{\partial x_i} \right) \right] + \frac{\partial J_\Omega}{\partial \mathbf{V}} \\
& - \frac{\partial}{\partial x_j} \left[\varepsilon_a \left(\frac{3\eta^2 - 4\eta^3 / \eta_0}{1 + \beta\eta^3} - \frac{(\eta^3 - \eta^4 / \eta_0) 3\beta\eta^2}{(1 + \beta\eta^3)^2} \right) C_\mu \varepsilon \right] = 0
\end{aligned} \tag{11}$$

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- Adjoint energy equation:

$$\begin{aligned}
& -(\mathbf{V} \cdot \nabla) T_a - \nabla \cdot (k \nabla T_a) - \gamma (\mathbf{V}_a \cdot \mathbf{g}) + \frac{\nu_t}{Pr_t} \gamma \mathbf{g} \nabla k_a \\
& + C_{\varepsilon 1} C_{\varepsilon 3} \frac{\nu_t}{Pr_t} \frac{\varepsilon}{k} \gamma \mathbf{g} \nabla \varepsilon_a + \frac{\partial J_\Omega}{\partial T} = 0
\end{aligned} \tag{12}$$

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- Adjoint turbulent kinetic energy equation:

$$\begin{aligned}
& -\nabla \nabla k_a - \nabla \cdot (\nu_k \nabla k_a) + 4\nu_t \frac{1}{k} D(\mathbf{V}) \nabla \mathbf{V}_a + \frac{2\nu_t}{Pr_t} \frac{1}{k} \nabla T \nabla T_a + \frac{2\nu_t}{\sigma_k} \frac{1}{k} \nabla k \nabla k_a \\
& + \frac{2\nu_t}{\sigma_k} \frac{1}{k} \nabla \varepsilon \nabla \varepsilon_a - 2\nu_t k_a \frac{1}{k} P_k - k_a \frac{2\nu_t}{Pr_t} \frac{1}{k} \gamma \mathbf{g} \nabla T - \varepsilon_a C_{\varepsilon 1} \nu_t P_k \frac{\varepsilon}{k^2} \\
& - \varepsilon_a C_{\varepsilon 1} C_{\varepsilon 3} \frac{\nu_t}{Pr_t} \gamma \mathbf{g} \nabla T \frac{\varepsilon}{k^2} - \varepsilon_a C_{\varepsilon 2} \frac{\varepsilon^2}{k^2} + \frac{\partial J_\Omega}{\partial k} \\
& + \varepsilon_a \left[\left(\frac{3\eta^2 - 4\eta^3 / \eta_0}{1 + \beta\eta^3} - \frac{(\eta^3 - \eta^4 / \eta_0) 3\beta\eta^2}{(1 + \beta\eta^3)^2} \right) C_\mu \frac{\eta \varepsilon^2}{k^2} - \frac{C_\mu \eta^3 (1 - \eta / \eta_0) \varepsilon^2}{1 + \beta\eta^3} \frac{1}{k^2} \right] = 0
\end{aligned} \tag{13}$$

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- Adjoint equation for the dissipation rate of turbulent kinetic energy:

$$\begin{aligned}
& -\nabla \nabla \varepsilon_a - \nabla \cdot (\nu_\varepsilon \nabla \varepsilon_a) - 2\nu_t \frac{1}{\varepsilon} D(\mathbf{V}) \nabla \mathbf{V}_a - \frac{\nu_t}{Pr_t} \frac{1}{\varepsilon} \nabla T \nabla T_a - \frac{\nu_t}{\sigma_k} \frac{1}{\varepsilon} \nabla k \nabla k_a - \frac{\nu_t}{\sigma_\varepsilon} \frac{1}{\varepsilon} \nabla \varepsilon \nabla \varepsilon_a \\
& + k_a \nu_t \frac{1}{\varepsilon} P_k + k_a \frac{\nu_t}{Pr_t} \frac{1}{\varepsilon} \gamma \mathbf{g} \nabla T + k_a + 2\varepsilon_a C_{\varepsilon 2} \frac{\varepsilon}{k} + \frac{\partial J_\Omega}{\partial \varepsilon} \\
& + \varepsilon_a \left[- \left(\frac{3\eta^2 - 4\eta^3 / \eta_0}{1 + \beta\eta^3} - \frac{(\eta^3 - \eta^4 / \eta_0) 3\beta\eta^2}{(1 + \beta\eta^3)^2} \right) C_\mu \frac{\eta \varepsilon}{k} + \frac{C_\mu \eta^3 (1 - \eta / \eta_0) 2\varepsilon}{1 + \beta\eta^3} \frac{1}{k} \right] = 0
\end{aligned} \tag{14}$$

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We also give final gradient of the augmented objective function over the design variables:

$$\frac{dL}{dV_{inlet,i}} = \left(\begin{array}{l} \mathbf{S}_{inlet} p_{a,inlet} + (\mathbf{S}_{inlet} \cdot \mathbf{V}_{inlet}) \mathbf{V}_{a,inlet} + \mathbf{S}_{inlet} (\mathbf{V}_{inlet} \cdot \mathbf{V}_{a,inlet}) - \frac{v_{eff} |\mathbf{S}_{inlet}|}{|\mathbf{d}_n|} \mathbf{V}_{a,inlet} \\ + \mathbf{S}_{inlet} T_{inlet} T_{a,inlet} + \mathbf{S}_{inlet} k_{inlet} k_{a,inlet} + \mathbf{S}_{inlet} \varepsilon_{inlet} \varepsilon_{a,inlet} \end{array} \right) \cdot \mathbf{e}_i \quad (15)$$

$$\begin{aligned} \frac{dL}{dT_{inlet}} = & -\gamma \mathbf{g} \cdot \mathbf{V}_{a,inlet} \text{Vol}_{inlet} + \mathbf{S}_{inlet} \cdot (\mathbf{V}_{inlet} T_{a,inlet} - \kappa \frac{\mathbf{d}_n}{|\mathbf{d}_n| |\mathbf{d}_n|} T_{a,inlet} + \frac{v_t}{Pr_t} \gamma \mathbf{g} k_{a,inlet} \\ & + C_{\varepsilon 1} C_{\varepsilon 3} \frac{v_t}{Pr_t} \frac{\varepsilon}{k} \gamma \mathbf{g} \varepsilon_{a,inlet}) \end{aligned} \quad (16)$$

221

222 where $\mathbf{V}_{inlet,i}$ is the inlet air velocity in the i^{th} direction, $i = x, y, z$; \mathbf{S}_{inlet} is the outward-pointing
223 face area vector of inlet cell; $p_{a,inlet}$, $\mathbf{V}_{a,inlet}$, $T_{a,inlet}$, $k_{a,inlet}$ and $\varepsilon_{a,inlet}$ are the adjoint pressure,
224 adjoint velocity, adjoint temperature, the adjoint turbulent kinetic energy and the adjoint
225 turbulent energy dissipation rate at the cell adjacent to the corresponding boundary face,
226 respectively; k_{inlet} and ε_{inlet} are the turbulent kinetic energy and the turbulent energy
227 dissipation rate at inlet boundary face, respectively; \mathbf{d}_n is the direction vector between an inlet
228 cell center and the boundary face center; \mathbf{e}_i is the unit vector in the i^{th} direction, $i = x, y, z$;
229 Vol_{inlet} is the inlet cell volume. When the RANS equations closed with the RNG k - ε
230 turbulence model and the adjoint equations are numerically solved in succession, all state
231 fields and adjoint fields needed for calculating the gradient of the objective function over the
232 design variables are available.

233

2.2 Adjoint boundary conditions

234 The adjoint inlet boundary conditions are that zero p_a , \mathbf{V}_a , T_a , k_a , and ε_a along the inlet. \mathbf{V}_a , T_a ,
235 k_a , and ε_a are set to zero and zero gradient boundary condition of p_a at the wall. While the
236 adjoint outlet boundary conditions can be determined by:

$$\begin{aligned} & \mathbf{n} (\mathbf{V}_a \cdot \mathbf{V}) + \mathbf{V}_a (\mathbf{V} \cdot \mathbf{n}) + v_{eff} (\mathbf{n} \cdot \nabla) \mathbf{V}_a - p_a \mathbf{n} + T_a \mathbf{n} T + k_a \mathbf{n} k + \varepsilon_a \mathbf{n} \varepsilon \\ 238 \quad & - 2 \left(k_a + \varepsilon_a C_{\varepsilon 1} \frac{\varepsilon}{k} \right) v_t \left(\frac{\partial V_i}{\partial x_j} + \frac{\partial V_j}{\partial x_i} \right) \mathbf{n}_j + \frac{\partial J_{\Gamma}}{\partial \mathbf{V}} \\ & + \varepsilon_a \left(\frac{3\eta^2 - 4\eta^3 / \eta_0}{1 + \beta\eta^3} - \frac{(\eta^3 - \eta^4 / \eta_0) 3\beta\eta^2}{(1 + \beta\eta^3)^2} \right) C_{\mu} \varepsilon \mathbf{n}_j = 0 \end{aligned} \quad (17)$$

$$T_a \mathbf{n} \mathbf{V} + \mathbf{n} k \nabla T_a - k_a \frac{v_t}{Pr_t} \gamma \mathbf{g} \mathbf{n} - \varepsilon_a C_{\varepsilon 1} C_{\varepsilon 3} \frac{v_t}{Pr_t} \frac{\varepsilon}{k} \gamma \mathbf{g} \mathbf{n} + \frac{\partial J_{\Gamma}}{\partial T} = 0 \quad (18)$$

$$v_k \mathbf{n} \nabla k_a + k_a \mathbf{n} \mathbf{V} - \frac{2v_t}{Pr_t} \frac{1}{k} \nabla T \mathbf{n} T_a - \mathbf{V}_a 4v_t \frac{1}{k} D(\mathbf{V}) \mathbf{n} + \frac{\partial J_{\Gamma}}{\partial k} = 0 \quad (19)$$

241
$$2\nu_t \frac{1}{\varepsilon} D(\mathbf{V}) \mathbf{V}_a \mathbf{n} + \varepsilon_a \mathbf{V} \mathbf{n} + \nu_\varepsilon \mathbf{n} \nabla \varepsilon_a + \frac{\partial J_\Gamma}{\partial \varepsilon} = 0 \quad (20)$$

242 where \mathbf{n} is the normal component and J_Γ is the objective function defined on boundary.

243

244 **2.3 Numerical method**

245 The adjoint method with the “frozen turbulence” assumption and the adjoint RNG k- ε
 246 turbulence model were previously implemented in OpenFOAM (Open Field Operation And
 247 Manipulation)²⁹. The convection and diffusion terms of the RANS equations closed with the
 248 turbulence model and adjoint equations were discretized by the first-order upwind scheme and
 249 the central difference scheme, respectively. Previous studies^{1,7-9, 11-13} all used the first-order
 250 upwind scheme to discretize the convection terms of both set of equations and none of the
 251 studies have not reported any accuracy issues. Thus, we used the same scheme and did not
 252 explore high-order numerical scheme. The Boussinesq approximation³⁰ was used to simulate
 253 the thermal plume generated by the temperature difference. The convergence criterion was set
 254 as $|J_n - J_{n-1}| < \delta$ (where $n \geq 2$ and $\delta = 10^{-3}$), where J_{n-1} , J_n are the objective function at previous
 255 and current design cycles, respectively.

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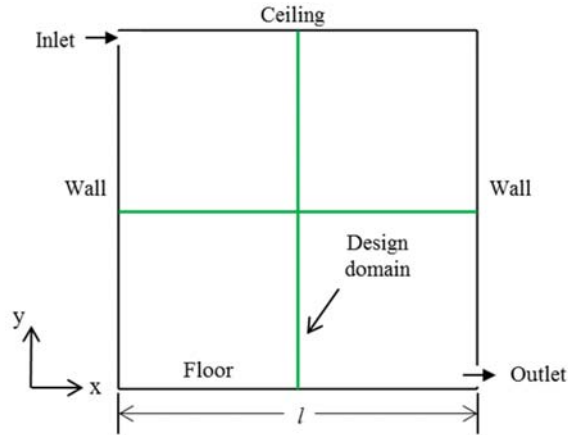
257 **3 RESULTS**

258 In order to verify the performance of the adjoint RNG k- ε turbulence model for inverse design
 259 of an indoor environment, this study first tested the proposed method by applying it to a
 260 two-dimensional ventilated cavity³¹ and a two-person office³² through a step-by-step process.
 261 The two-dimensional ventilated cavity is a simple case with mixed ventilation, while the
 262 three-dimensional ventilated office represents a complete indoor environment with complex
 263 geometries and displacement ventilation. To prove the accuracy of the proposed method and
 264 the necessity of developing a new method, the adjoint method with the “frozen turbulence”
 265 assumption was conducted as a comparison. Finally, this study used the validated method to
 266 inversely design an optimal indoor environment for the two-person office.

267

268 **3.1 Two-dimensional ventilated cavity**

269 The first case is a simple two-dimensional ventilated cavity, as shown in Figure 1, with
 270 experimental data (i.e., velocity and temperature) along the green center lines and detailed
 271 information about all B.C. available³¹. The experiment supplied air through the inlet at the top
 272 of the left wall and exhausted air through the outlet at the bottom of the right wall. With the
 273 exception of the floor, which was heated to 35.5 °C, the temperatures of the ceiling and walls
 274 were all controlled at 15.0 °C.



275

276 FIGURE 1 Schematic of a two-dimensional ventilated cavity

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- CFD simulation:

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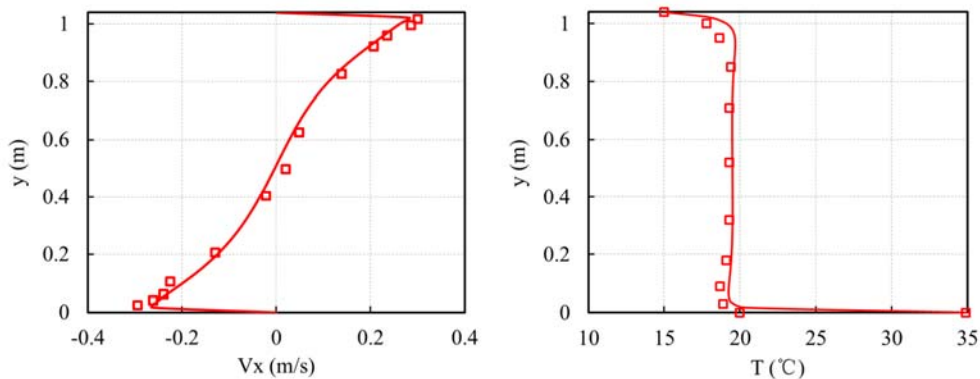
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Before starting the inverse design process, we needed to prove that we had the ability to conduct the CFD simulation correctly. This investigation conducted CFD simulation using experimental B.C. Experimental data along the green center lines was used to verify the results of the simulation. Figure 2 compares the air velocity and temperature profiles predicted by CFD with the RNG k- ϵ turbulence model and the experimental data at $x = 0.5 l$. The comparison indicates that we were able to predict the air distribution accurately in the ventilated cavity. However, the simulation results did not coincide completely with the experimental results. The mean relative error of the air temperature between the CFD simulation results and the experimental data is 2.2%. However, the mean relative error of the air velocity is a little higher, that is because the velocity at the center of the cavity is very small. Thus, there were errors between the experimental data and the results predicted by CFD. The discrepancies may have arisen from measurement errors and uncontrollable factors in the experiment and numerical errors in the CFD simulation, especially the RNG k- ϵ model. Such errors may affect the inverse design results. If the experimental data along the green lines were used as target function for inverse modeling, the solution would not be ideal due to the inherited errors from the CFD in the adjoint equations. Thus, it is better to use CFD results along the green lines as target function because the numerical errors in the CFD simulation are the same as those in the inversed simulation.



297

□ Experimental data — CFD with RNG k-ε turbulence model

298

299 FIGURE 2 Comparison of air velocity and temperature profiles predicted by CFD with the
300 RNG k-ε turbulence model and experimental data from Blay et al.³¹ along the $x = 0.5$ /
301 section.

302

303 • Inverse design process:

304 Since our purpose was to verify the performance of the proposed method using the numerical
305 method, we set the CFD simulation results along the two mid-sections (green lines) as target
306 values to eliminate the influence of experimental errors and numerical errors, and the air
307 supply parameters as the design variables to construct the objective function. If the air supply
308 parameters identified were found to be consistent with the experimental air supply parameters,
309 the proposed method would be verified. For this case, we used the predicted air velocity $\mathbf{V}_{0,ii}$
310 (vector) and air temperature $T_{0,ii}$ in a design domain (the values along the two mid-sections),
311 Θ , as the target values to construct the objective function, which can be expressed as:

312

$$313 \quad J(\xi) = W_1 * V_{\text{norm}} \sum_{ii=1}^m |\mathbf{V}_{ii} - \mathbf{V}_{0,ii}|^2 + W_2 * T_{\text{norm}} \sum_{ii=1}^m (T_{ii} - T_{0,ii})^2 \quad (21)$$

314 where

$$315 \quad V_{\text{norm}} = \frac{1}{V_{\text{inlet},x}^2}; T_{\text{norm}} = \frac{1}{(T_{\text{max}} - T_{\text{min}})^2} \quad (22)$$

316

317 where W_1 and W_2 are the weighting factors, assumed to be 0.5 in this study; V_{norm} and T_{norm}
318 are the normalization factors; and T_{max} and T_{min} are equal to 35.5°C and 15.0°C, respectively;
319 m is the total number data in the design domain; \mathbf{V}_{ii} and T_{ii} are the inversely designed results
320 in the design domain, in this case, respectively. With the above objective function, the adjoint
321 method with the adjoint RNG k-ε turbulence model started its inverse design process from the
322 initial inlet B.C. These B.C. were $\mathbf{V}_{\text{inlet}} = (0.8, 0)$ m/s and $T_{\text{inlet}} = 22.0^\circ\text{C}$. For each design
323 cycle, both the RANS equations closed with the turbulence model and the adjoint equations
324 were calculated with the use of 2,000 iterations to ensure convergence.

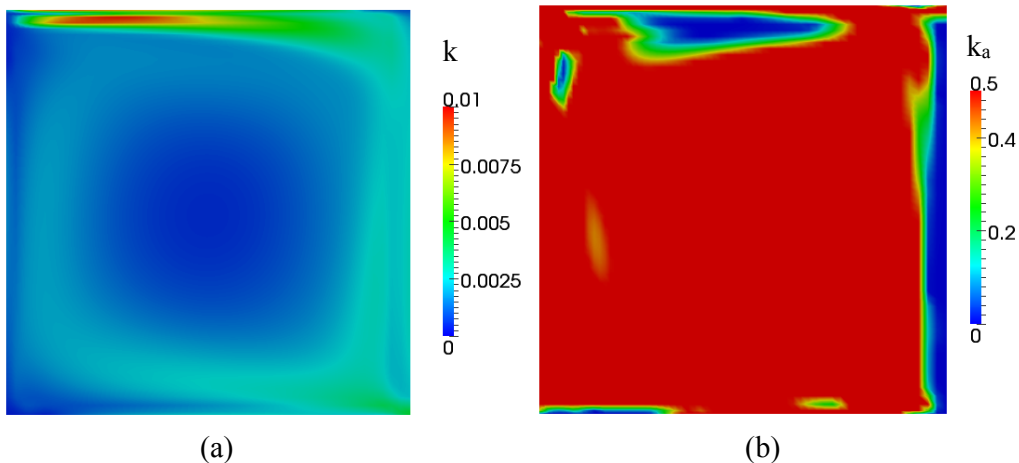
325

326 To study the accuracy of the gradient calculated by the proposed method and rule out the
327 influence of step size, we adopted the steepest descent method with proper constant step size
328 in Eq. (8) to update the design variables. Since an improper constant step size could cause the
329 objective function to become trapped in local optima or the calculation to diverge, this
330 investigation test different step sizes for proper step sizes (0.016 for updating V_{inlet} is 0.016
331 and 200 for updating T_{inlet}).

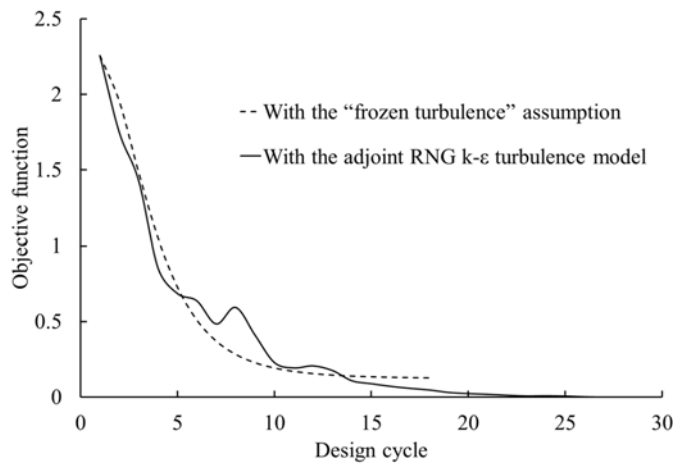
332

333 With the proper step sizes and the same initialized air supply parameters, we also used the
334 adjoint method with the “frozen turbulence” assumption to inversely identify the optimal
335 design variables for the ventilated cavity, in order to illustrate the need for the new method.
336 Figure 3 compared the k with k_a (adjoint k) of these two methods when solving the adjoint

337 equations at the first design cycle that was because these two methods have the same air
 338 supply parameters only in this design cycle during the inverse design process. The k is the
 339 turbulent energy, while the k_a is adjoint turbulent energy. The turbulent energy represents the
 340 degree of chaos of the flow, while the adjoint turbulent energy is the opposite. They are very
 341 different because two sets of equations are different. As a result, the gradients calculated by
 342 these two methods are different. When the objective function minimized by the adjoint
 343 method with the “frozen turbulence” assumption met the convergence criteria, as shown in
 344 Figure 4, the objective function minimized by the adjoint RNG k - ϵ turbulence model reached
 345 the same value at almost the same time. However, the new method can cause the objective
 346 function to continue to decrease. The design variables identified by the adjoint RNG k - ϵ
 347 turbulence model for Case A-2 were almost the same as the known experimental data, as
 348 shown in Table 1. This indicates that the adjoint RNG k - ϵ turbulence model has higher
 349 calculation accuracy.
 350



351
 352 (a) (b)
 353 FIGURE 3 Comparison of (a) k solution of the flow equation and (b) k_a calculated by the
 354 adjoint RNG k - ϵ turbulence model at the first design cycle
 355



356
 357 FIGURE 4 Variation in the objective function with the design cycle for the two-dimensional
 358 ventilated cavity
 359

360

TABLE 1 Design variables identified by two methods compared with the experimental B.C.

	$V_{inlet, x}$ (m/s)	$V_{inlet, y}$ (m/s)	T_{inlet} (K)	J
With the “frozen turbulence” assumption	0.54	4.3E-4	287.36	0.13
With the adjoint RNG k-ε turbulence model	0.56	7.2 E-3	288.15	4.2 E-4
Experimental data ³¹	0.57	0	288.15	

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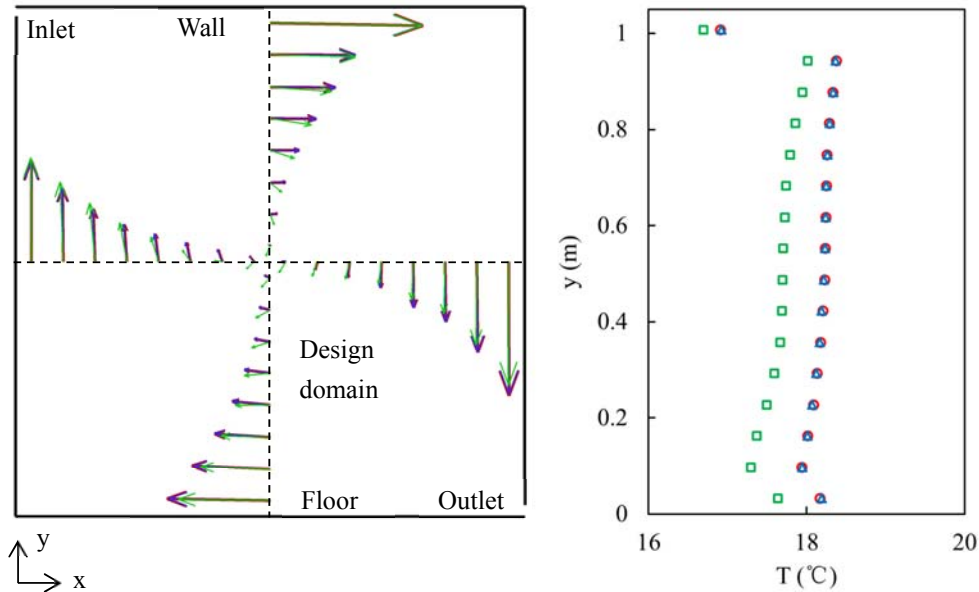
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374

After the optimal air supply parameters, as shown in Table 1, had been identified by the two methods, we conducted CFD simulation with the RNG k-ε turbulence model to determine whether the optimized air distribution was consistent with the target values (red line/dots). Figure 5 illustrates the velocity vector and temperature profiles in the design domain predicted with the air supply parameters identified by the adjoint method with the “frozen turbulence” assumption (green line/dots) and with the adjoint RNG k-ε turbulence model (blue lines/dots). The air distribution in the design domain optimized with the adjoint RNG k-ε turbulence model was closer to the target values than the distribution optimized with the “frozen turbulence” assumption, especially in the center of the design domain. The air temperatures at $x = 0.5 l$ predicted with the “frozen turbulence” assumption were also much lower than the actual values. Thus, the performance of the adjoint RNG k-ε turbulence model was better.



375

- ← CFD simulation results with experimental boundary condition³¹
- ← With B.C. from the “frozen turbulence” assumption
- ← With B.C. from the adjoint RNG k-ε turbulence model
- CFD simulation results with experimental boundary condition³¹
- With B.C. from the “frozen turbulence” assumption
- △ With B.C. from the adjoint RNG k-ε turbulence model

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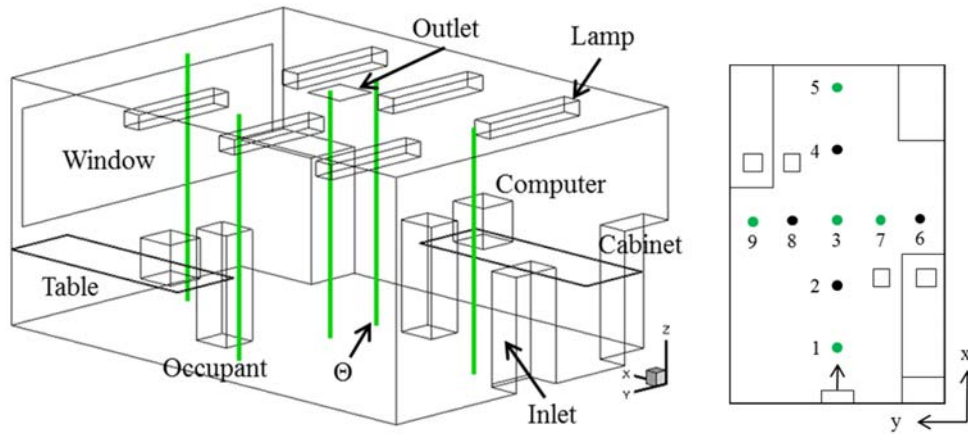
FIGURE 5 Comparison of air velocity vectors in the design domain and temperature profiles at $x = 0.5 l$ predicted by the experimental B.C. from Blay et al.³¹ and the air supply parameters identified by the adjoint method with the “frozen turbulence” assumption and with the adjoint RNG k-ε turbulence model

381

382 3.2 Three-dimensional ventilated office

383 The second case is a three-dimensional ventilated office, as shown in Figure 6, with
384 experimental data³². In the experiment, the office was ventilated by a displacement ventilation
385 system³³ with an air supply velocity of $\mathbf{V}_{\text{inlet}} = (0.09, 0, 0)$ m/s and air supply temperature of
386 $T_{\text{inlet}} = 17$ °C. The air was supplied through a wall diffuser at floor level and exhausted
387 through an outlet at ceiling level. This office was more complicated than the two-dimensional
388 ventilated cavity and closer to reality, although the occupants, computers, cabinets, and lights
389 were all simplified as rectangular boxes. Therefore, it was more practical to use this office for
390 inversely identifying the optimal air supply parameters. Similar to the previous case, we used
391 CFD simulation results along lines 1, 3, 5, 7, and 9 to construct the objective function as
392 expressed by Eqs. (21) and (22). We used a structured grid of hexahedral elements in this
393 geometric model, with 272,513 cells according to our grid-independence tests.

394



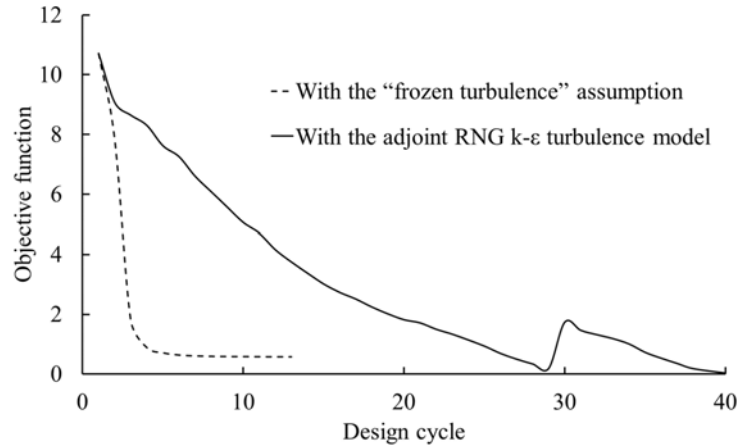
395

396 FIGURE 6 Schematic of a two-person office, and the design domain location where the
397 numbers 1 to 9 indicate locations where air velocity and temperature profiles were measured
398 in the experiment³².

399

400 To start the inverse design process, the inlet B.C. were initialized as $\mathbf{V}_{\text{inlet}} = (0.5, 0, 0)$ m/s and
401 $T_{\text{inlet}} = 20$ °C. Note that the step sizes for updating $\mathbf{V}_{\text{inlet}}$ and T_{inlet} were 0.02 and 100,
402 respectively. Figure 7 shows the variation in the objective function with the design cycle
403 during the inverse design process, with the “frozen turbulence” assumption and with the
404 adjoint RNG k- ϵ turbulence model. The convergence speed with the “frozen turbulence”
405 assumption was ten times faster than that the adjoint RNG k- ϵ turbulence model. This
406 occurred because the gradients calculated with the “frozen turbulence” assumption were
407 larger than those calculated with the adjoint RNG k- ϵ turbulence model in the first two design
408 cycles. However, with the adjoint RNG k- ϵ turbulence model, one could obtain an objective
409 function close to zero.

410



411

412 FIGURE 7 Variation in the objective function with the design cycle for the three-dimensional
 413 ventilated office case

414

415

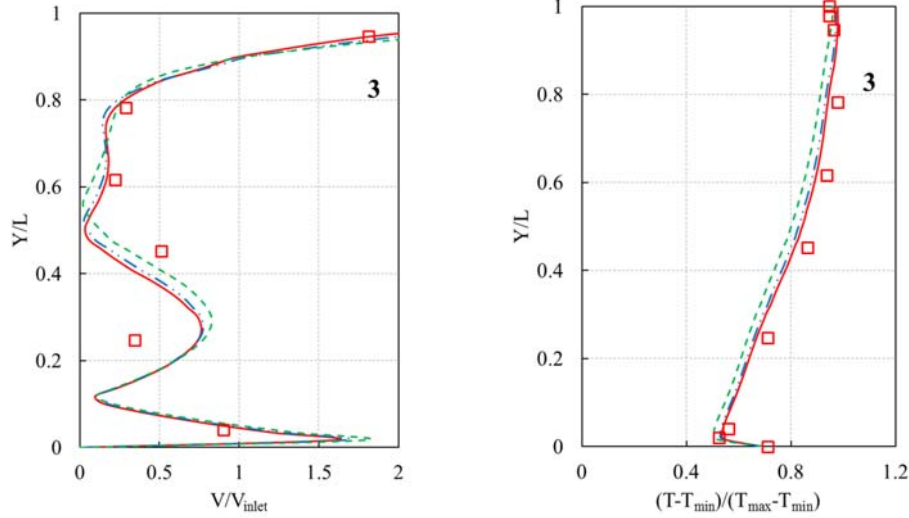
TABLE 2 Design variables identified by two methods compared with experimental B.C.

	$V_{inlet, x}$ (m/s)	$V_{inlet, y}$ (m/s)	$V_{inlet, z}$ (m/s)	T_{inlet} (°C)
With the “frozen turbulence” assumption	0.11	3.0E-4	2.5E-6	17.45
With the adjoint RNG k-ε turbulence model	0.098	6.1E-6	2.1E-6	17.36
Experimental data ³²	0.09	0	0	17.00

416

417 As shown in Table 2, the air supply parameters finally identified by the adjoint RNG k-ε
 418 turbulence model were close to the experimental data. We then used these parameters to
 419 conduct CFD simulations to determine whether the identified air supply parameters formed
 420 the same air distribution as the results with experimental B.C. Figure 8 quantitatively
 421 compares the predicted air velocity and air temperature profiles with the experimental profiles
 422 at the center of the room (line 3) as an example. The air supply parameters identified by the
 423 adjoint RNG k-ε turbulence model were much better than those with the “frozen turbulence”
 424 assumption.

425



426

- Experimental data³²
- CFD simulation results with experimental boundary condition
- - - With B.C. from the “frozen turbulence” assumption
- · - With B.C. from the adjoint RNG k-ε turbulence model

427

428 FIGURE 8 Comparison of air velocity and temperature profiles predicted by the experimental
 429 B.C. from Yuan et al.³² and the air supply parameters identified by the adjoint method with the
 430 “frozen turbulence” assumption and with the adjoint RNG k-ε turbulence model with the
 431 experimental data in the center of the room

432

433 3.3 Inverse design of a comfortable indoor environment for the three-dimensional 434 ventilated office

435 Using the validated adjoint RNG k-ε turbulence model, this investigation next conducted an
 436 optimal design of the indoor environment for the two-person office. In the optimal design
 437 process, the design domain was the surfaces at a distance of 0.1 m away from the occupants,
 438 as shown in Figure 9. We used the predicted mean vote (PMV)³⁴ to evaluate the thermal
 439 comfort level in the domain. The range of PMV values is from -3 to 3, and PMV = 0 signifies
 440 a high comfort level. Our design goal was to identify the optimal ξ (air supply velocity V_{inlet}
 441 and air supply temperature T_{inlet}) that would make the average $|PMV|$ in the design domain
 442 close or equal to zero. Thus, the objective function can be expressed by

443

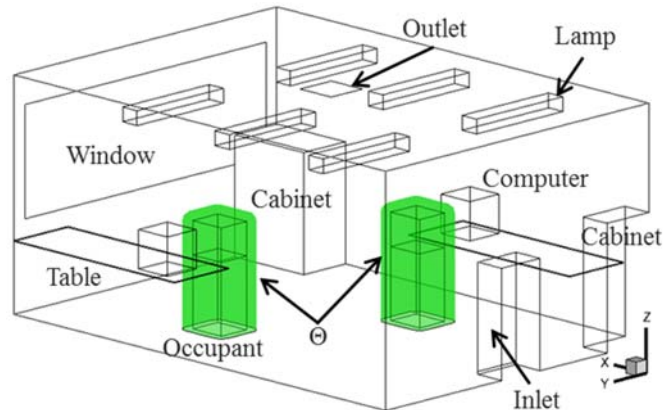
$$444 \quad J(\xi) = \frac{\int_{\Theta} |PMV| d\Theta}{\int_{\Theta} d\Theta} \quad (23)$$

445

446 For a comfortable indoor environment, the air quality must satisfy the design criteria.
 447 According to the ASHRAE Handbook³⁵, the general design criterion used for office buildings
 448 is at least 4 air changes per hour, and the corresponding face velocity for this office was 0.086
 449 m/s. Therefore, we added this constraint ($V_{inlet, x} \geq 0.086$ m/s) to insure indoor air quality

450 during the inverse design process.

451



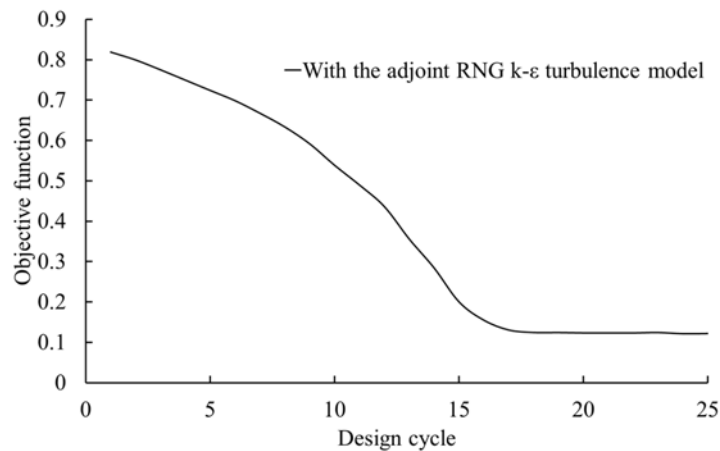
452

453 FIGURE 9 Schematic of a two-person office and inverse design domain

454

455 Figure 10 depicts the variation in the objective function with the design cycle during the
456 inverse design process. The inverse design process achieved the convergence criterion after 18
457 design cycles. The final objective function minimized by the adjoint method with the adjoint
458 RNG k- ϵ turbulence model was 0.122 and did not achieve zero. The reason may be that the
459 objective function was trapped in local optima. The final B.C. identified were $\mathbf{V}_{inlet} = (0.184,$
460 $-5.02E-04, 1.23E-04)$ m/s and $T_{inlet} = 295.16$ °C.

461



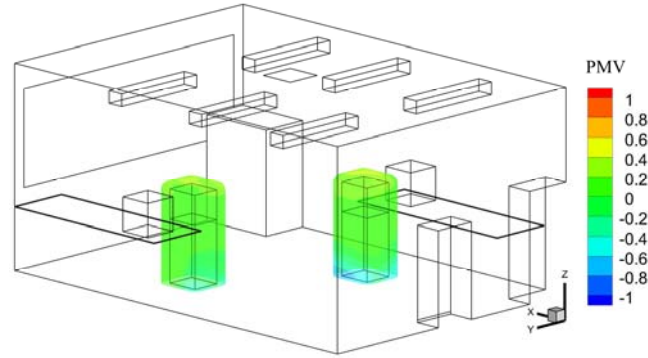
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463 FIGURE 10 Variation in the objective function with the design cycle during the optimal
464 design processes with the adjoint RNG k- ϵ turbulence model.

465

466 Figure 11 depicts the PMV distributions around the occupants inversely designed by the
467 adjoint method with the adjoint RNG k- ϵ turbulence model. The mean PMV value around the
468 occupant is 0.122 and PMV distributions are all near zero. Hence, the adjoint method with the
469 adjoint RNG k- ϵ turbulence model has good performance in the optimal design of the indoor
470 environment.

471



472

473 FIGURE 11 PMV distributions around the occupants inversely designed by the adjoint
 474 method with the adjoint RNG k- ϵ turbulence model

475

476 4 DISCUSSION

477 The adjoint RNG k- ϵ turbulence model improved the optimal design accuracy, but it could not
 478 overcome the inherent disadvantages of the adjoint method. The objective function could also
 479 become trapped in local optima. If the initial design variables were far away from the optimal
 480 values or the step size was not appropriate, the calculation might not lead to the optimal
 481 design.

482

483 This study used only PMV to evaluate the thermal comfort level and to add a constraint to
 484 ensure that the indoor air quality met the standard. With the adjoint method, one could add
 485 further design objectives without increasing the computing costs.

486

487 The design variables determined in Section 3.3 did not make the objective function equal to
 488 zero. This result may imply that the constraints were too restrictive. For example, the design
 489 process did not allow changes in the position, size, or shape of the air supply inlets.

490

491 For the same case with different objective functions in sections 3.2 and 3.3, the convergence
 492 speed is not comparable. This was because the relationship between the objective function
 493 and the design variables was not explicit.

494

495 This investigation only used the first-order upwind scheme and the central difference scheme
 496 to discretize the convection and diffusion terms of the RANS equations closed with the
 497 turbulence model and adjoint equations, respectively. While, the influence of different
 498 discretization schemes on the inverse design results is not clear at the moment and our future
 499 research will conduct an in-depth study and analysis of this issue.

500

501 5 CONCLUSIONS

502 This investigation developed an adjoint RNG k- ϵ turbulence model for the adjoint method for
 503 optimal design of the indoor environment. The following conclusions can be drawn from this
 504 study:

505

- 506 • The design process with the adjoint RNG k- ϵ turbulence model identified design

507 variables that were more accurate than those identified with the “frozen turbulence”
508 assumption. However, the design variables identified with the “frozen turbulence”
509 assumption were more stable during the inverse design process.

- 510 • The adjoint method with the adjoint RNG k- ϵ turbulence model can be used to design the
511 optimal indoor environment for an office. Design objective could include air velocity
512 distribution, temperature distribution, and the combination of variables calculated by
513 CFD, such as thermal comfort, indoor air quality, etc.
- 514 • The design variables must not be overly restrictive. Otherwise, the objective functions
515 may not reach zero, even after numerous design cycles.

516

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522

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