

A&AE 613
Spring 2000
Professor Steve Schneider
Problem Set 5
Handed Out: Monday, 6 March
Due: Friday, 24 March

1. For a 1-D shock wave with a Pr of 1, in a perfect gas with constant specific heats, we derived the following equation:

$$\dot{m} \frac{du}{dx} \frac{\gamma + 1}{2\gamma} \left(1 - \frac{(a^*)^2}{u^2} \right) = \frac{d}{dx} \left(\tilde{\mu} \frac{du}{dx} \right),$$

where $u \rightarrow u_1$ as $x \rightarrow -\infty$ and $u \rightarrow u_2$ as $x \rightarrow \infty$, and where $\tilde{\mu} = \lambda + \frac{4}{3}\mu$.

- (a) Nondimensionalize this equation using the mean free path l , the sound speed at Mach 1, a^* , and the mass flux \dot{m} ; non-dimensionalize $\tilde{\mu}$ in terms of its value at Mach 1, $\tilde{\mu}^*$ (we will define various representations for $\tilde{\mu}/\tilde{\mu}^*$). All the remaining parameters can be grouped into one shock-Reynolds number parameter, Re . What is this parameter?
- (b) Use Stoke's hypothesis that $\lambda = -\frac{2}{3}\mu$, and Chapman's expression for the mean free path in dilute non-polar gases, $\mu \approx 0.67\rho la$, (see White eqn. 1-32) to simplify the expression for Re .
- (c) Reduce the second order ODE to a quadrature (meaning an integral) using analytical methods. Discuss the meaning of any integration constants which may appear, and provide a means of evaluating these constants. Hint: Remember that as $x \rightarrow \pm\infty$, $\frac{du}{dx} \rightarrow 0$. If $u \rightarrow u_1$ upstream, what must u go to downstream? (see part (1f) below). What is the only reasonable choice for u/a^* at $x = 0$?
- (d) Assume that the viscosity μ is linear in the temperature T . Simplify the integral of part (1c). For air with $\gamma = 1.4$ and for upstream velocities $u_1/a_1 = 1.1, 2, \text{ and } 10$, perform the numerical integration, and plot the non-dimensional velocity profile through the shock. How many mean-free paths is the shock width? Hint: What is the behavior of the integral as $u \rightarrow u_1$ or u_2 ? What does

$u(x)$ look like? How can you change the usual method of integration in order to capture the behavior of the solution near $x = 0$?

- (e) Assume that the viscosity μ follows a Sutherland law,

$$\mu = 10^7 \frac{AT^{3/2}}{T + B},$$

where $A = 145.8$ and $B = 110.4$ for air, and T is measured in Kelvin and μ in poise (see Hilsenrath et al., NBS circular 564, 1955). Repeat the evaluations of part (1d) for this more accurate viscosity law. Are the differences significant?

- (f) Compare your results for the downstream velocities to those expected using the jump conditions for shock waves (in particular, the Prandtl-Meyer relation $u_1 u_2 = (a^*)^2$).

2. White, problem 7-8.