## Stationary crossflow vortices on a $10-$ deg. half-angle sharp cone at 15-deg. angle of attack, Mach 8, Sandia tunnel.



Copy from Dan Reda 3/2003, scanned 600 dpi at Ames from photographic print. $\mathrm{M}=7.84, \operatorname{Re}=2 \mathrm{x} 10^{\wedge} 6 / \mathrm{ft}$, base diam. 4 inches, liquid crystal image. See `Experimental Aerodynamics Research on a Hypersonic Vehicle', by W. Oberkampf, D. Aeschliman, R. Tate, and J. Henfling, SAND92-1411, April 1993. Appears to be same image as Fig.21b, where Re_L given as $4.86 \times 10^{\wedge} 6$. Similar to McDevitt. AEDC may have other images.

