

# Issam Mudawar

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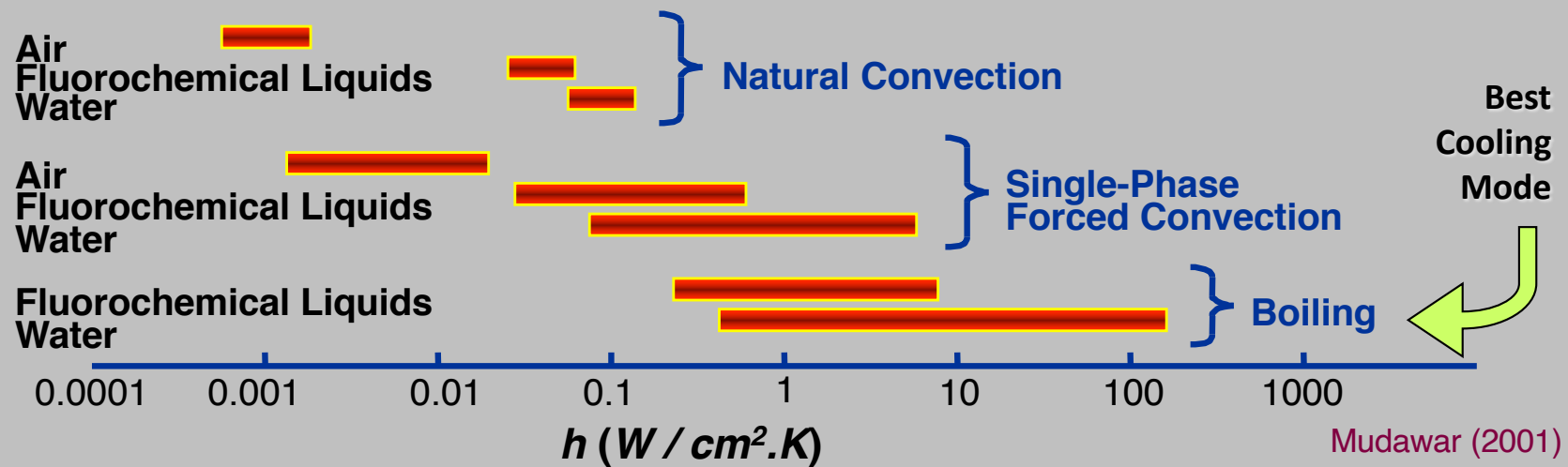
# *Boiling Configurations*

## High-Flux ( $25 - 10^3 \text{ W/cm}^2$ )

Power plant boilers  
Food production  
Geothermal systems  
Metal production  
Computer chips  
Fuel cells  
Spacecraft electronics

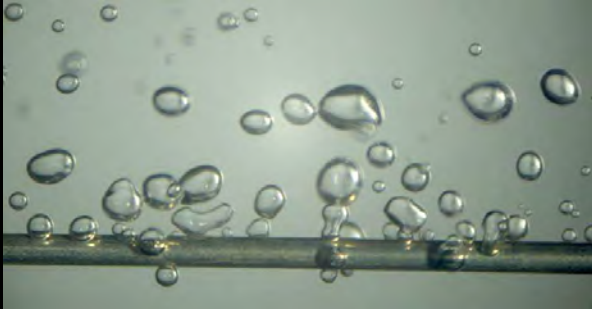
## Ultra-High-Flux ( $10^3 - 10^5 \text{ W/cm}^2$ )

Pressurized water reactors  
Particle accelerators  
Laser & microwave weapons  
Radar systems  
Rocket nozzles  
Turbine blades  
Afterburners



- Constant pursuit of more compact and lightweight packaging, and better performance (lower junction temperature) has lead to alarming increases in power density
- Because thermal management is mostly an afterthought, pushing limits of existing cooling solutions (e.g., using air-cooled heat sinks) often results in costly upper bounds to many technologies
- What is presently needed is a new class of thermal management solutions capable of dissipating very high power densities
- With this capability, these solutions will allow vast advances in performance without the need for costly changes in system packaging
- Systems utilizing liquid-to-vapor phase-change are the most effective thermal management solutions for high-power-density applications

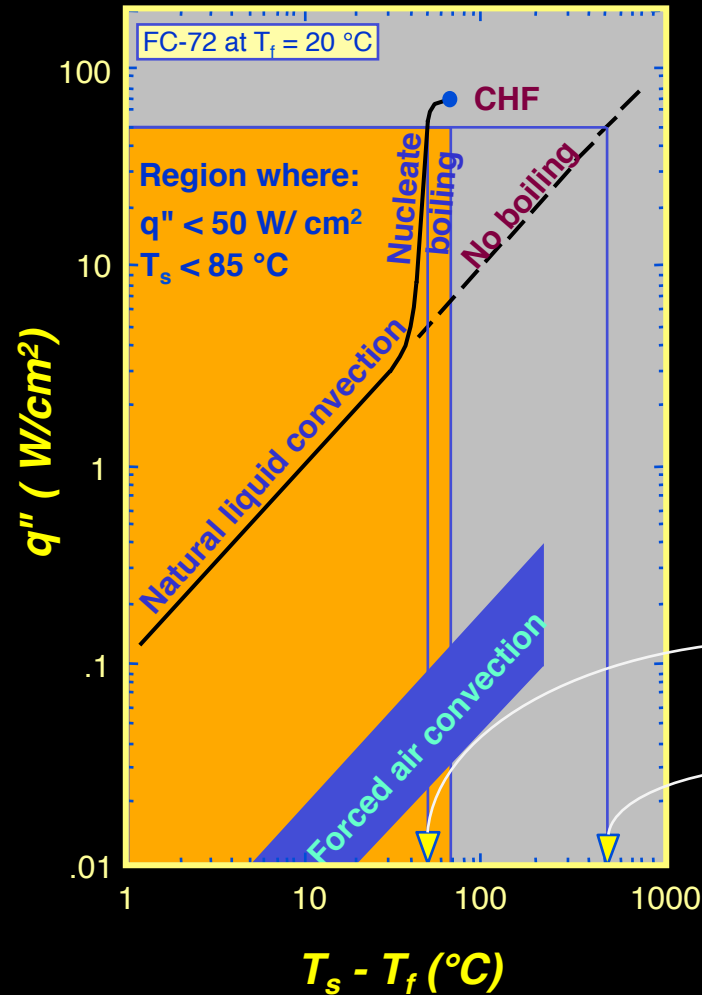
## Nucleate Boiling



## Critical Heat Flux



## Film Boiling



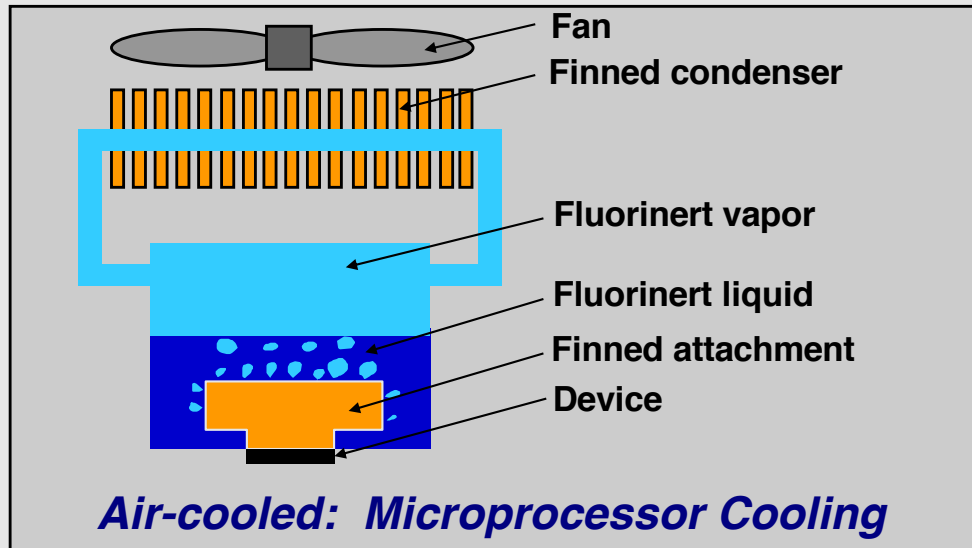
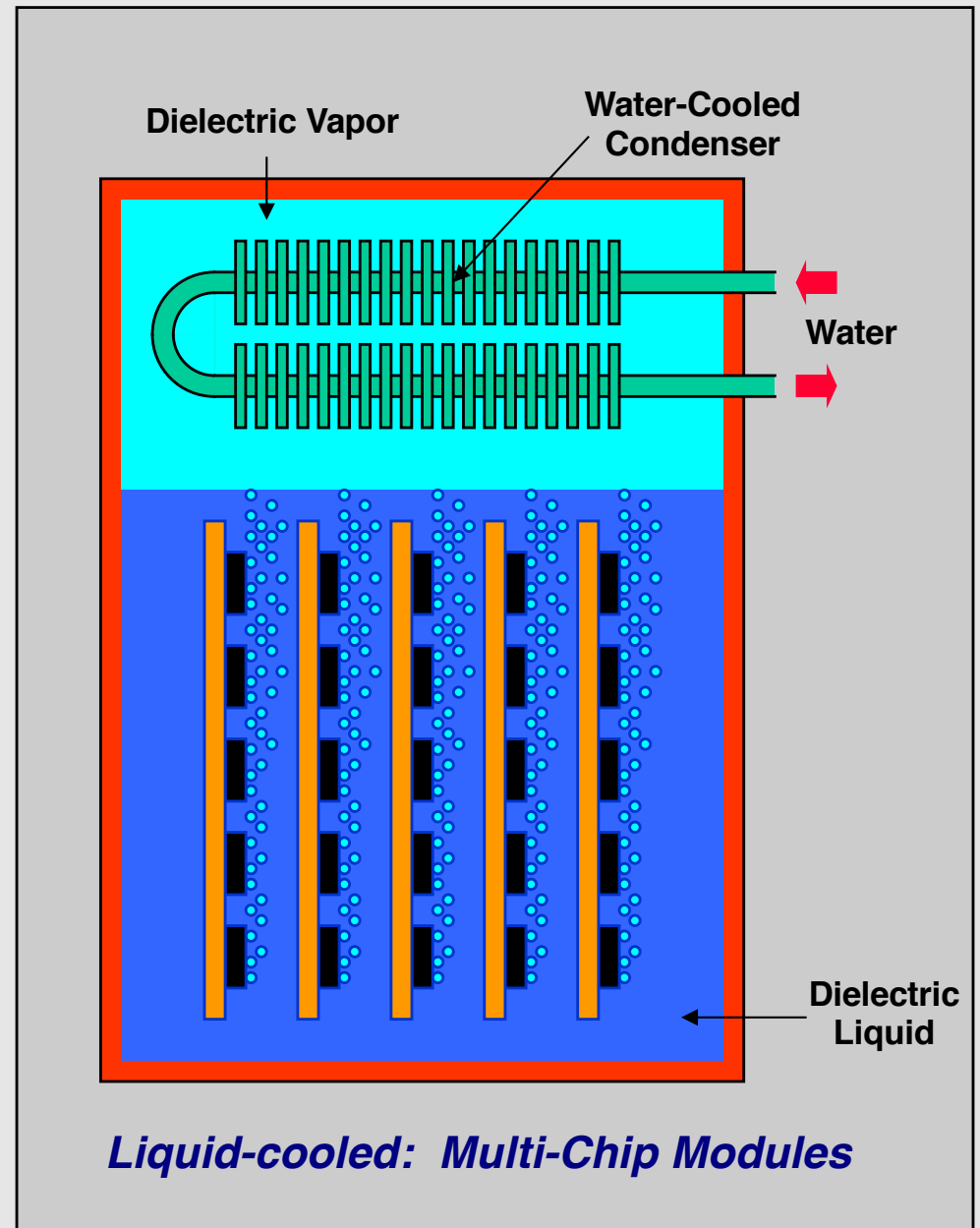
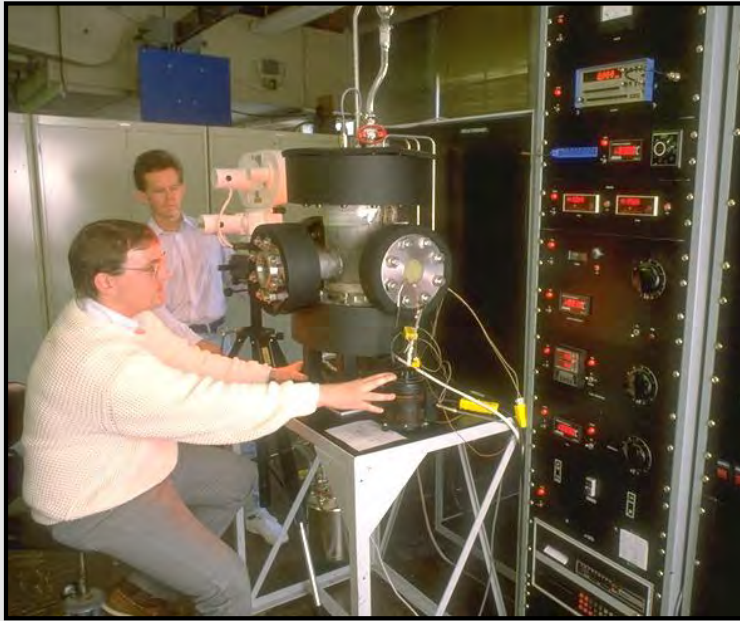
## Key Merits of Boiling:

- (1) Modest changes in device temperature corresponding to broad changes in device heat flux (in nucleate boiling region)
- (2) Much lower device temperatures than with single-phase cooling. For this device:

$T_s = 70\text{ }^{\circ}\text{C}$  at  $50\text{ W/cm}^2$  with boiling

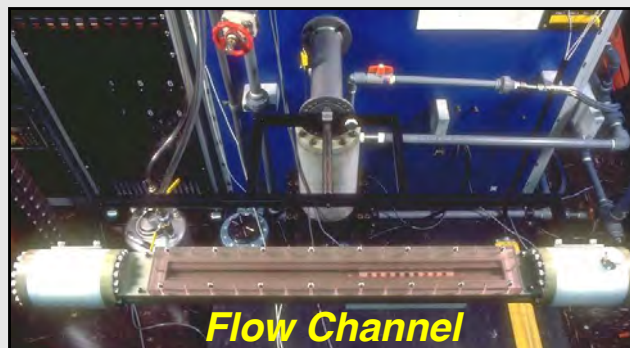
$T_s = 525\text{ }^{\circ}\text{C}$  at  $50\text{ W/cm}^2$  without boiling



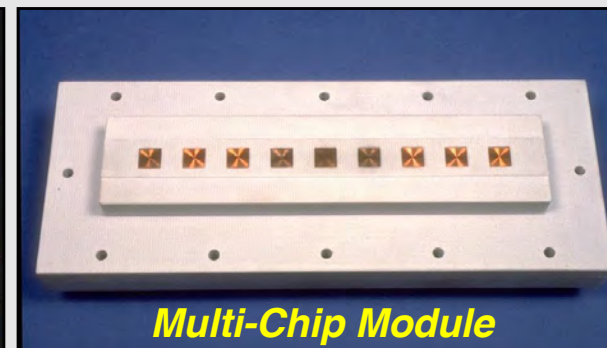




$CHF = 361 \text{ W/cm}^2$

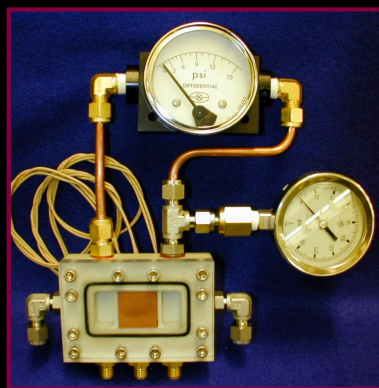
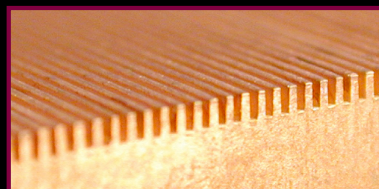
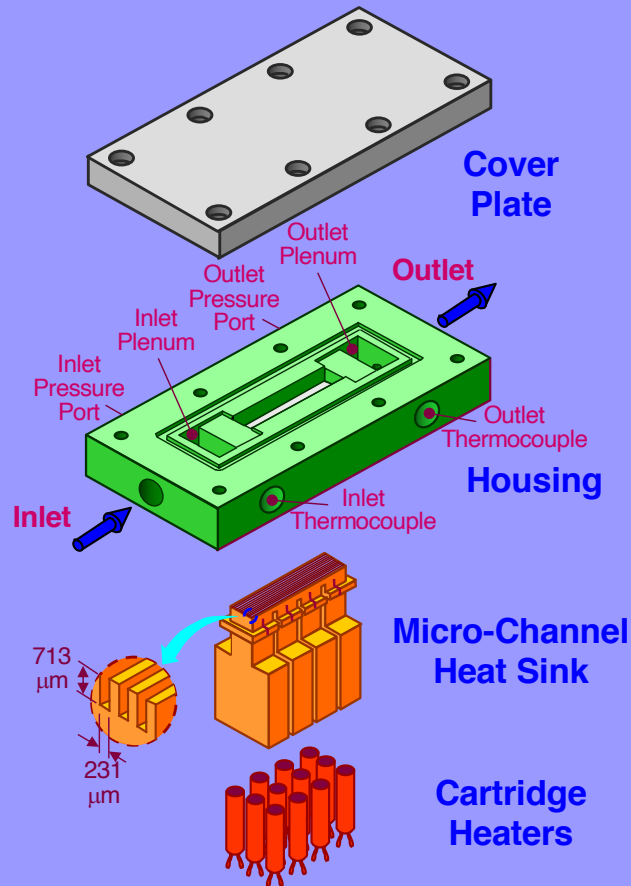


*Flow Channel*



*Multi-Chip Module*

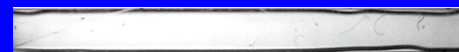




Slug Flow



Annular Flow



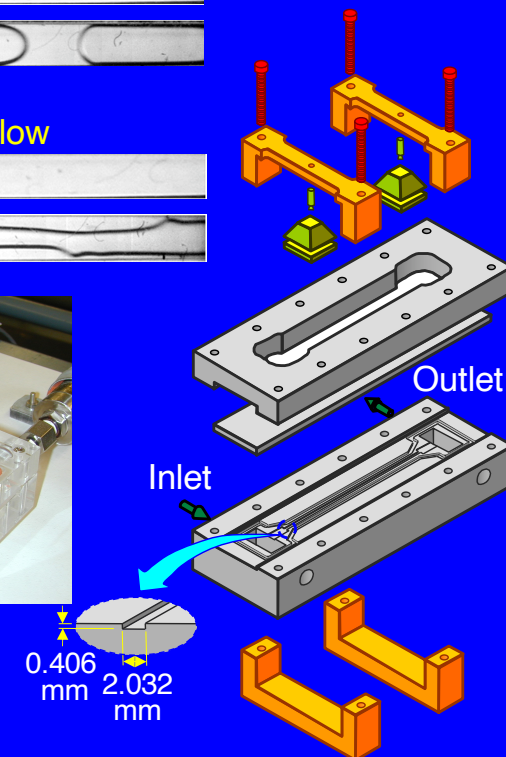
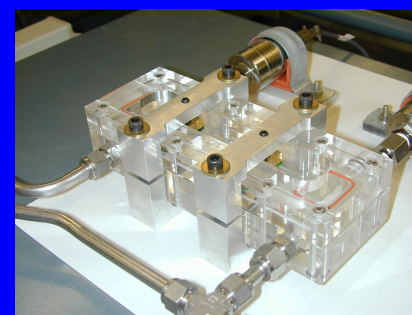
Bubbly/Slug Flow

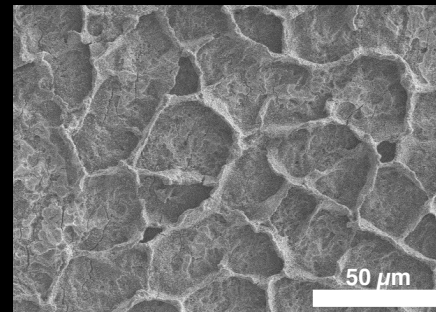
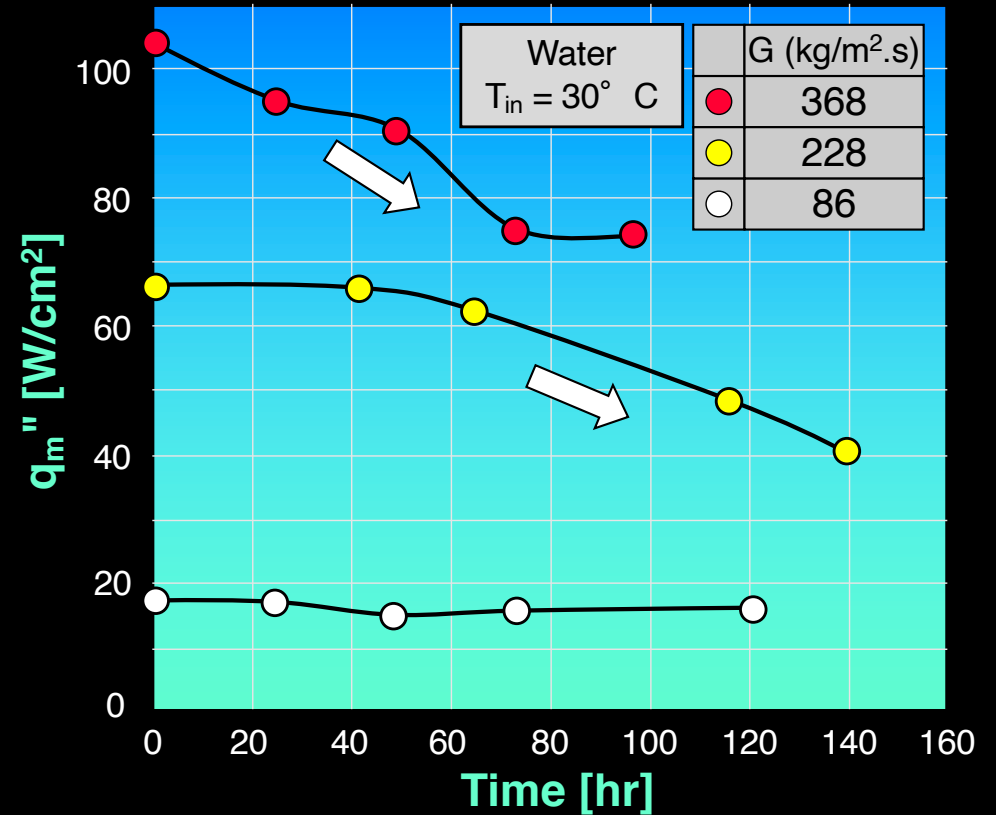
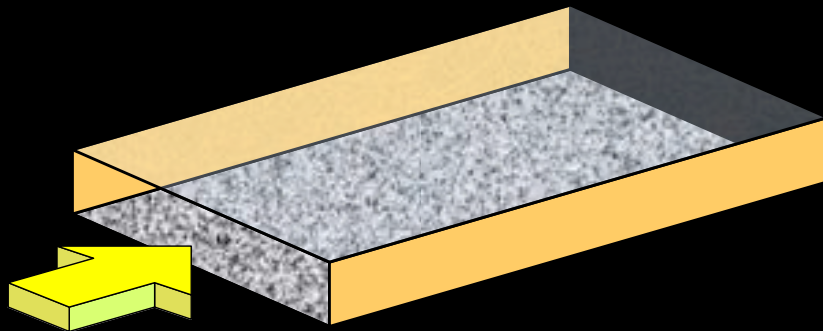
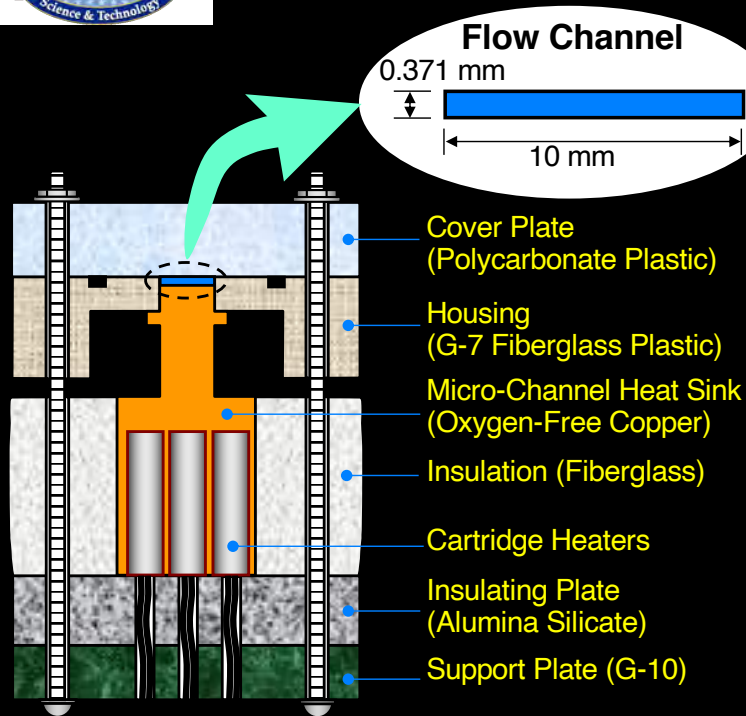


Liquid/Slug

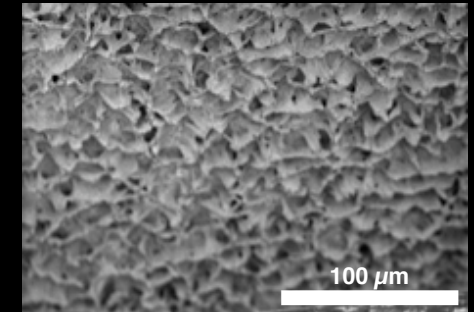


Liquid/Annular Flow





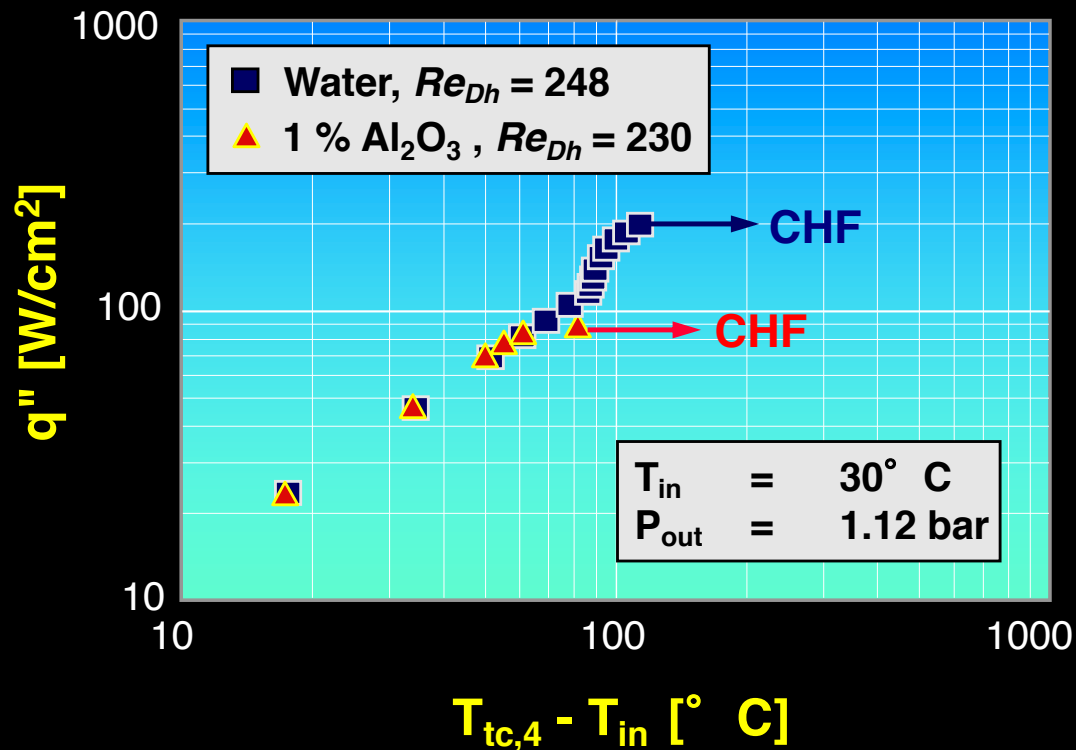
Isolated cellular formations



Dominant 'fish-scale' pattern

$G = 368 \text{ kg/m}^2\cdot\text{s}$ ,  $T_{in} = 30^\circ \text{C}$ , after 5 tests

Khanikar, Mudawar & Fisher (2009)



- Stable flow boiling never achieved due to CHF occurrence shortly following onset of boiling
- Severe nano particle deposition inside micro-channels
- Particle clustering began forming at channel outlet, then propagated rapidly upstream
- Temperature of heat sink rose continuously after particle clogging



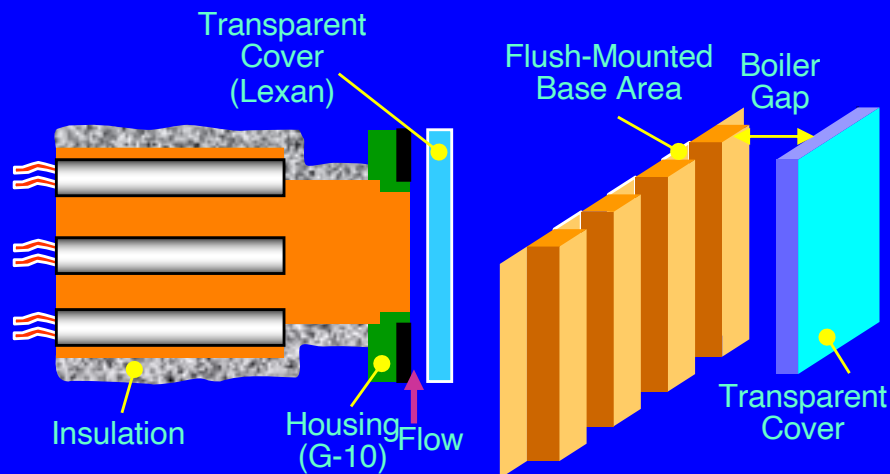
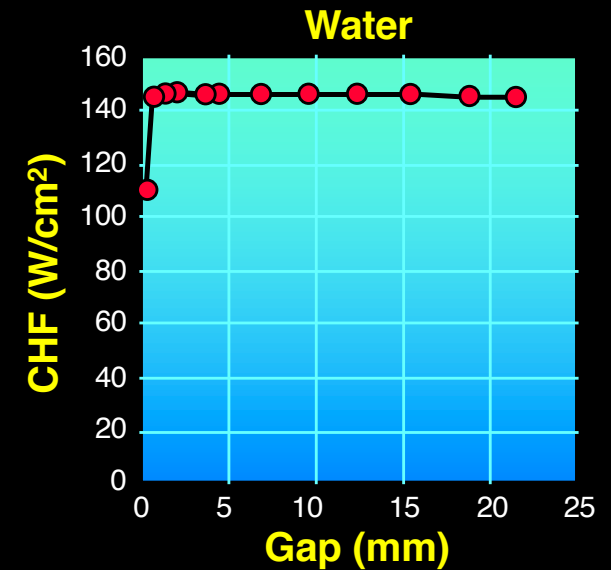
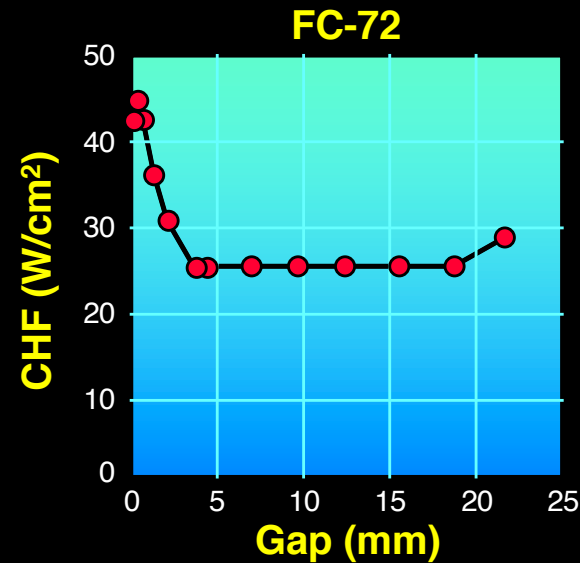
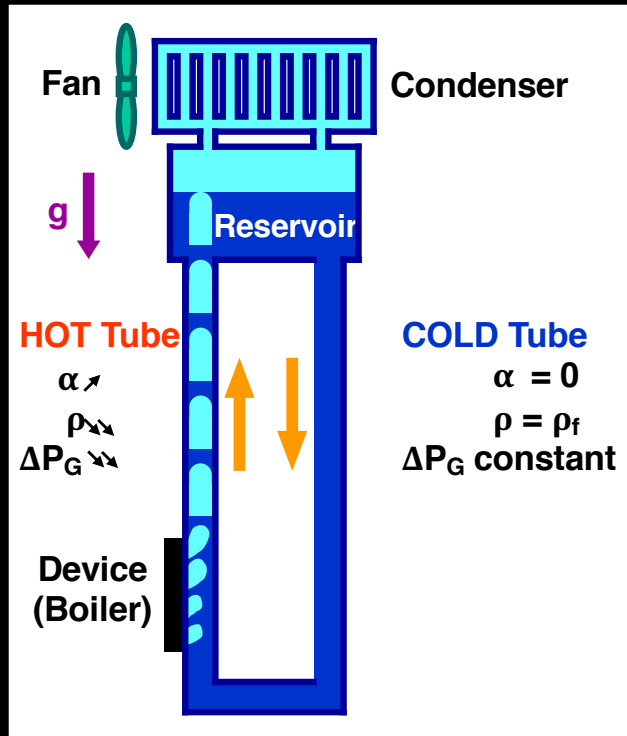
215  $\mu m$  x 821  $\mu m$

**Enhancement efforts should focus on surface itself, not fluid!**



Lee & Mudawar (2006)





**FC-72**  
0.51 mm gap  
50% CHF

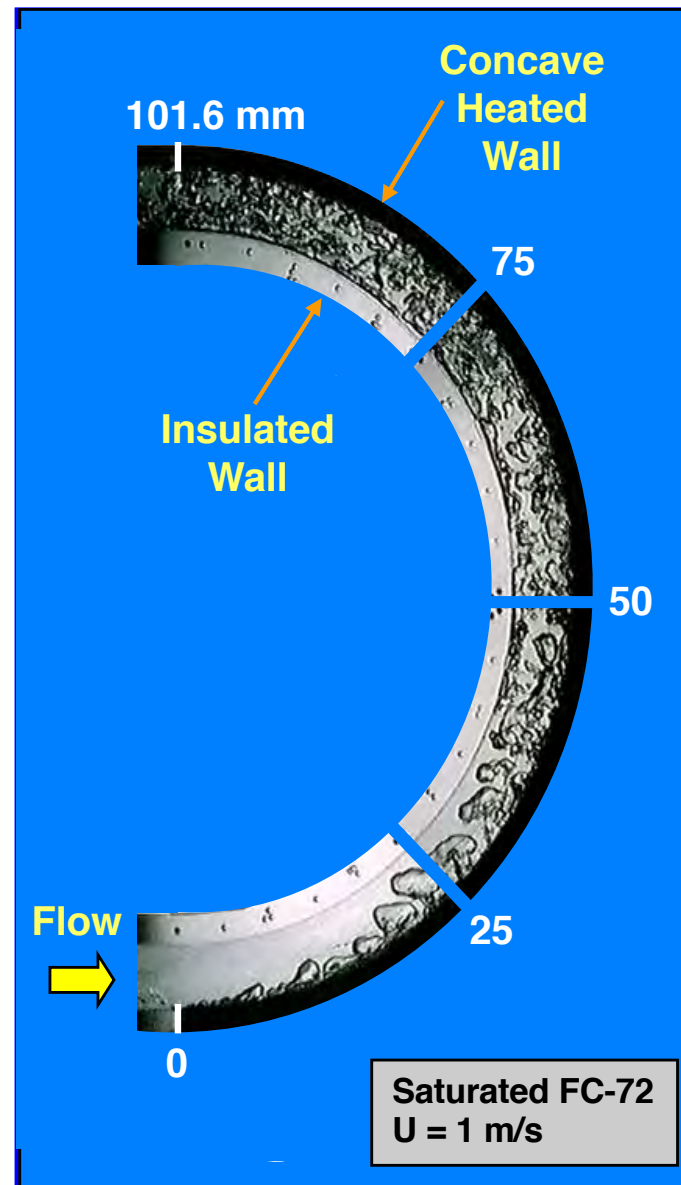
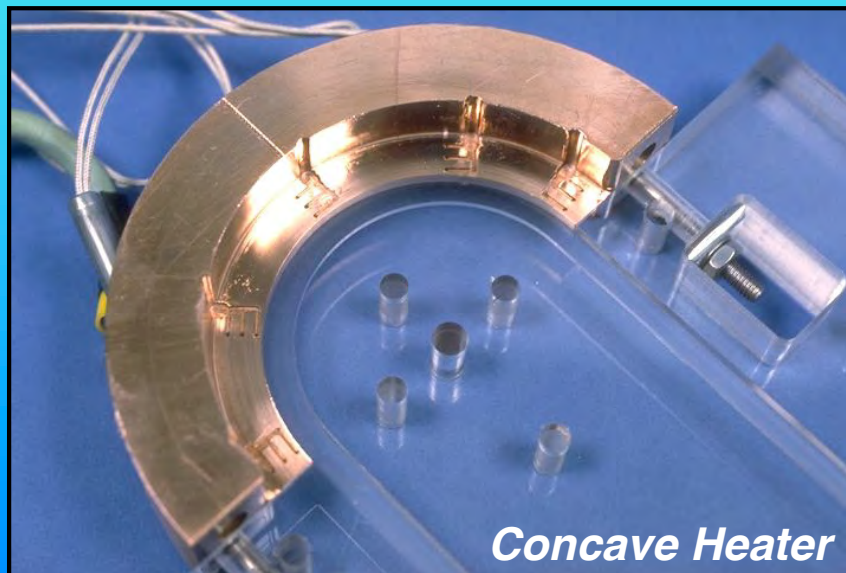
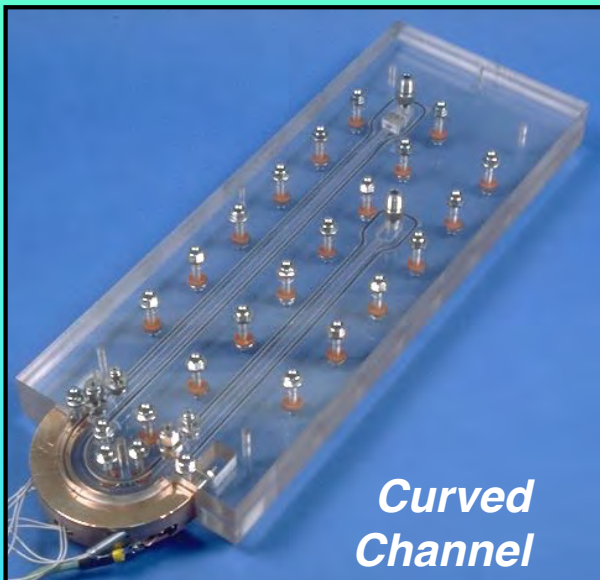


**Water**  
0.13 mm gap  
50% CHF

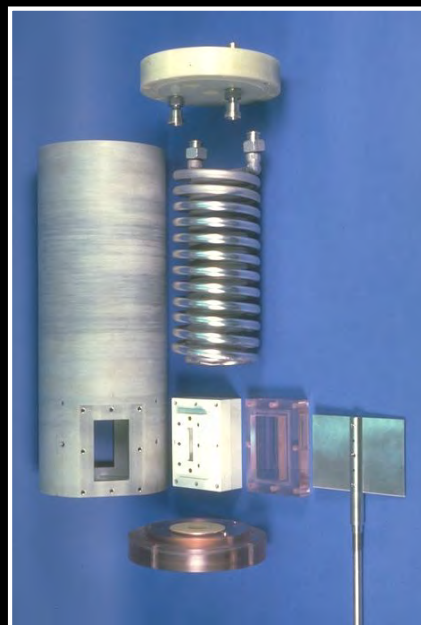
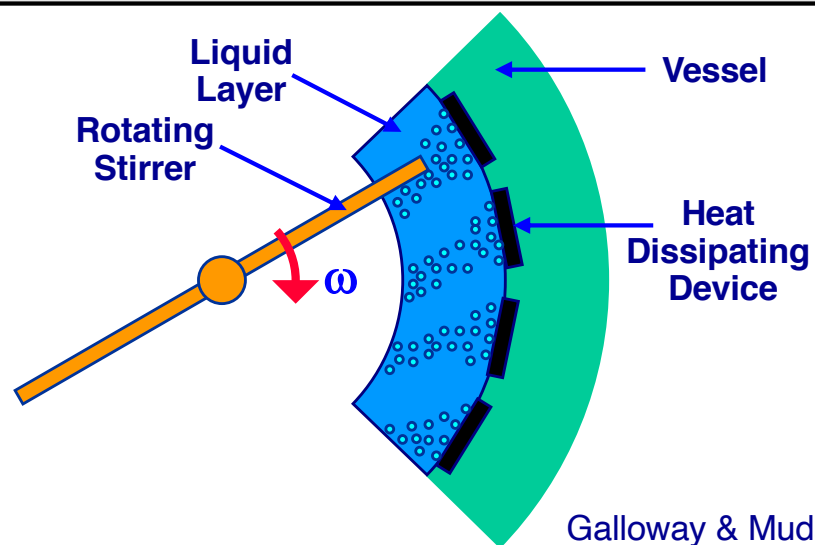


Mukherjee & Mudawar (2003)

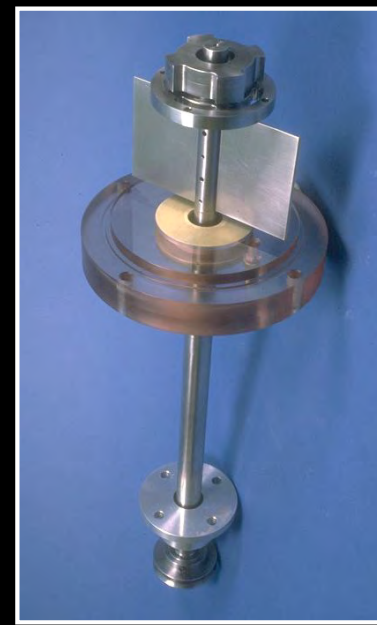




Galloway & Mudawar (1995)

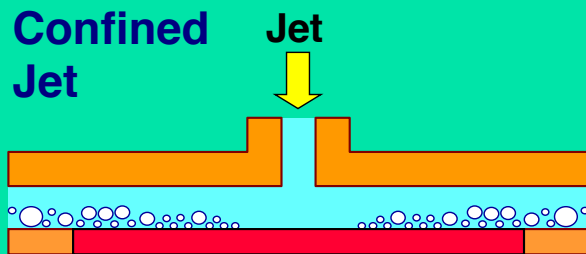
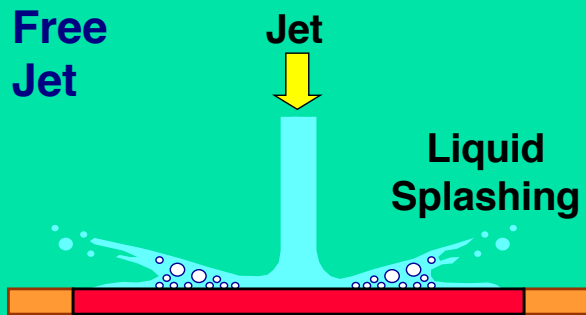
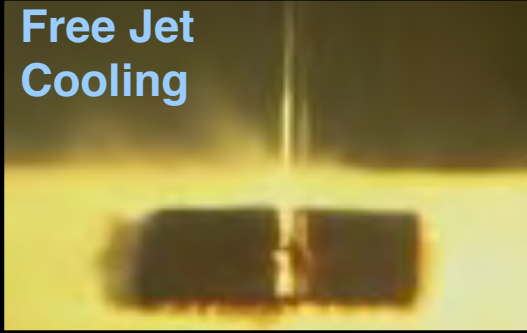


**Test Apparatus**



**Stirrer Assembly**

Free Jet  
Cooling



Estes & Mudawar (1995)

**IBM Raytheon**



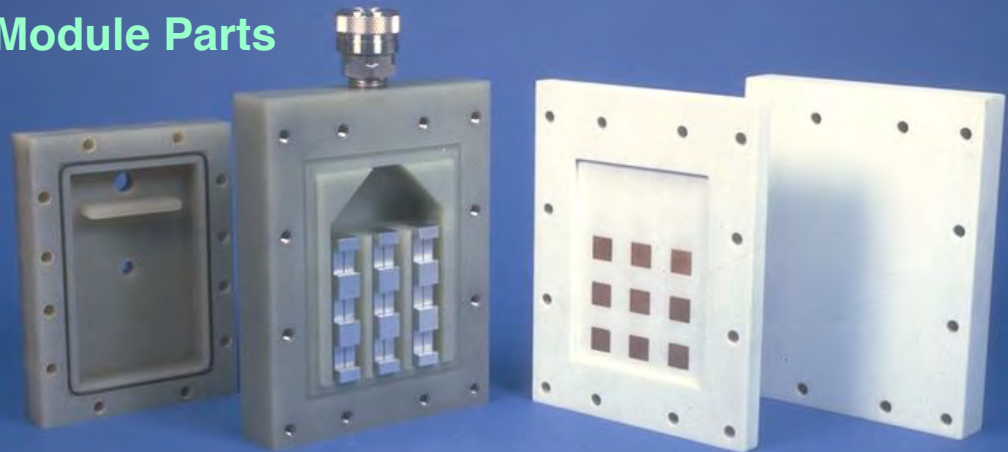
**Highest CHF of 411 W/cm<sup>2</sup>  
achieved with subcooled  
flow boiling FC-72 &  
Surface Enhancement**



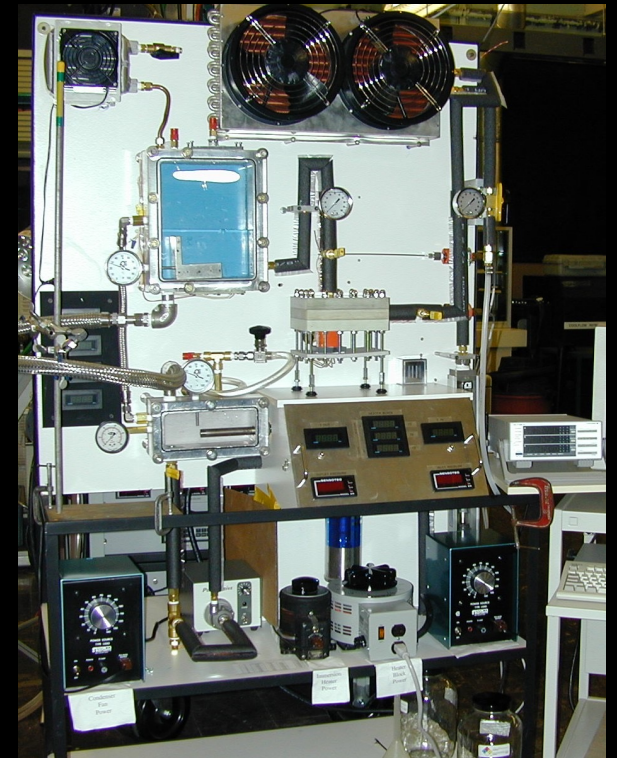
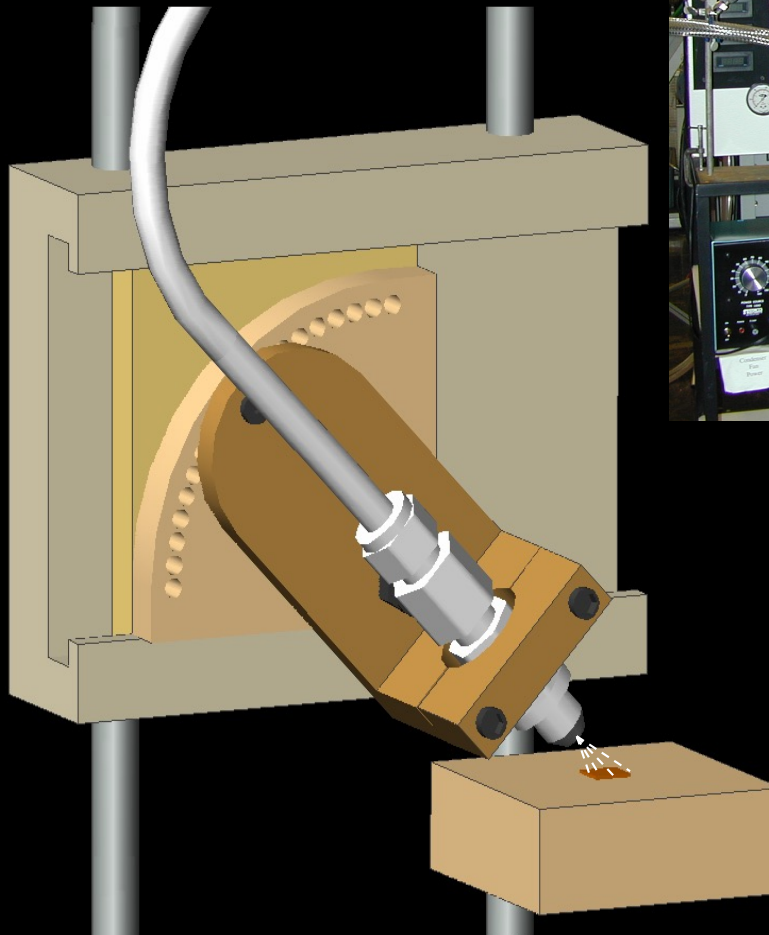
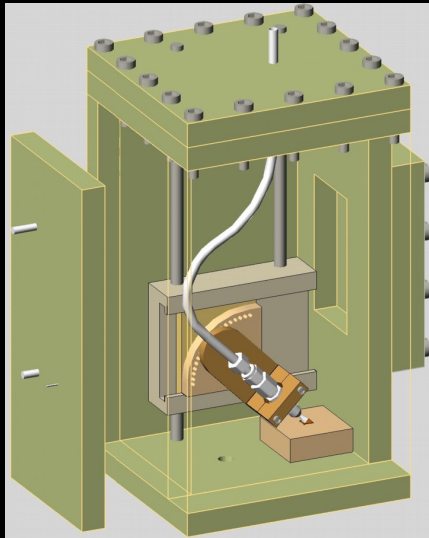
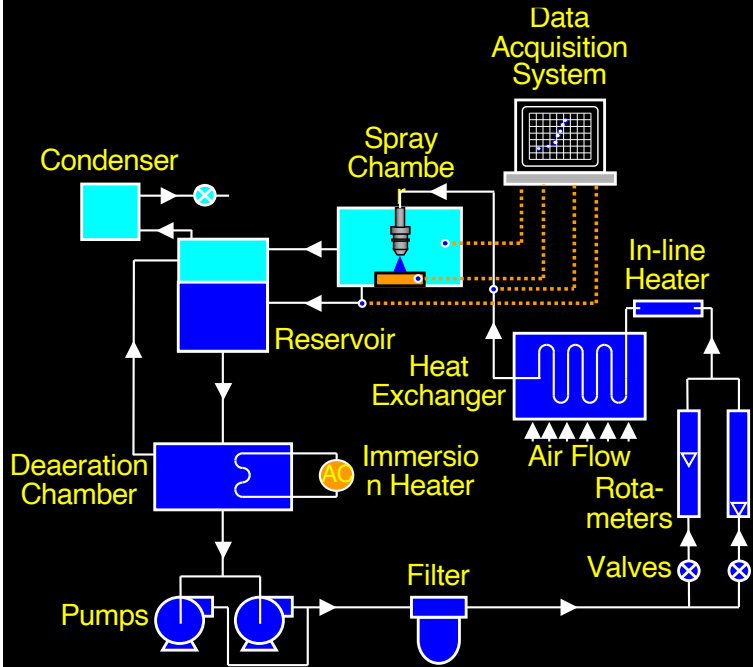
Confined Jet Module

Confined Jet  
Module Parts

Wadsworth & Mudawar (1990)







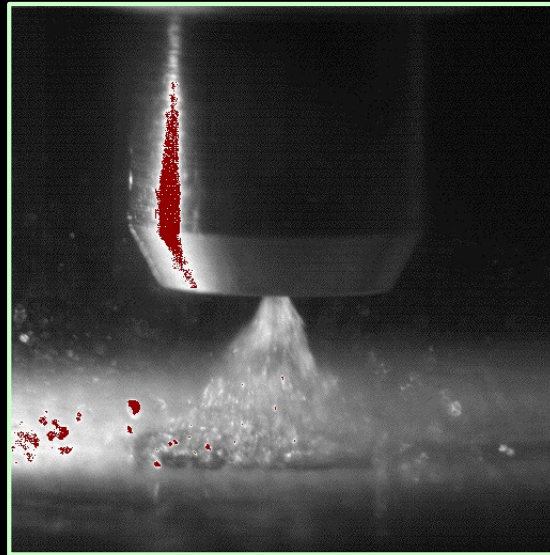
Visaria & Mudawar (2008)

# Boiling Configurations: Spray Cooling

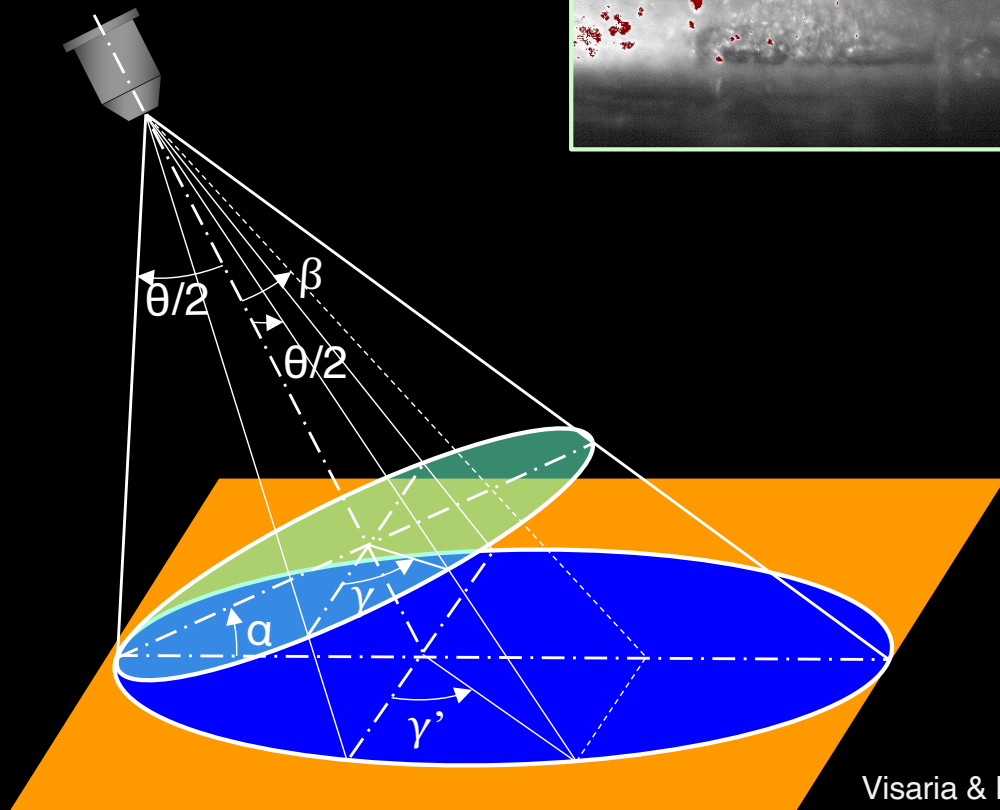
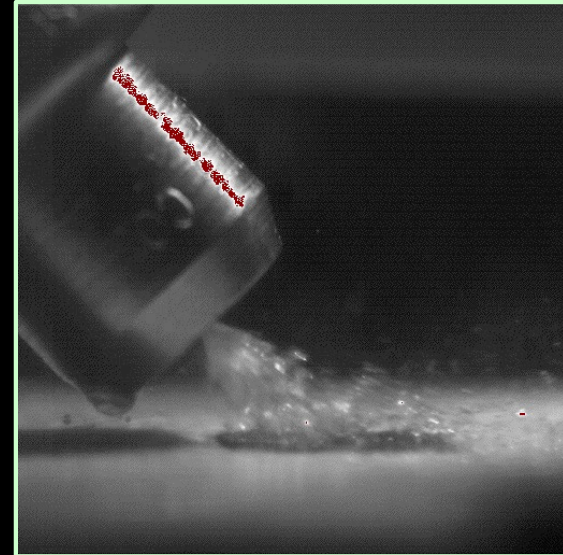
Nozzle 1 ( $\alpha = 55.8^\circ$ )  
PF 5052 ( $T_{\text{sat}} = 50^\circ \text{ C}$ )  
 $T_{\text{in}} = 24^\circ \text{ C}$   
 $Q = 3.86 \times 10^{-6} \text{ m}^3/\text{s}$

Recording speed: 6000 fps  
Playback speed: 60 fps

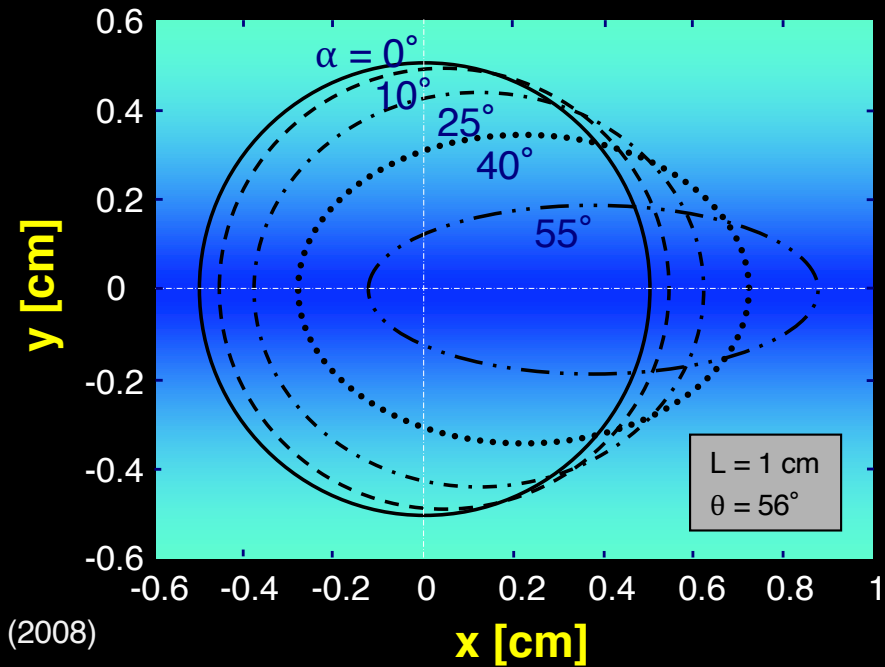
$\alpha = 0^\circ$



$\alpha = 50^\circ$



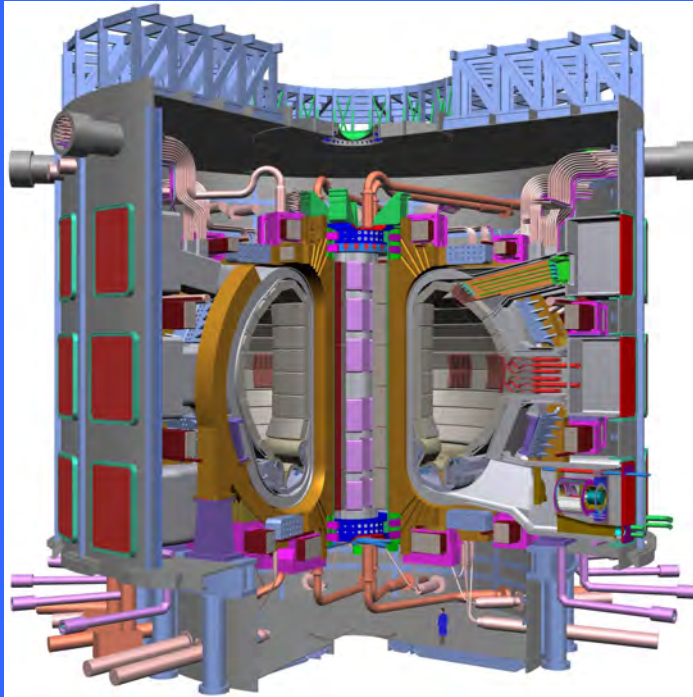
Visaria & Mudawar (2008)



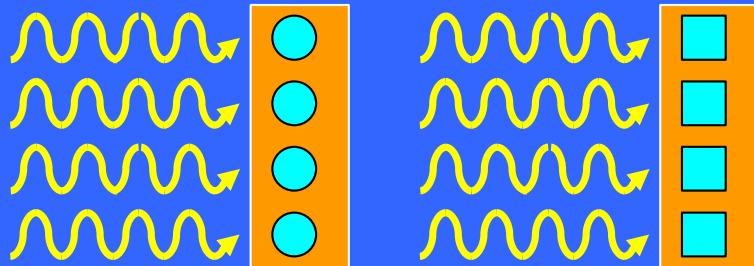
# *BTPFL Applications*



## Fusion Reactor Blanket



International Thermonuclear  
Experimental Reactor (ITER)



Heat flux in cooling channels can exceed  $10^4 \text{ W/cm}^2$

- Magnetic containment of plasma in donut-shaped vacuum vessel (tokamak)
- Fusion reactor blanket extracts fusion power to generate electricity
- Highly subcooled water cooling at high pressure and high mass velocity required to guard against CHF

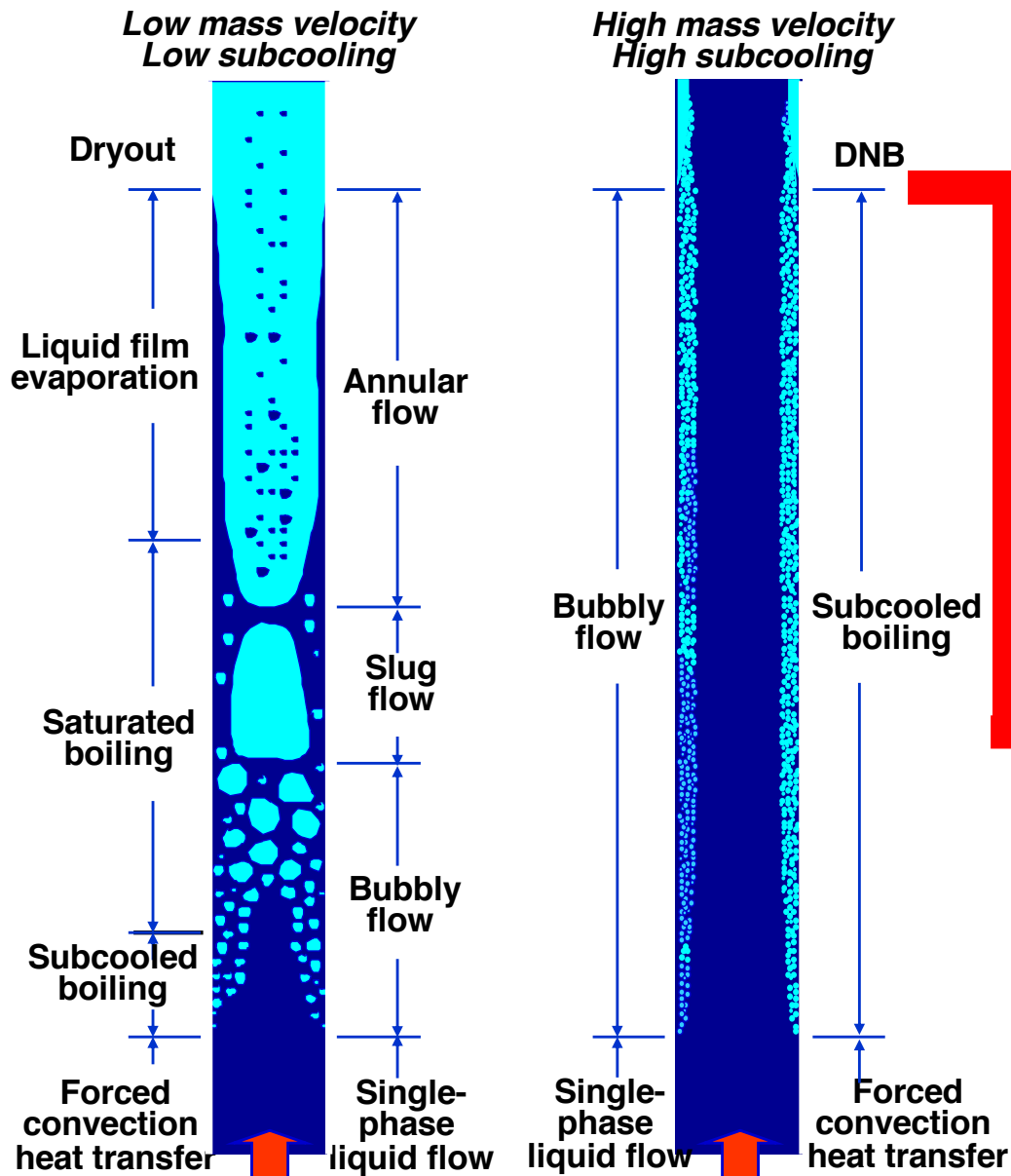


**Particle  
Accelerator  
Target**

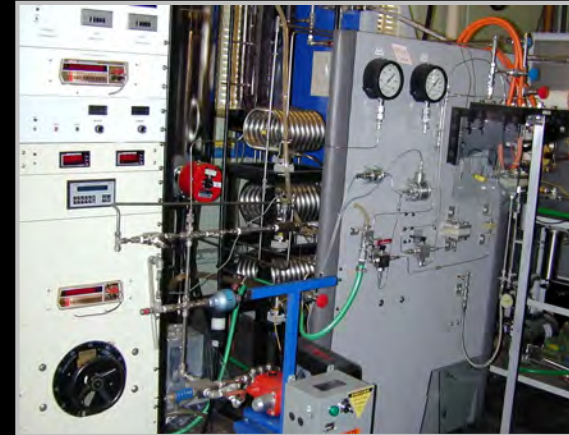


**MHD  
Electrode  
Walls**

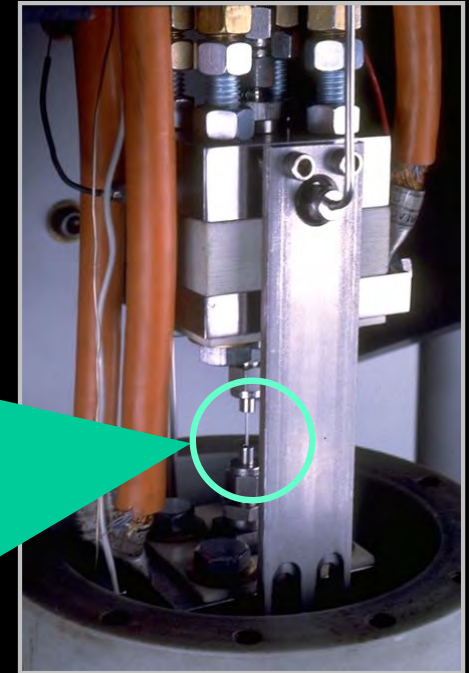
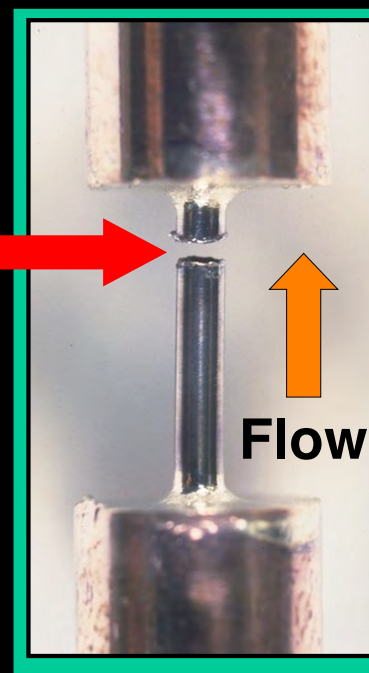
## Ultra-High-Flux Flow Boiling in Micro-Channels



**$CHF = 27,600 \text{ W/cm}^2$   
demonstrated experimentally**



$G = 120,000 \text{ kg/m}^2 \cdot \text{s}$   
 $D = 0.406 \text{ mm}$



Mudawar & Bowers (1999)



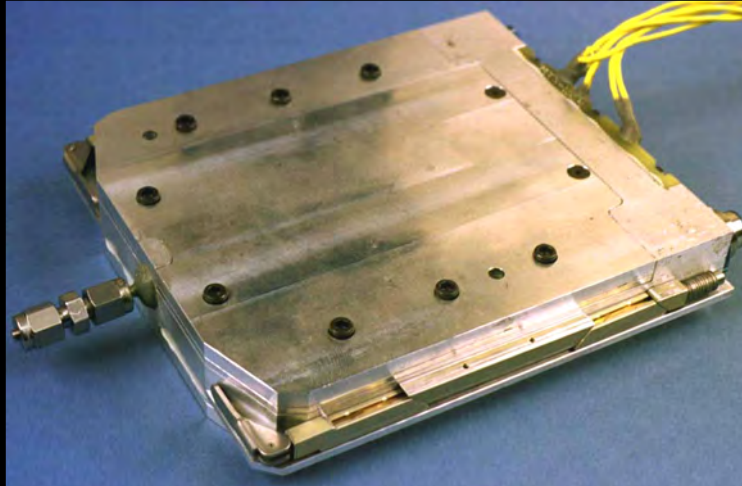
## Applications: Avionics

- 40 W cooling capacity for conventional 5.38 x 6.41 x 0.59 in<sup>3</sup> edge air-cooled module
- 200 W capacity for equivalent flow-through polyalphaolafin (PAO) module

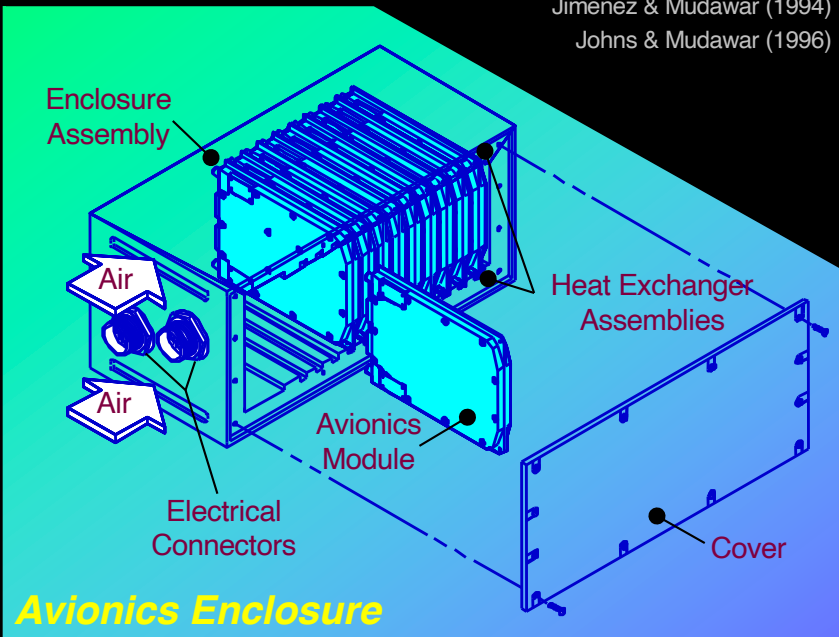
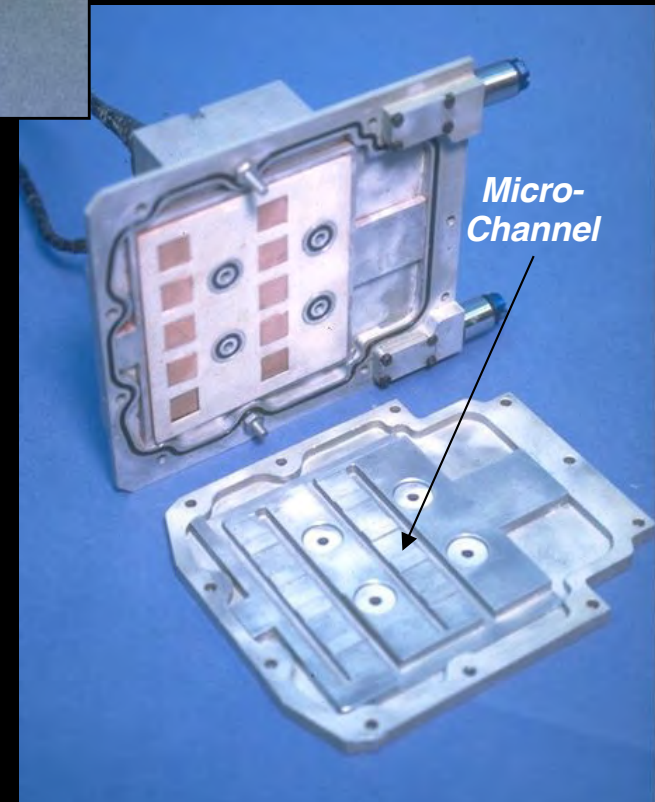


Mudawar, Jimenez & Morgan (1994)  
Jimenez & Mudawar (1994)  
Johns & Mudawar (1996)

### *BTPFL-C3* 12 kW Jet-Impingement Clamshell Module

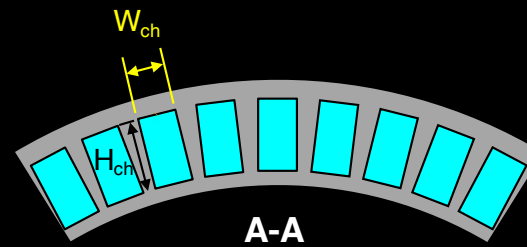
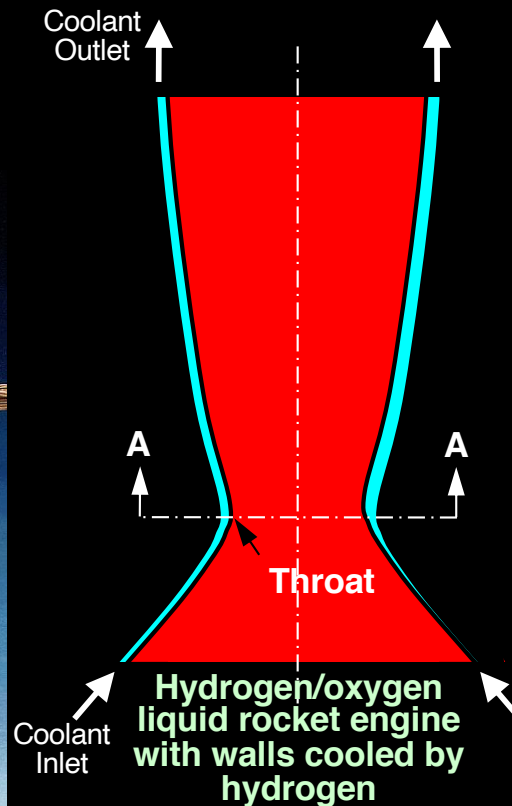
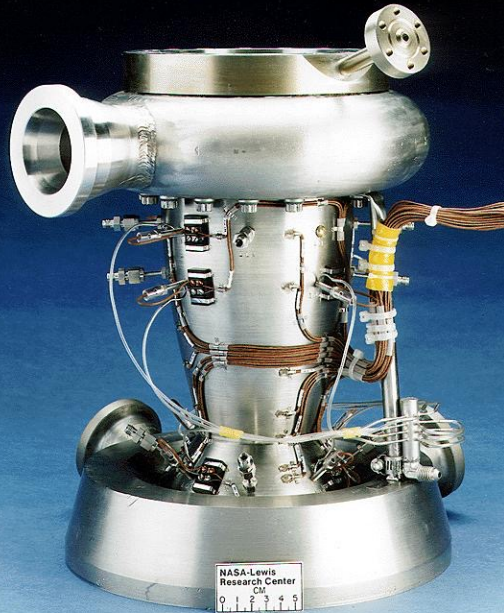


### *BTPFL-C2* 3 kW Micro-Channel Clamshell Module



# Applications: Boiling in Rocket Engines, Turbine Blades

**NASA rocket engine with high aspect ratio cooling channels**

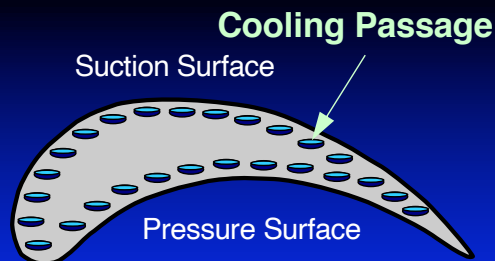
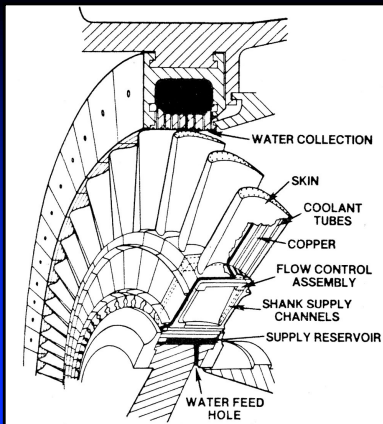


Conventional channels  
 $H_{ch}/W_{ch} \sim 2$

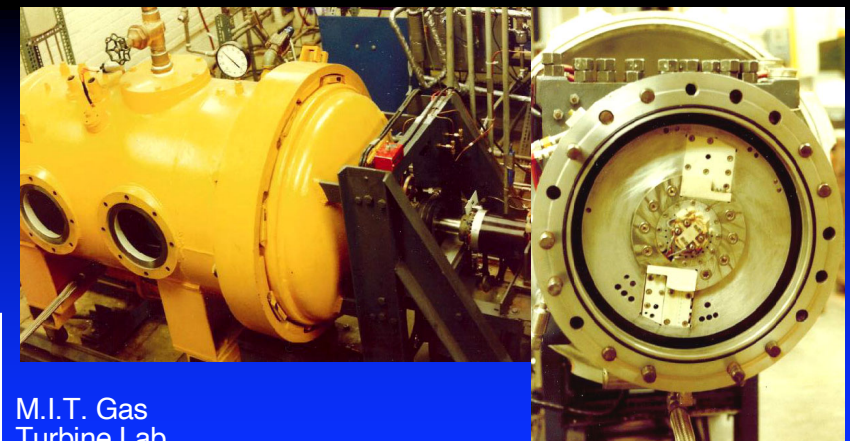
High aspect ratio channels  
 $H_{ch}/W_{ch} \sim 8$

$$\varepsilon = (2k_w / h W_w)$$

- Engine life doubled by decreasing wall temperature by 50-100° C
- Cooling performance in wall enhanced by using closely-spaced micro-channels separated by thin walls
- Depending on rocket engine type, boiling may or may not be desirable



Combined effects of centrifugal and Coriolis forces help achieve 4500 W/cm<sup>2</sup> inside blade



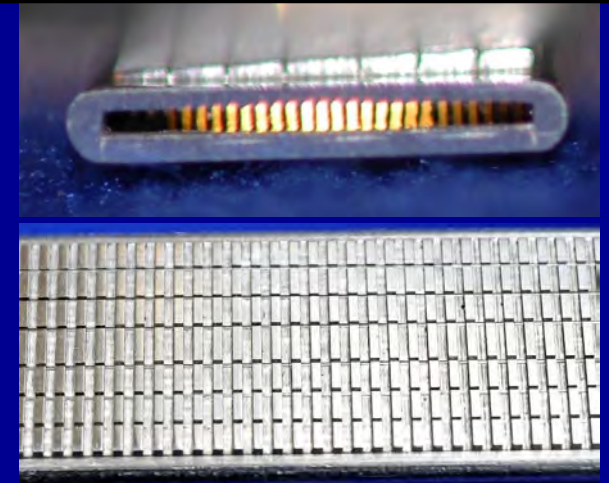
M.I.T. Gas Turbine Lab





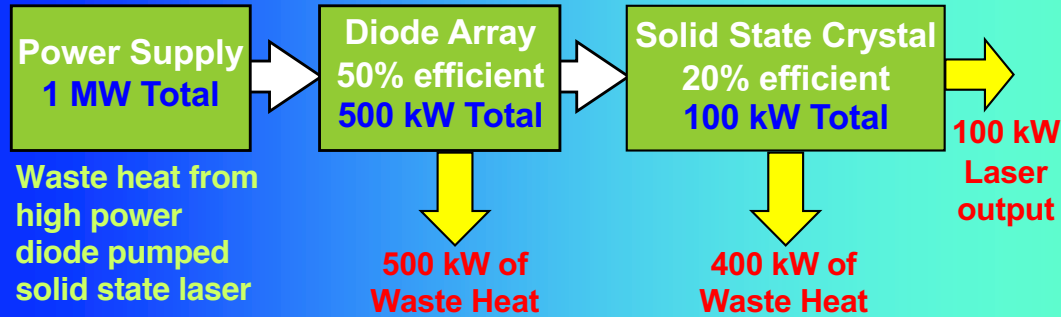


Nacke, Northcutt & Mudawar (2011)

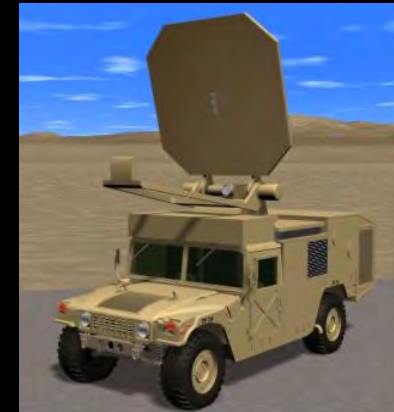


Micro-Channel Cross-Flow Air-to-Fuel (JP-8) Heat Exchanger

## High-Power Laser, Radar and Microwave Systems

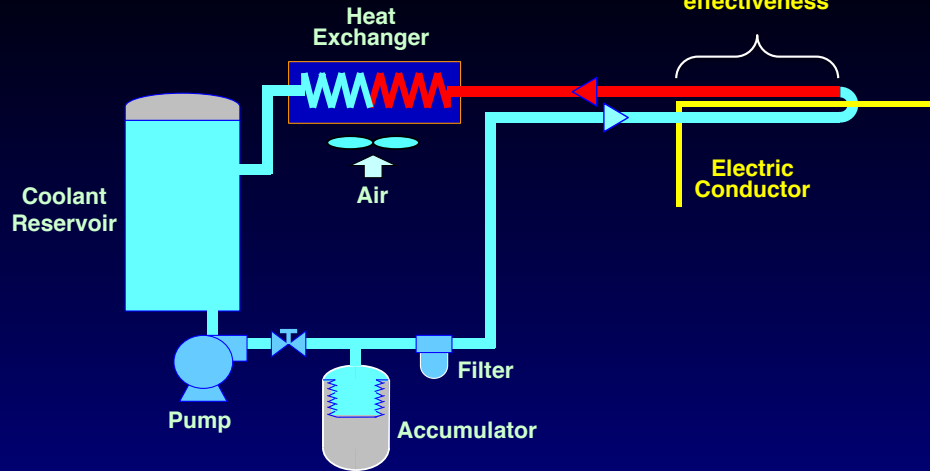


Mudawar (2001)

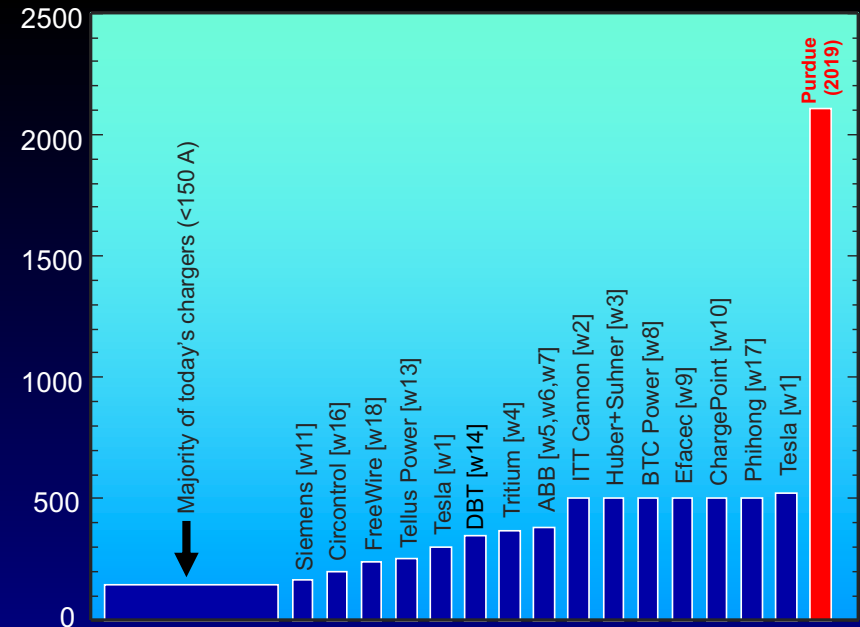


# Applications: 5-Minute Charging of Electric Vehicles

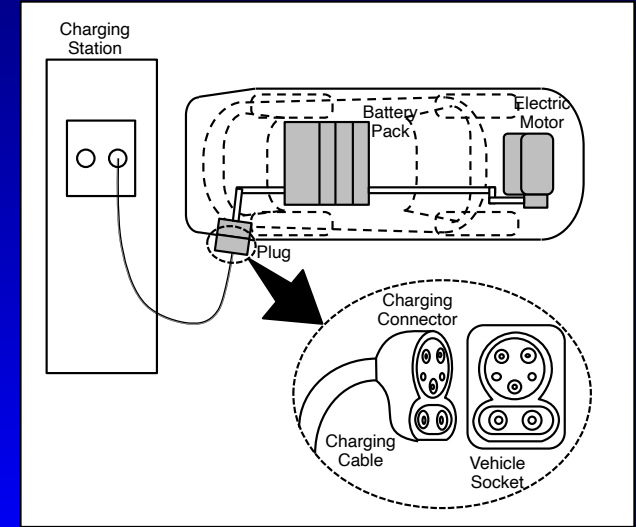
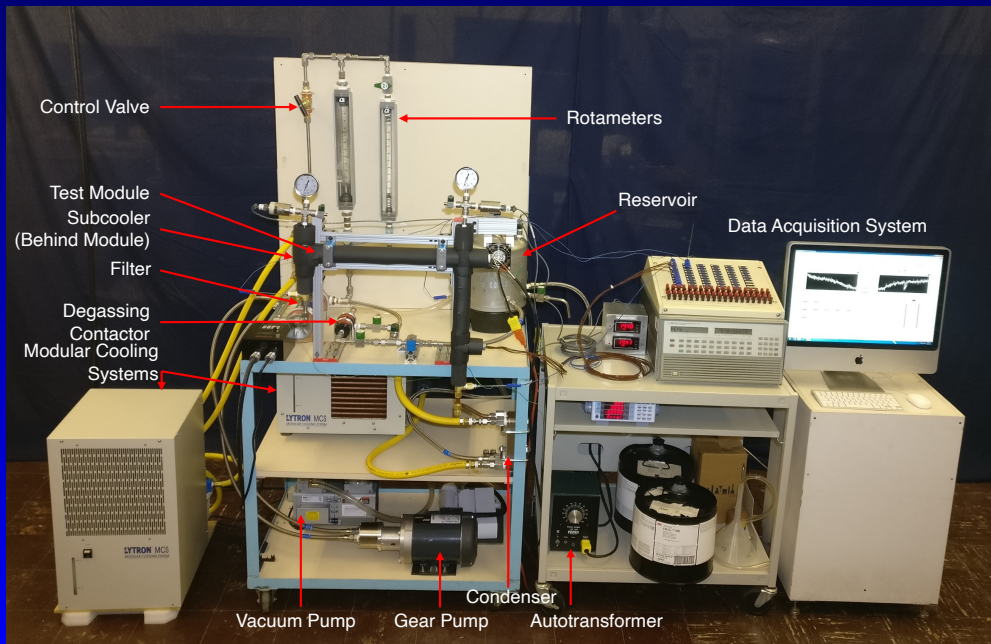
Subcooled flow boiling with flow passage enhancement to maximize heat transfer effectiveness



Maximum Current Through Charging Cable [A]

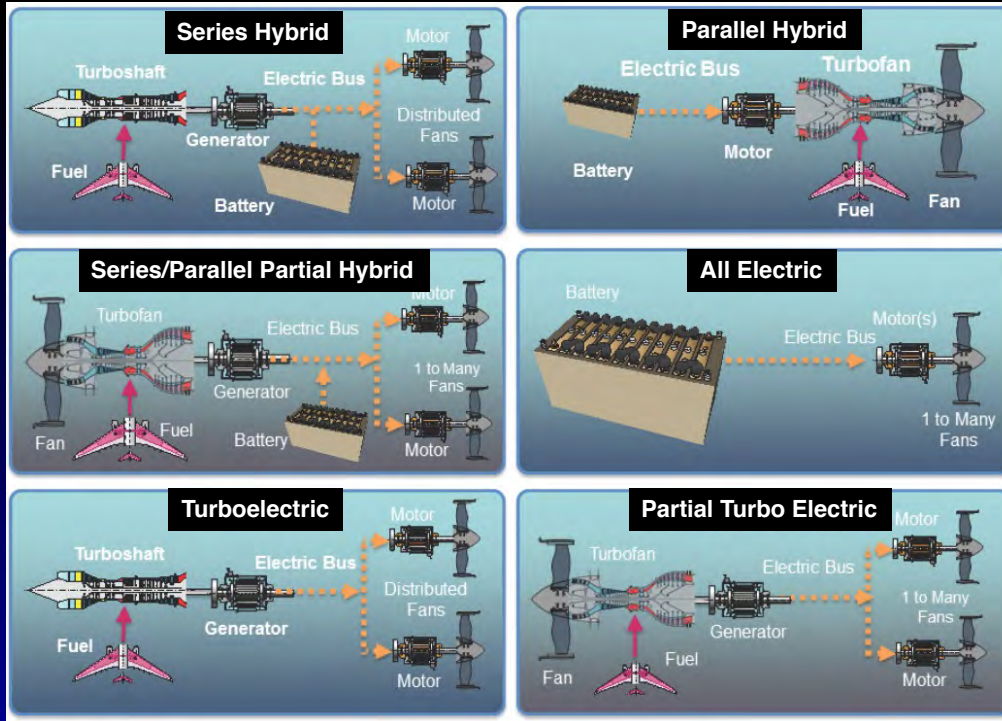


EV Chargers Available Worldwide



## Research Presentation





## Component Requirements for Hybrid Electric/All Electric Aircraft



Glenn Research Center at Lewis Field



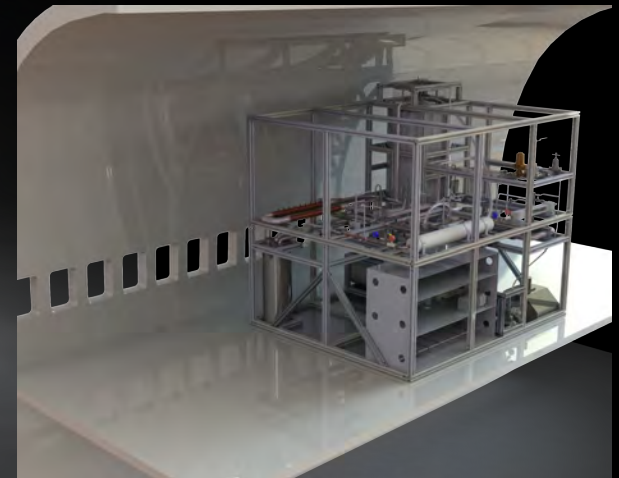
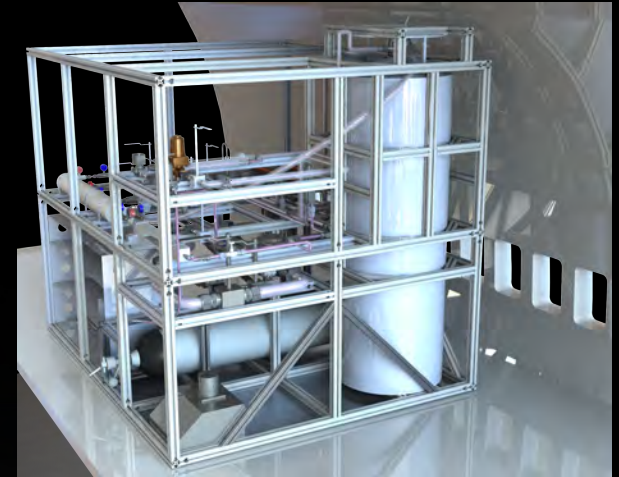
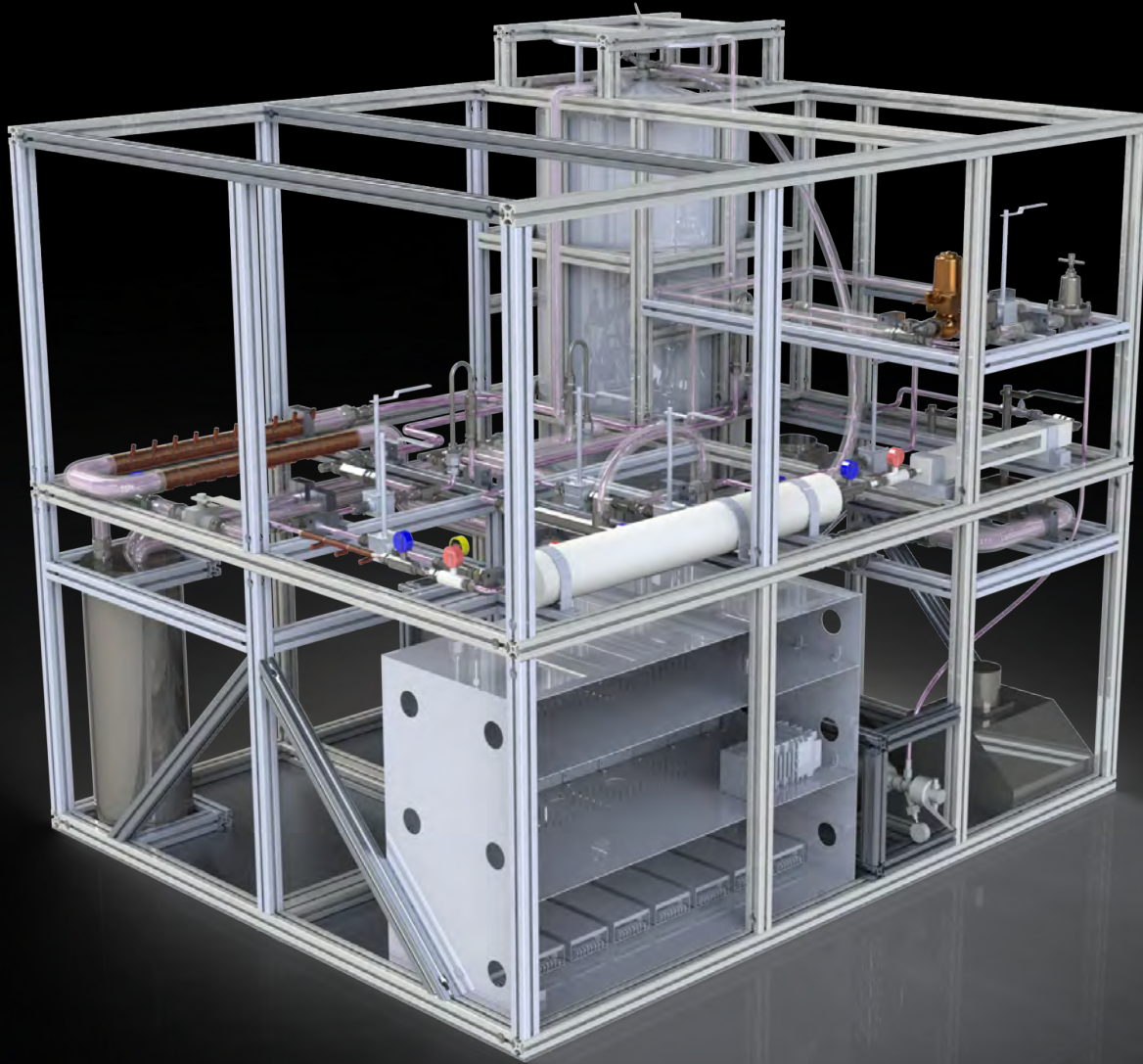
## Immediate Goals: High-flux liquid cooling of

- Power electronics
- Motor
- Power transmission cable
- Other components important to industry partners

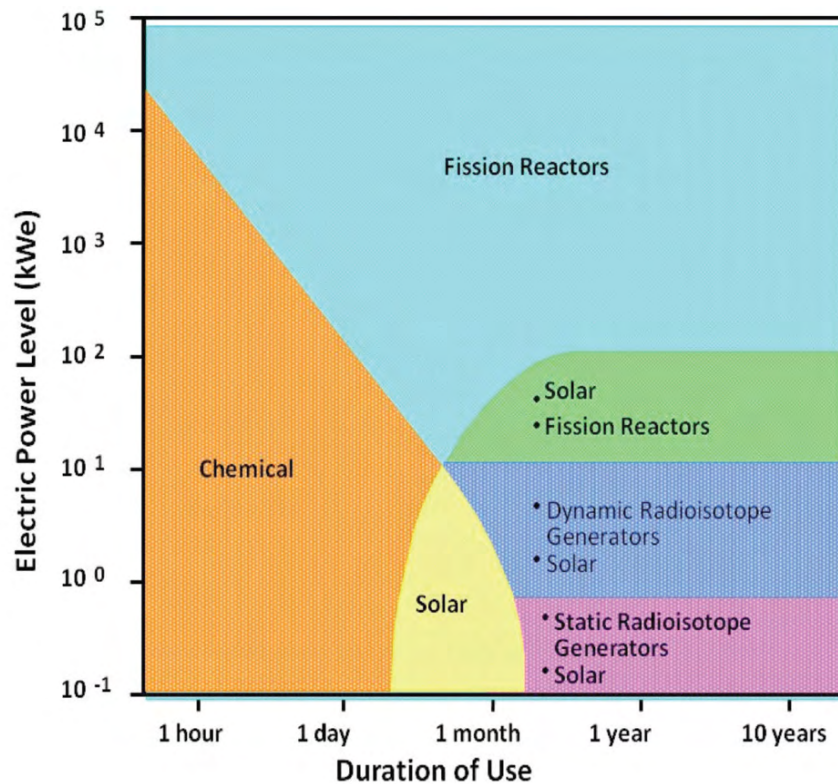


X-57 Maxwell



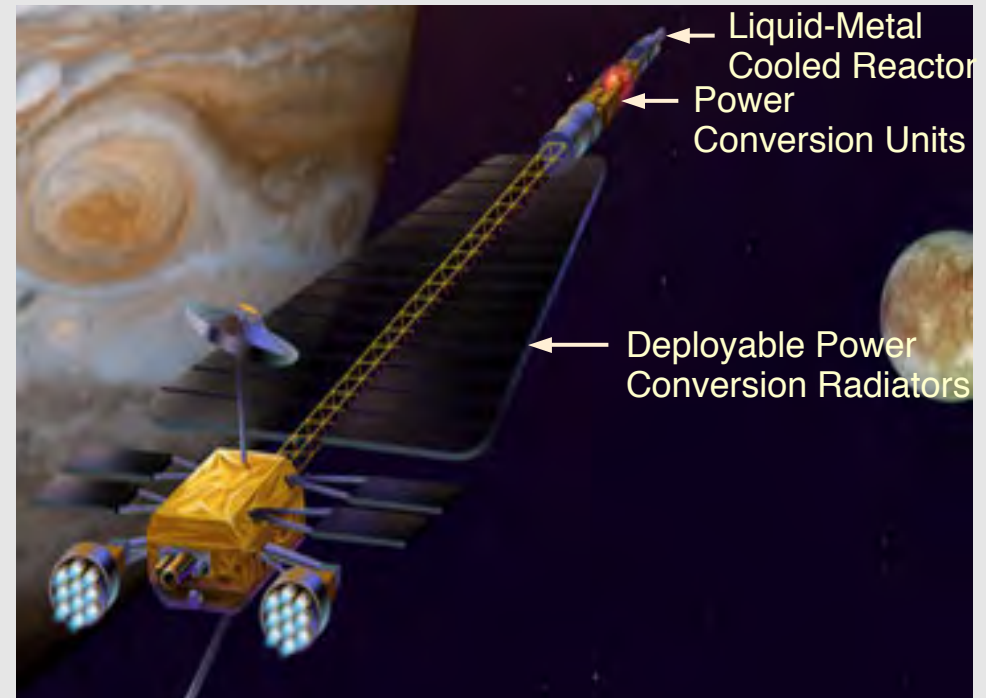


*Flow Boiling & Condensation Experiment (FBCE)  
for the International Space Station (ISS)*



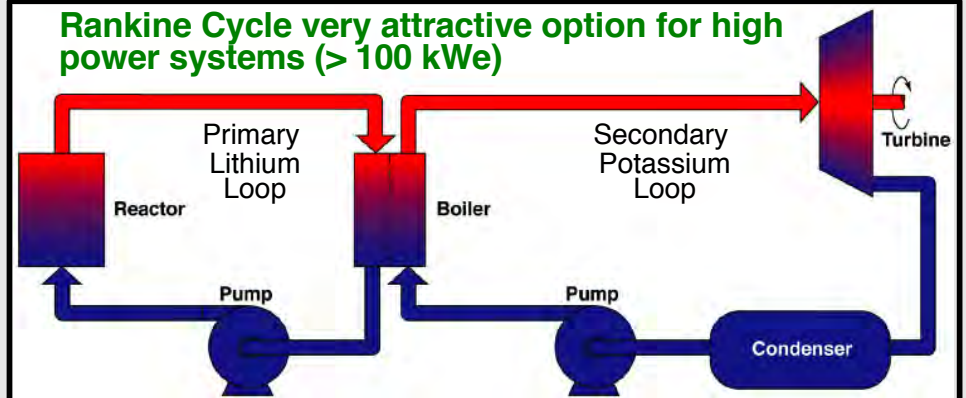
**Project Prometheus:**

Developing means to efficiently power advanced spacecraft for Solar System exploration



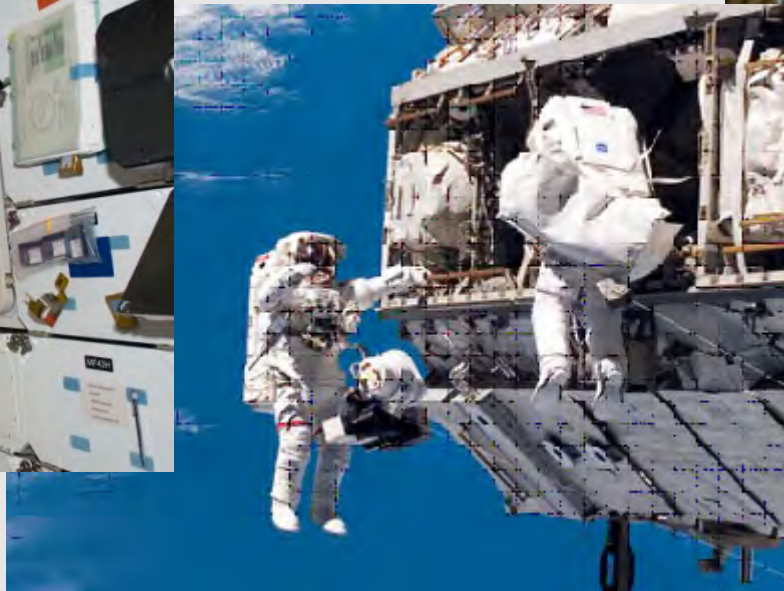
	Present - 2016	2017 - 2022	2023 - 2028
Fission 1.0– 100 kWe	Non-nuclear System Demo	Nuclear Criticality Demo	kWe-class Science
Fission > 100 kWe	1 MWe Design	1 MWe Components	50 kWe Space/Surf Sys.
Exploration Missions		1 MWe Integrated	kWe-class Science
Science Missions		MWe-class Exploration	MWe-class Exploration
		Flexible Path Robotic	Flexible Path Human
		Flagship #1	Flagship #2

**Rankine Cycle very attractive option for high power systems (> 100 kWe)**



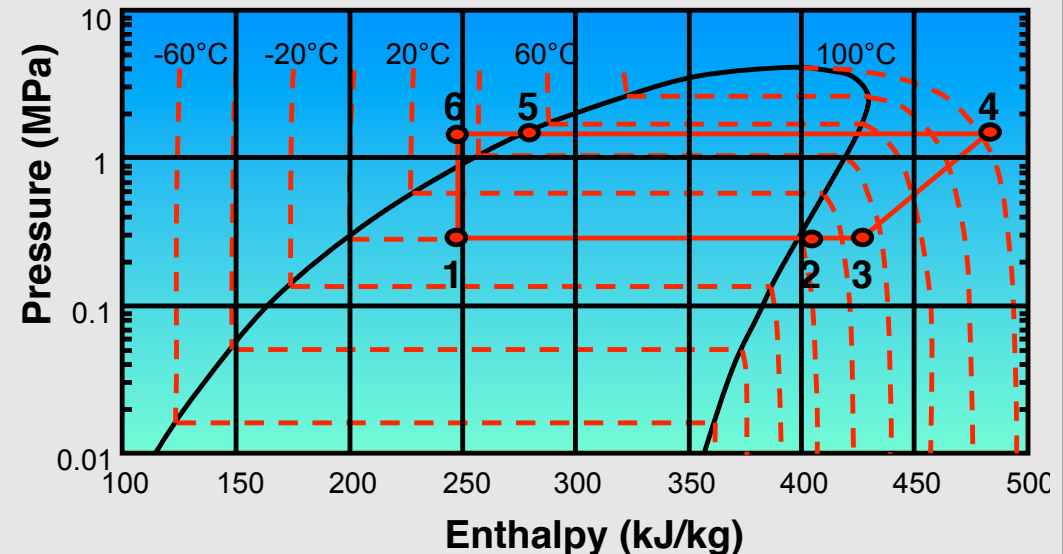
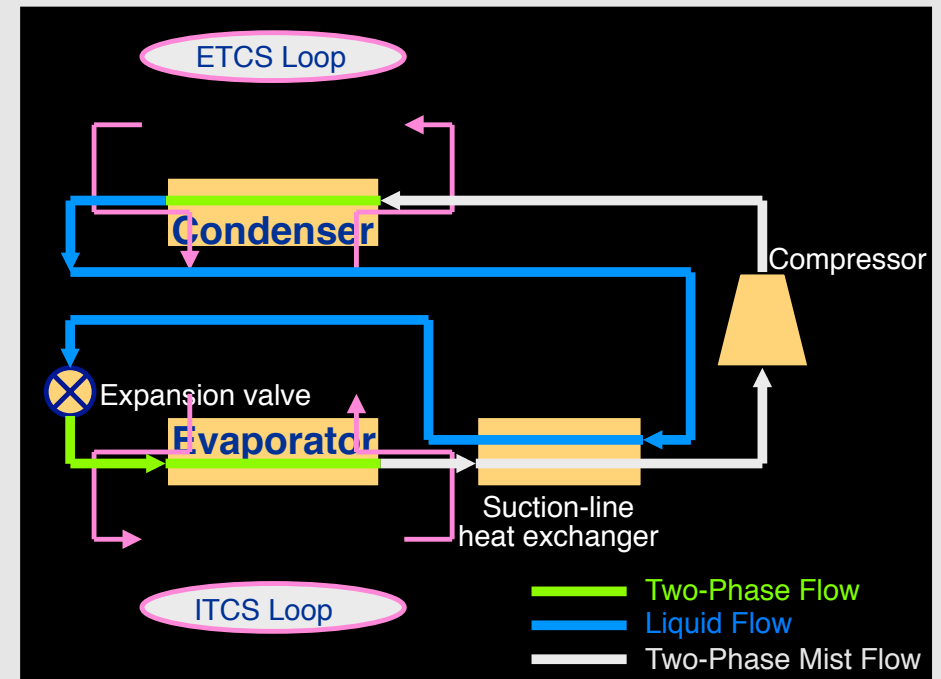
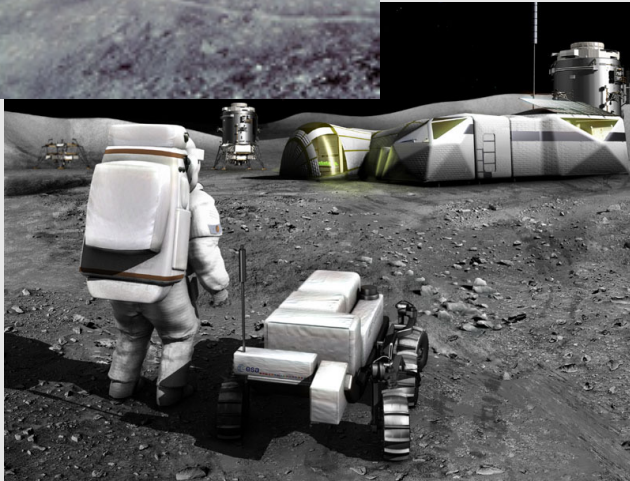
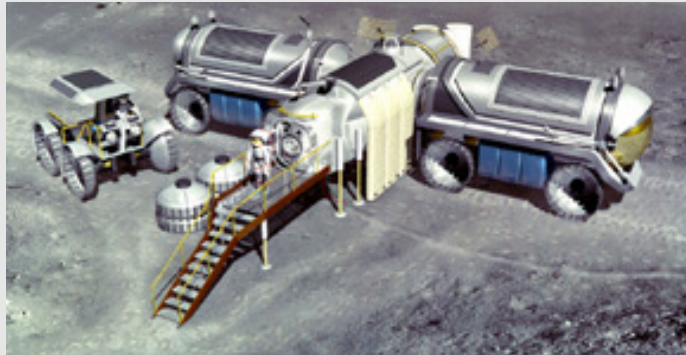


- Thermal management systems responsible for controlling temperature and humidity using Thermal Control System (TCS) consisting of Heat Acquisition, Heat Transport and Heat Rejection hardware
- Refrigerator/freezer components provide cooling for science experiments and food storage
- Advanced water recovery systems transfer crew and system wastewater into potable water for crew and system reuse

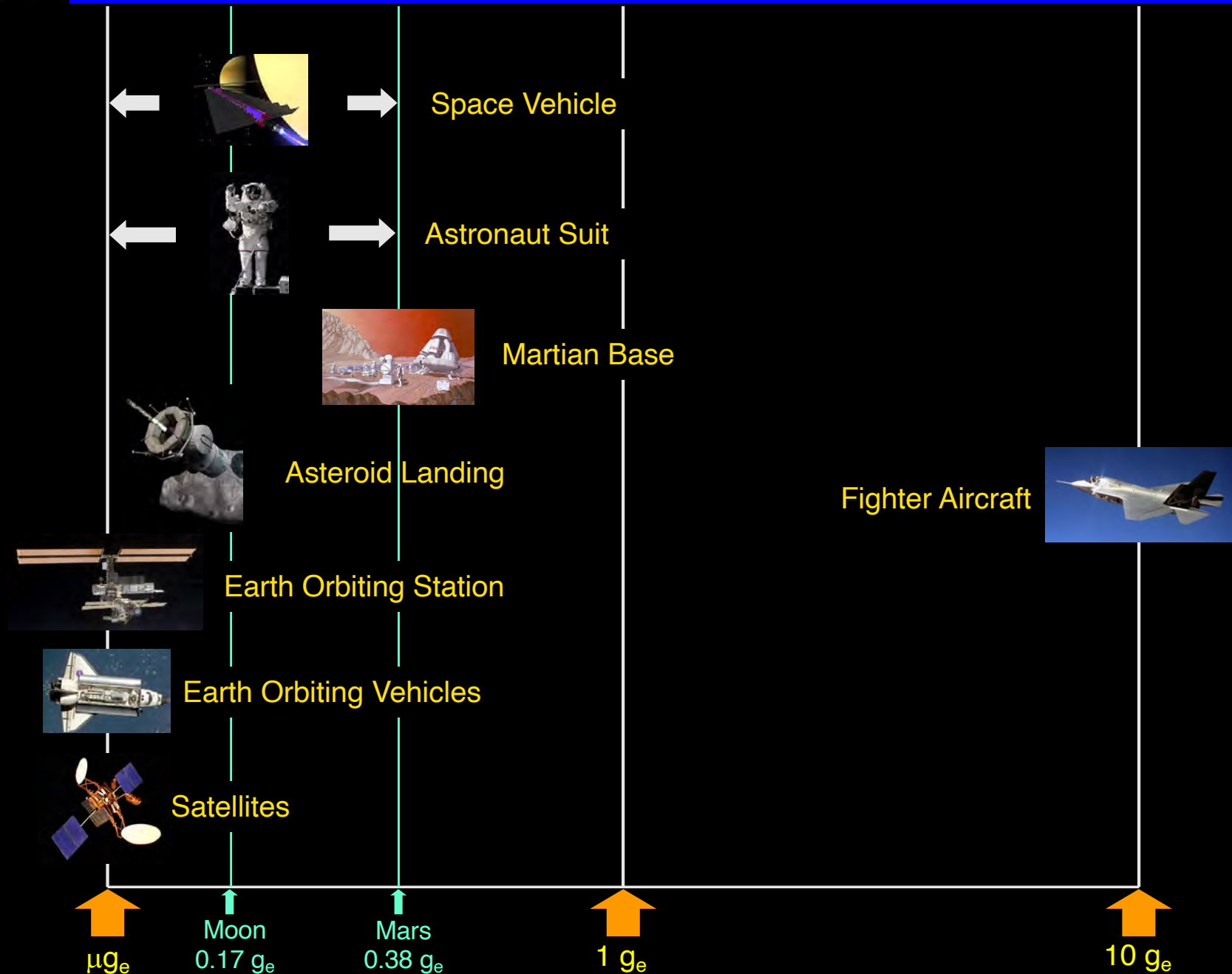


**NASA Johnson Space Center (JSC) Ground based Solar Heat Pump (SHP):**

- Precursor to R134a vapor compression heat pump for future space vehicles and planetary bases
- Absorbs 5 to 15 kW
- Heat to evaporator provided by single-phase (liquid) Internal Thermal Control Cycle (ITCS)
- Heat from condenser by single-phase (liquid) External Thermal Control Cycle (ETCS)
- Evaporator and Condenser are SWEP Copper Brazed Plate (CBP) heat exchangers



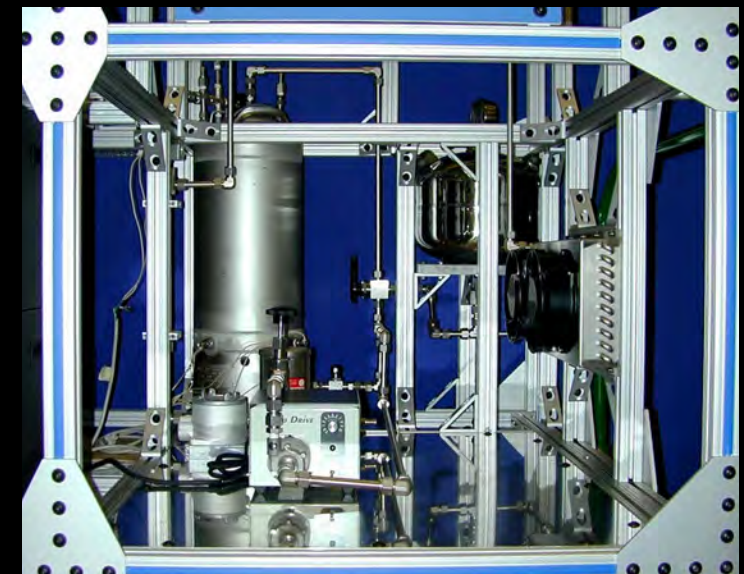
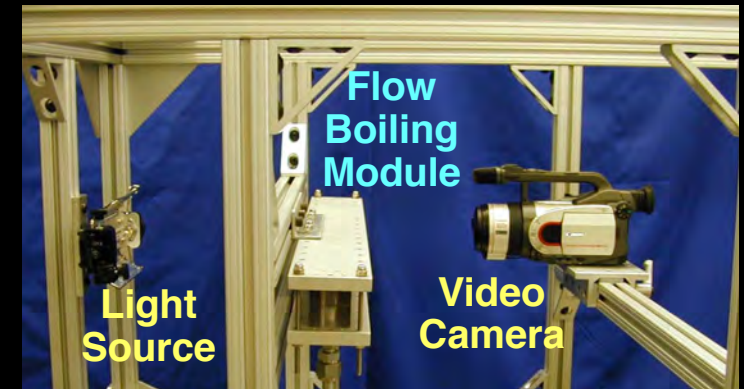




## *Purdue Microgravity Flow Boiling Apparatus*



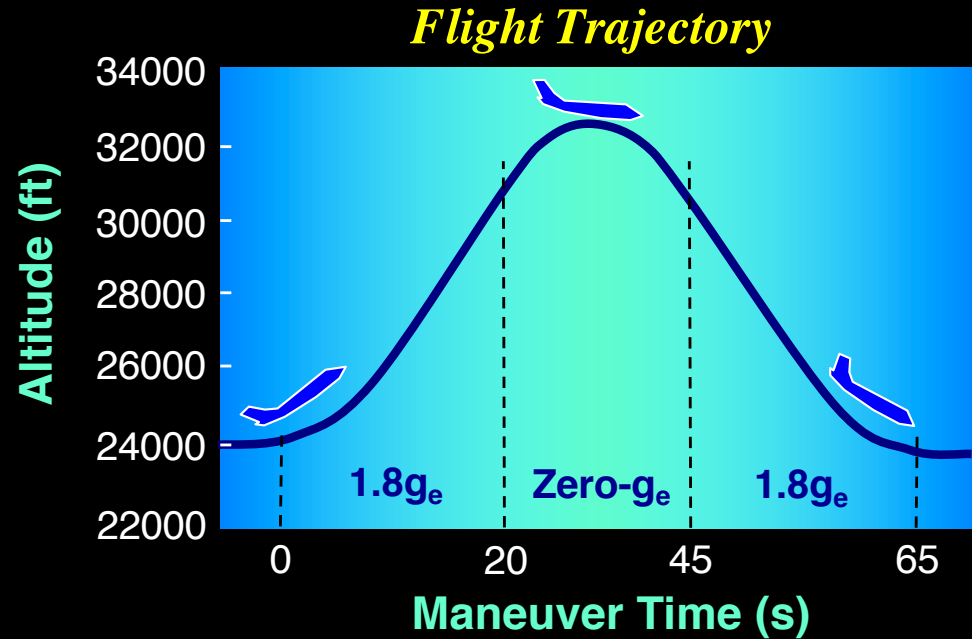
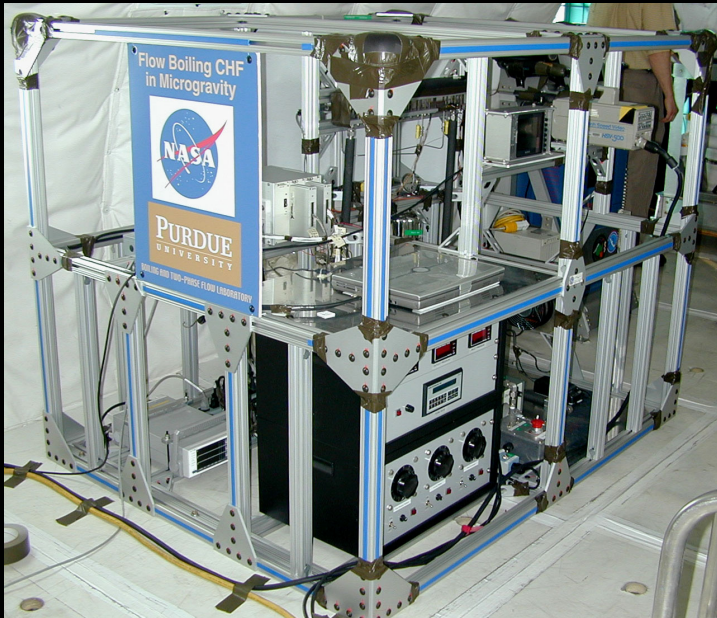
**Flight Rack**



**Two-Phase Flow Loop**



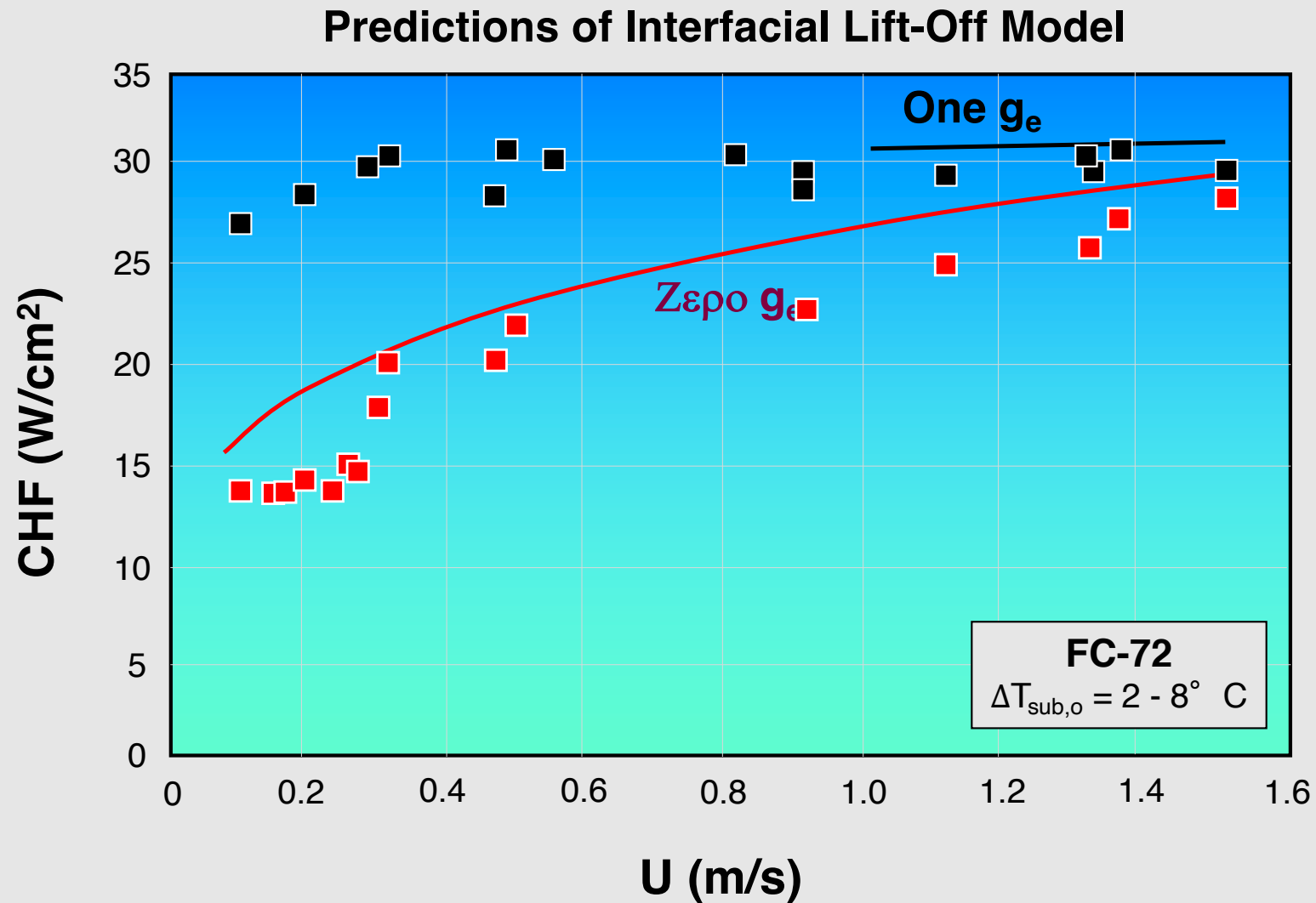
KC-135 Tests performed at NASA Glenn Research Center

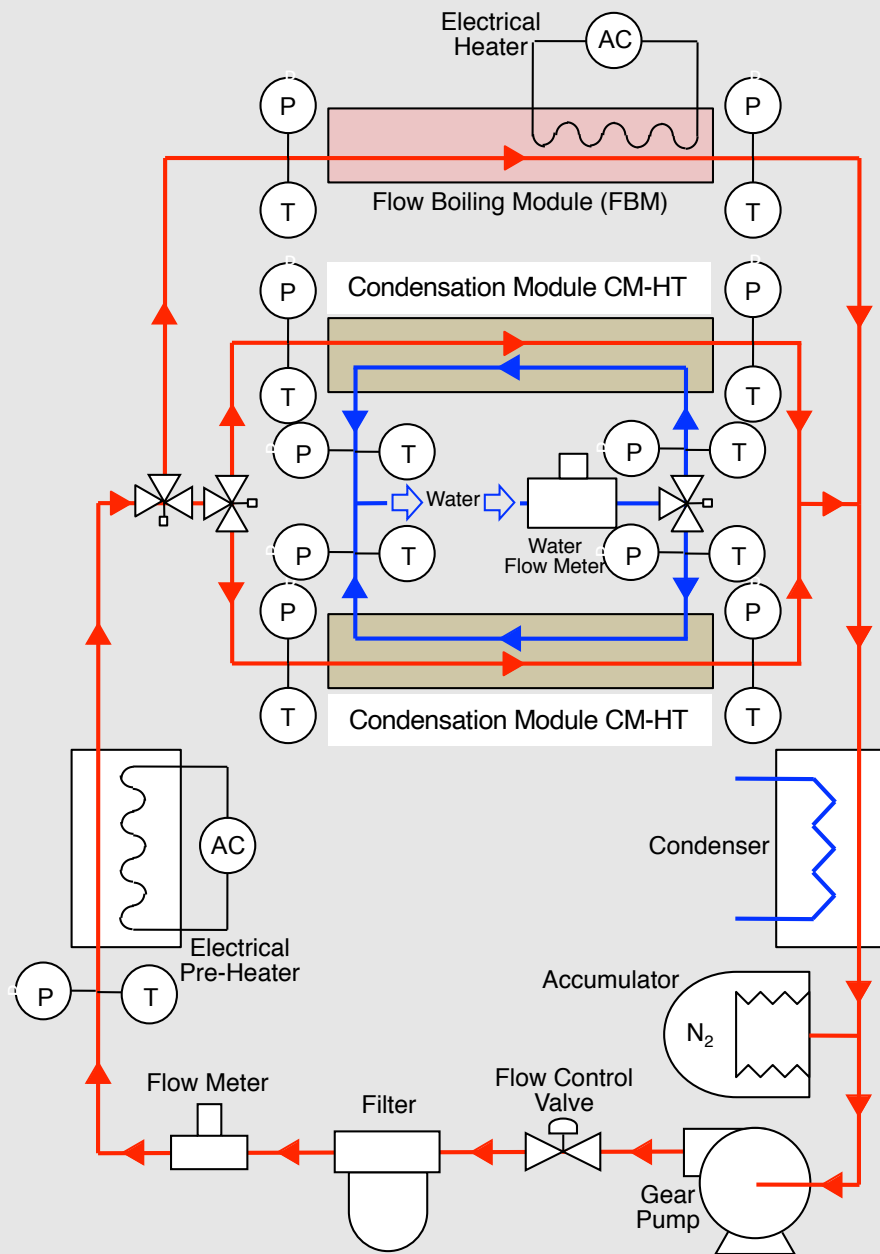


Phase Change Photo Library  
(Mudawar, 1984 - 2014)









**Consists of:**

- nPFH sub-loop
- Water sub-loop

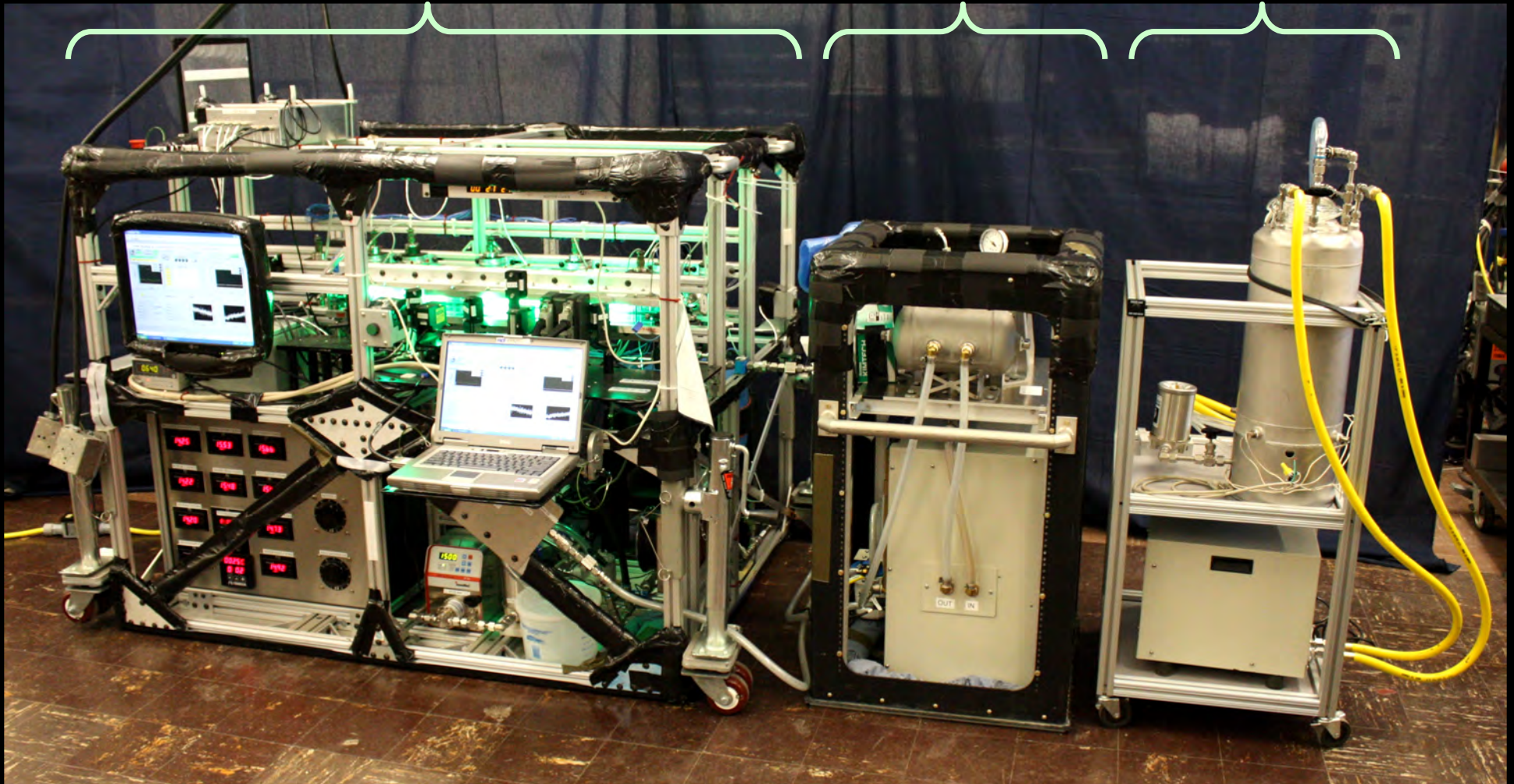
**Contains three test modules:**

- Flow Boiling Module (FBM)
- Condensation Module CM-HT for heat transfer measurements
- Condensation Module CM-FV for flow visualization

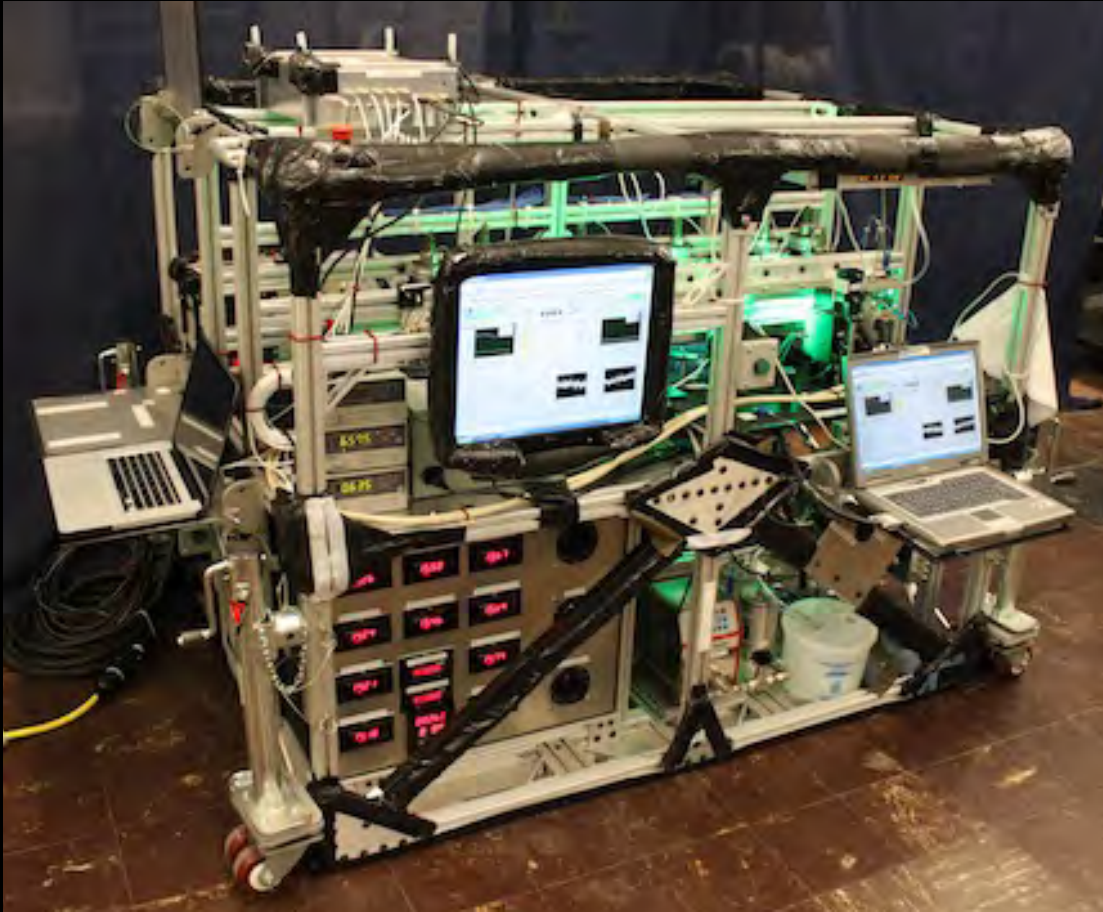
Condensation  
Rig

Water  
Conditioning  
Rig

Deaeration  
Rig





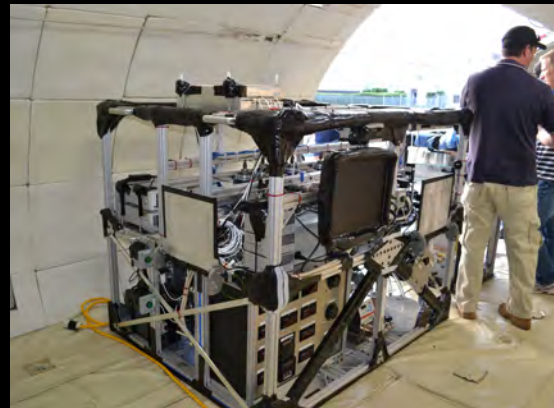


**CM-HT: Condensation  
Module for Heat Transfer  
Measurements**



**CM-FV: Condensation  
Module for Flow  
Visualization**





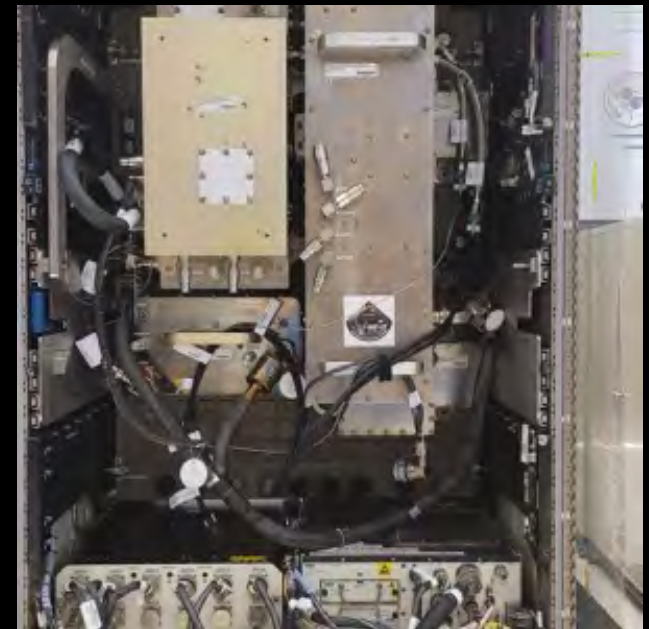
## Flight Campaigns:

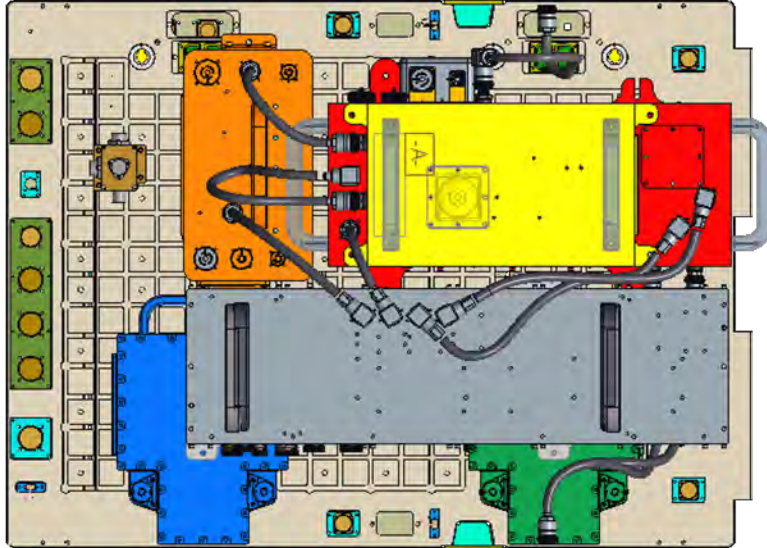
- May 2012
- August 2012









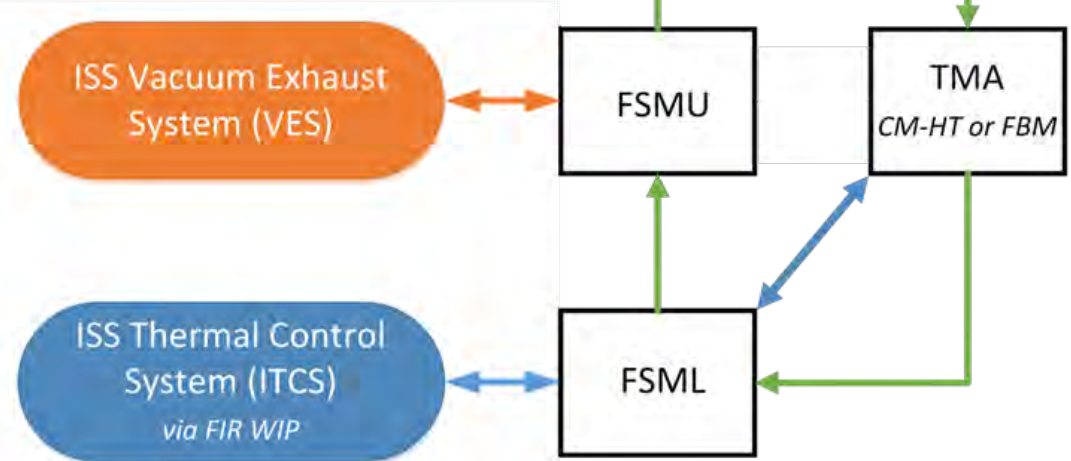


## FBCE Modules:

- **BHM** – Bulk Heater Module
- **FSMU** – Fluids System Module - Upper
- **FSML** – Fluids System Module - Lower
- **RDAQM 1** – Remote Data Acquisition Module 1
- **RDAQM 2** – Remote Data Acquisition Module 2
- **TMA** – Test Module Assembly (1 of 2 installed):
  - **FBM** – Flow Boiling Module
  - **CM-HT** – Condensation Module - Heat Transfer

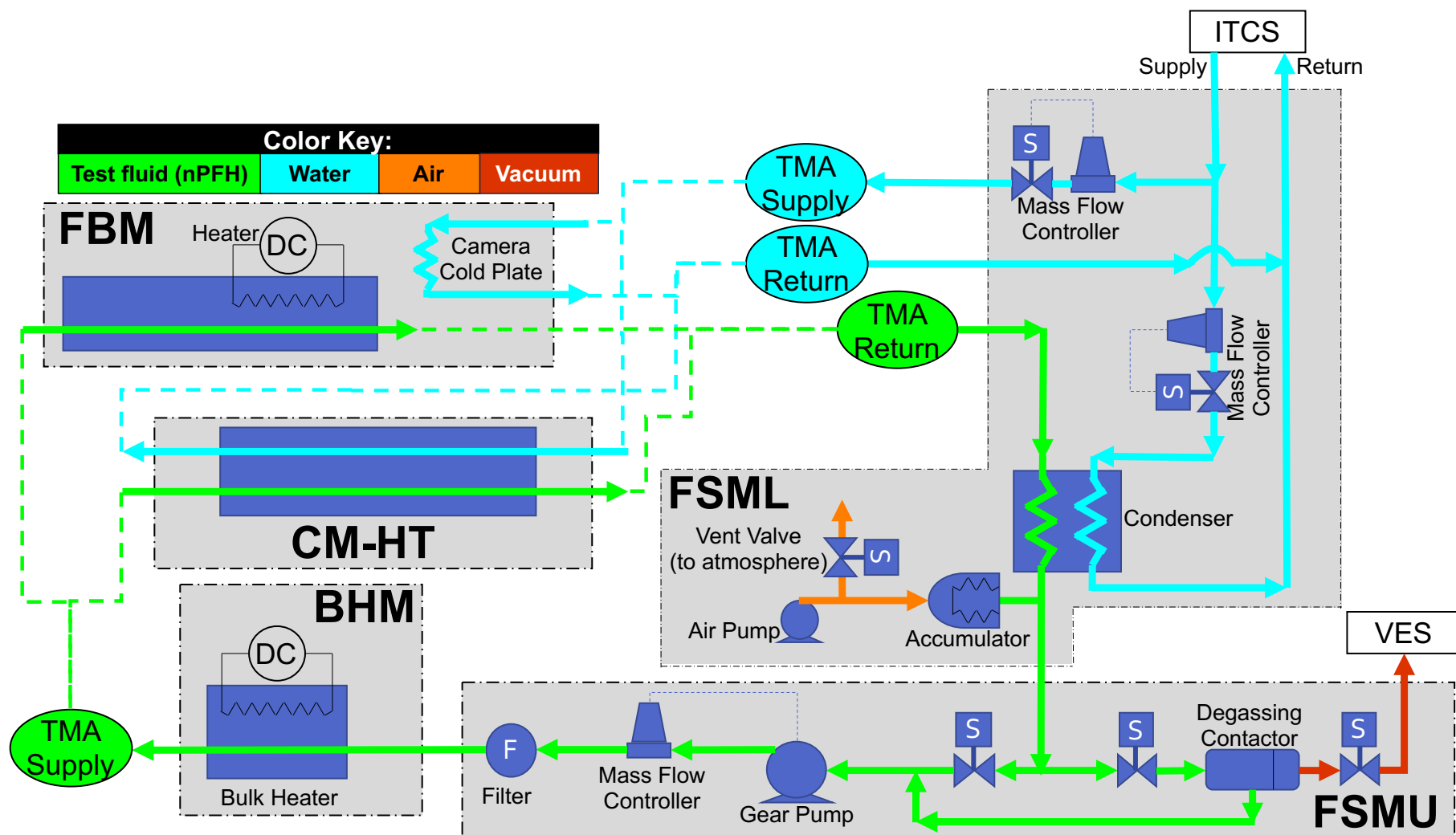
### LEGEND:

- Coolant Water
- Vacuum
- nPFH



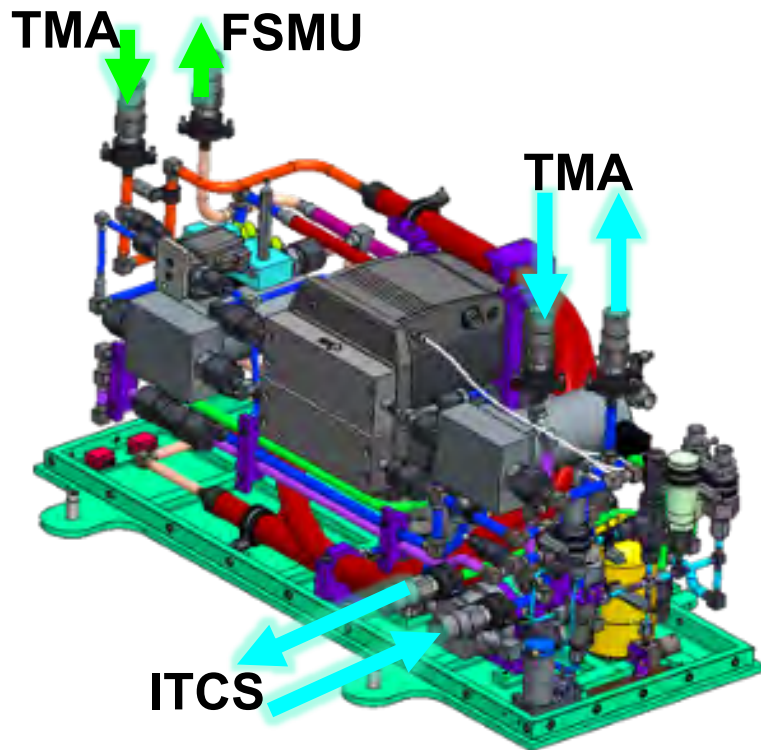
## FIR Provided Hardware:

- SAMS – Space Acceleration Measurement System
- CCU – Confocal Control Unit (on back of rack)
- IPSU-CL – Imaging Processing Storage Unit – Camera Link (on back of rack)

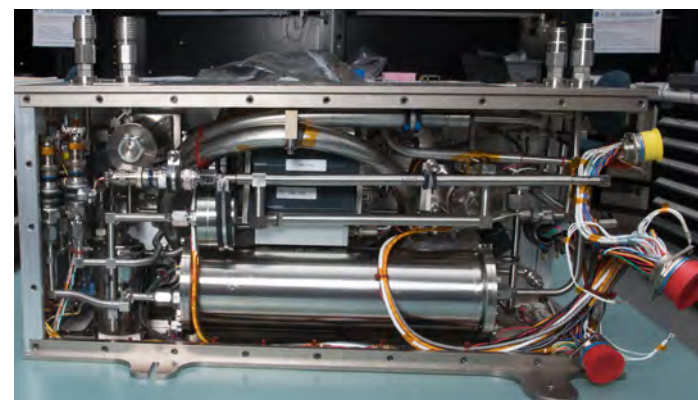
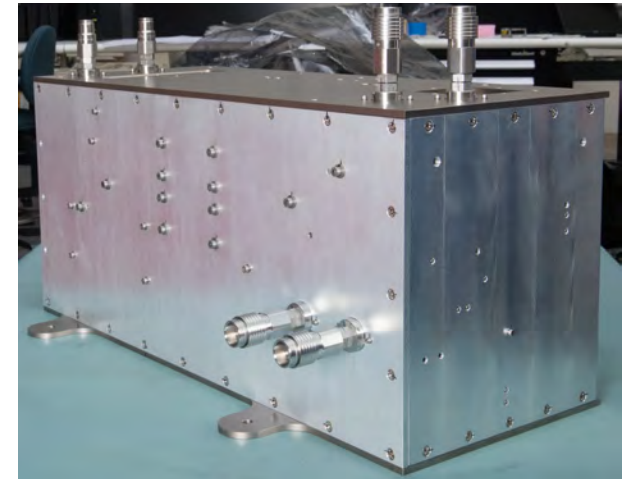
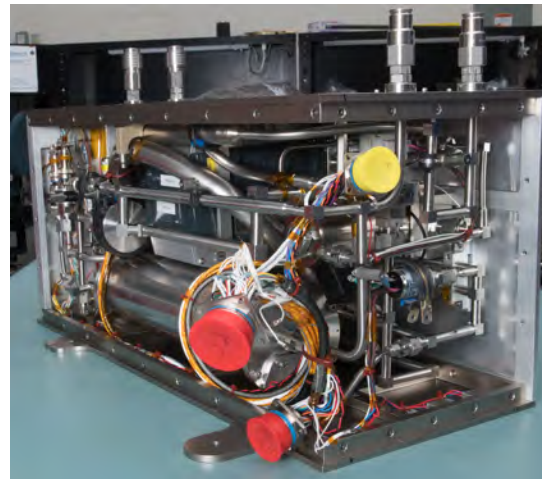




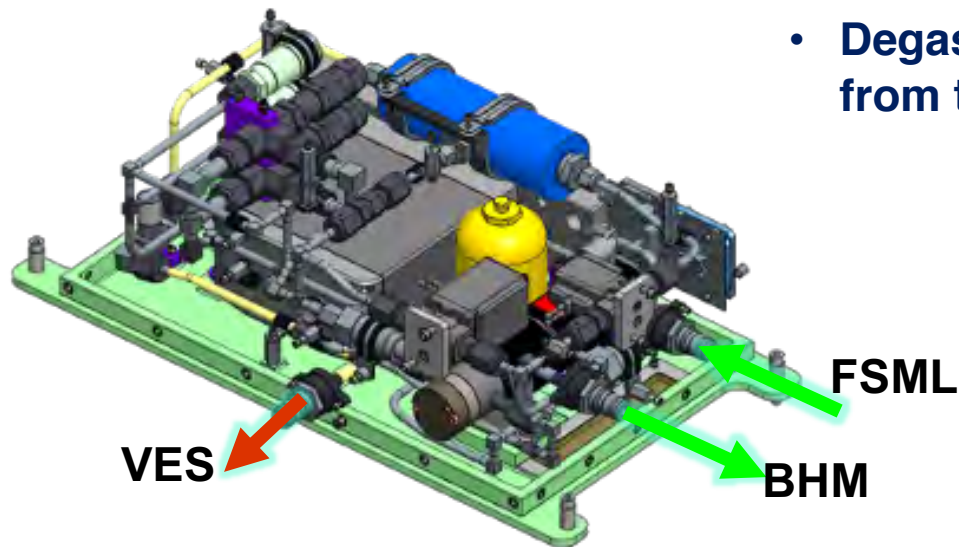
- Provides primary cooling for the test fluid exiting the test section, and the test section itself
- System pressure set by pressurizing or venting the air-side of a bellows accumulator



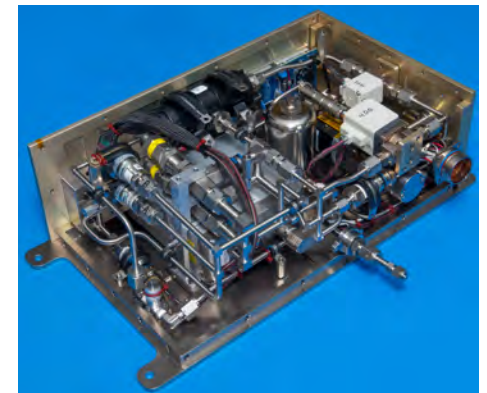
Color Key:		
Test fluid (nPFH)	Water	Air



- Mass flow controller drives a gear pump to provide flow throughout the closed loop system
- Multiple controls in place to prevent over-pressurization
- Degassing contactor removes dissolved gases from test fluid when membrane exposed to vacuum

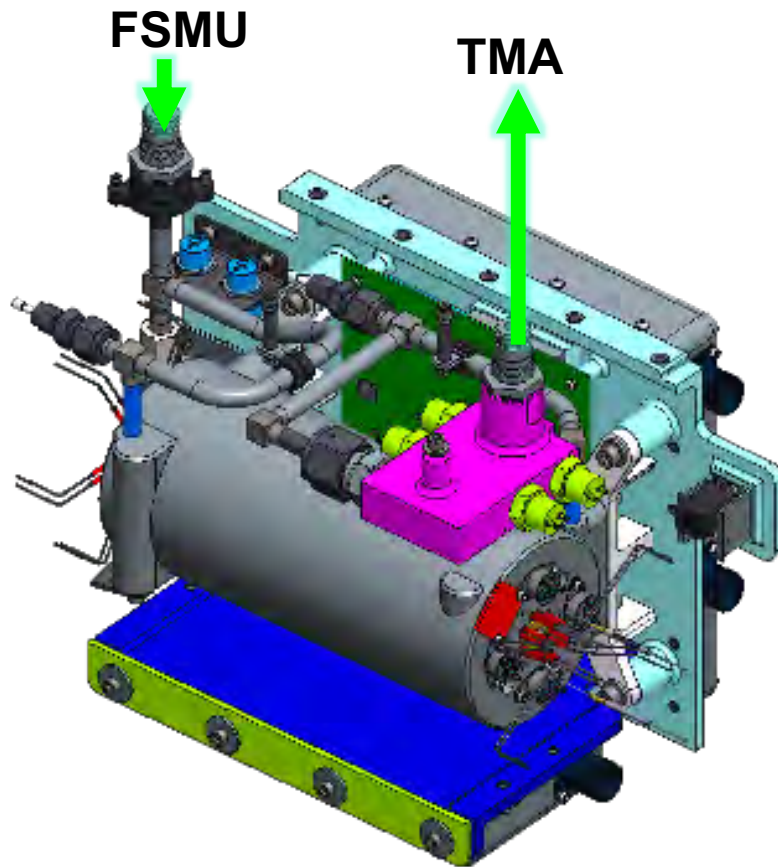


Color Key:	
Test fluid (nPFH)	Vacuum

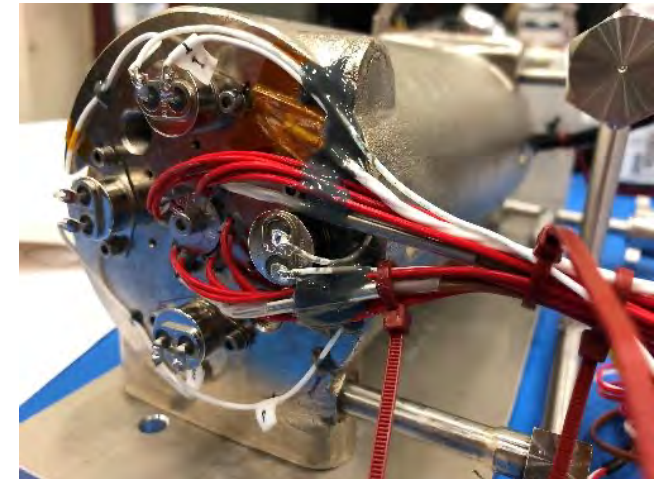




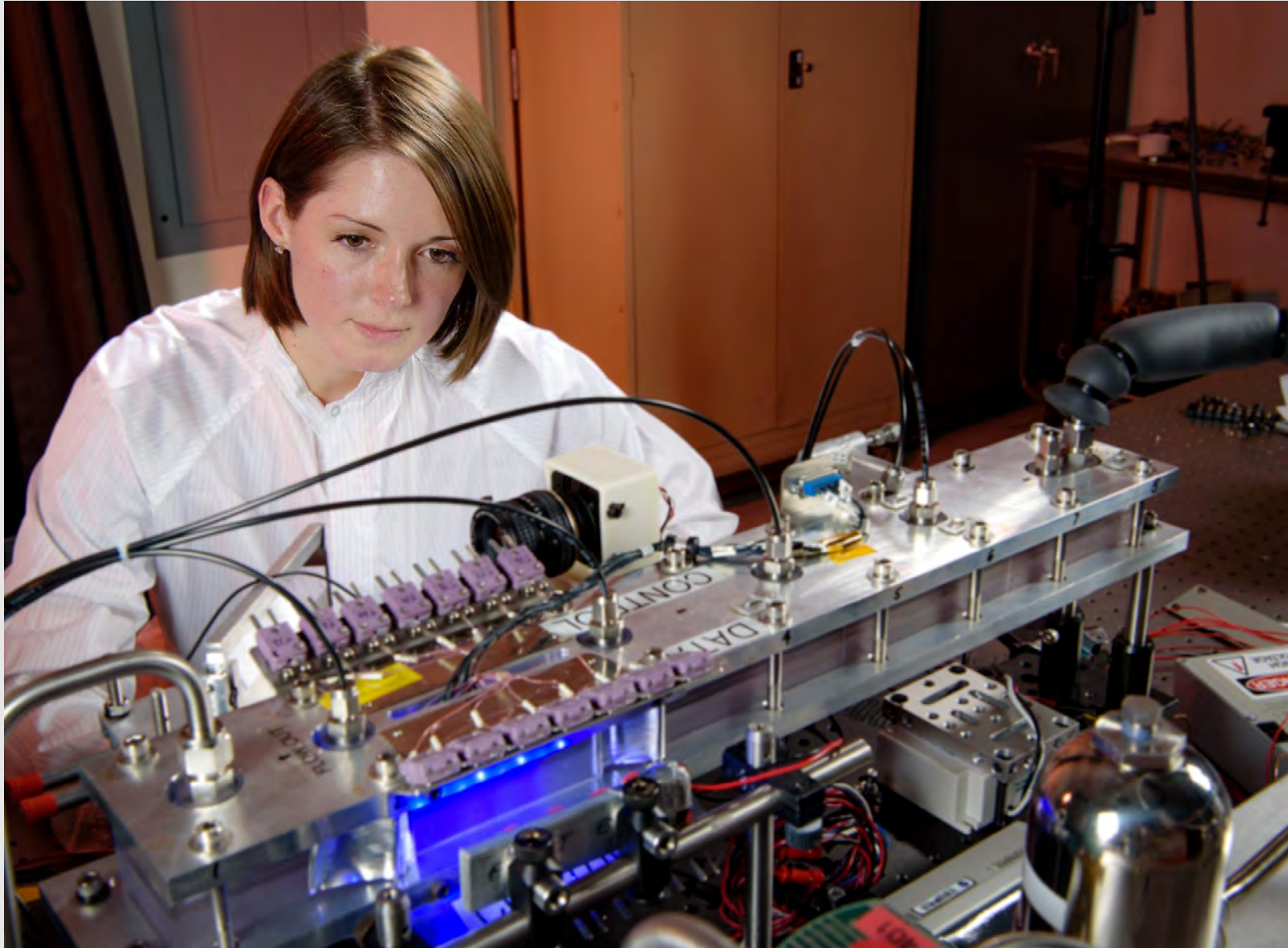
- Primary source of heating to condition test fluid to required test section inlet conditions
- Three 120V primary heaters and three 28V booster heaters can be operated at any time, with backup heaters available



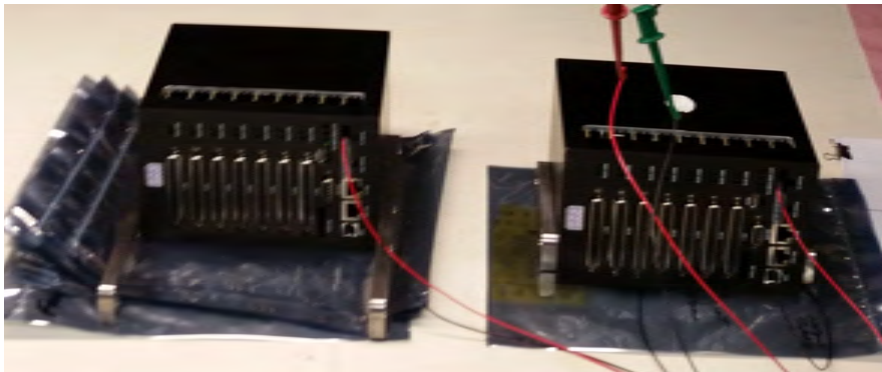
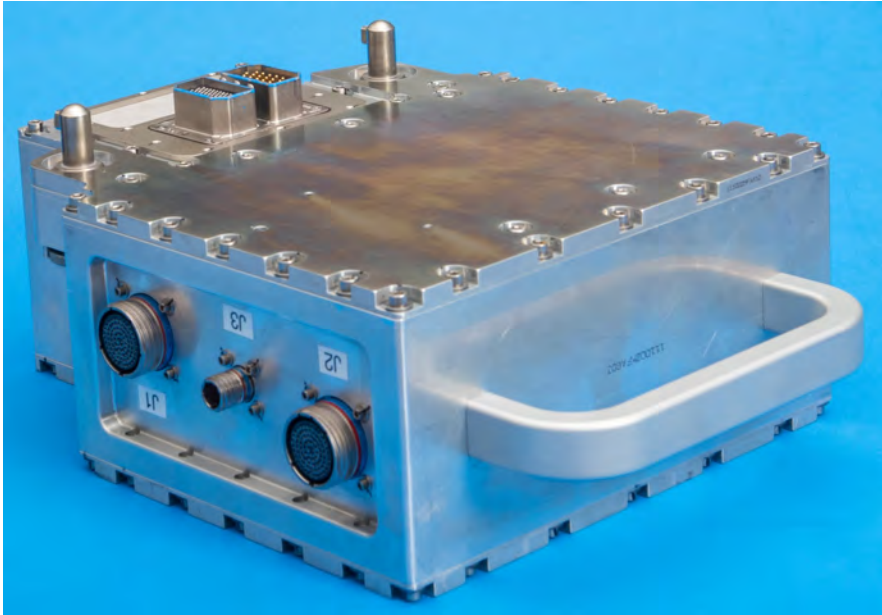
**Color Key:**  
Test fluid (nPFH)





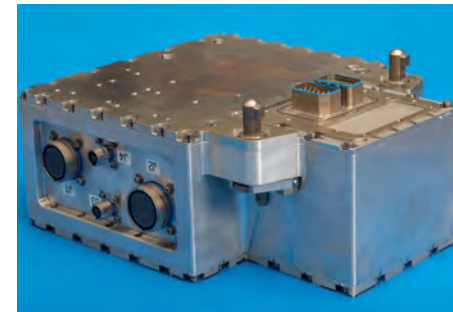
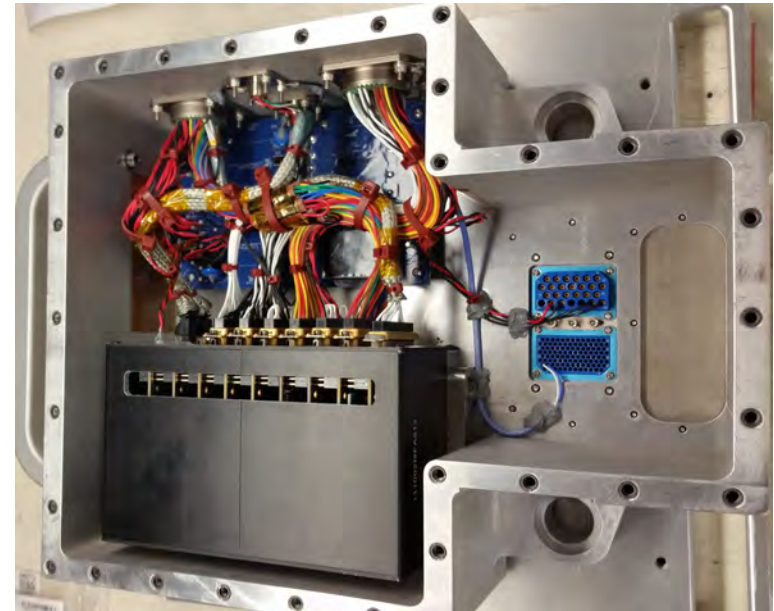


## **Remote Data Acquisition Module 1 (RDAQM1)**



**UEI Data Cubes (Thermocouple Signal Conditioning)**

## **Remote Data Acquisition Module 2 (RDAQM2)**



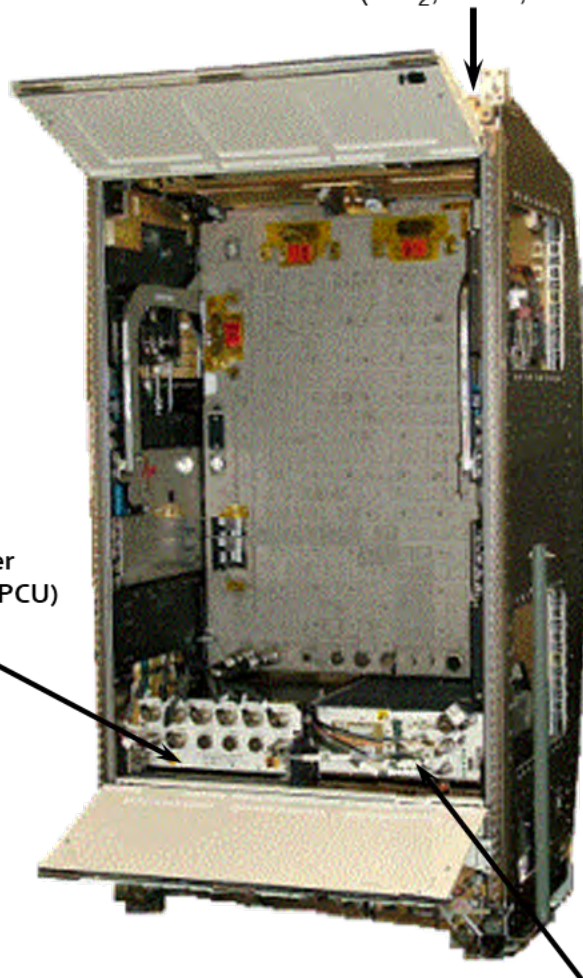
**UEI Data Cube and Custom Sensor Supply Printed  
Circuit Board  
(Signal Conditioning and Power Distribution)**



**Environmental Control (ECS)**

- Air Thermal Control
- Fire Detection & Suppression
- Water Thermal Control
- Gas Interfaces (GN<sub>2</sub>, VES, VRS)

Electrical Power  
Control Unit (EPCU)



Gas Interface  
Panel (GIP)

Water Interface Panel  
(WIP)

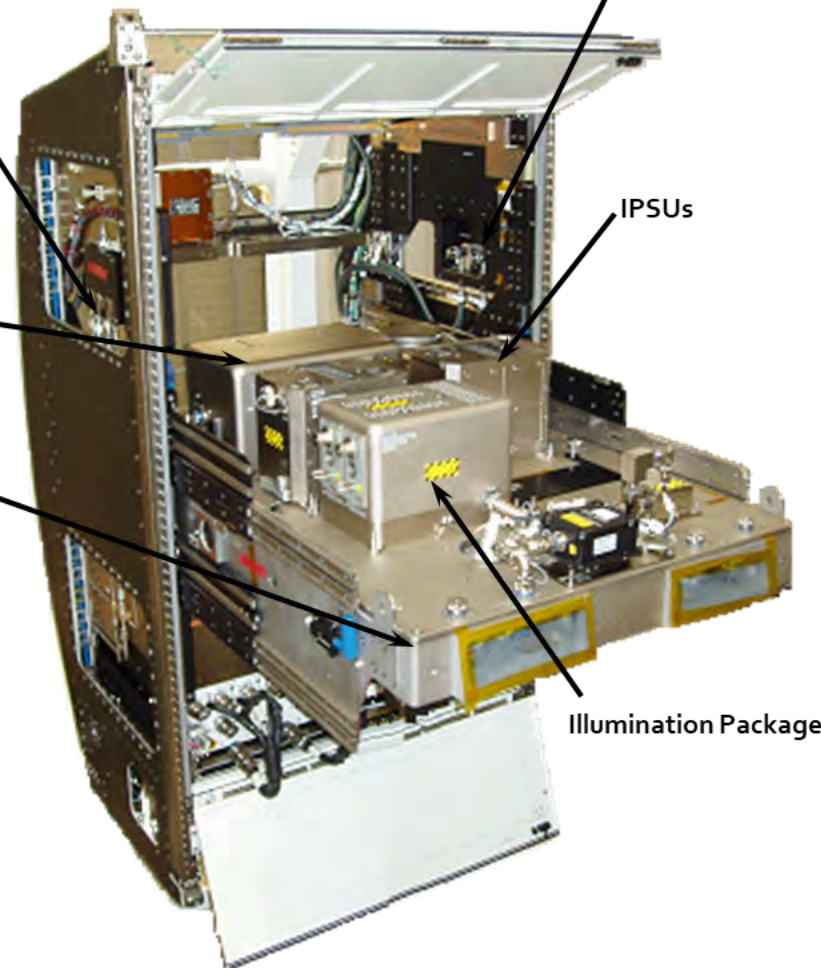
CCU

Optics Bench

PI-specific hardware  
and FIR diagnostics

IPSUs

Illumination Package







## NG-16 Cygnus

SS Ellison Onizuka

**Operator**  
Northrop Grumman

**Launch site**  
Launch Pad 0A  
Wallops Island, Virginia

**Launch vehicle**  
Antares 230+

**ISS docking location**  
Unity nadir

**Cargo mass**  
~3,700 kilograms

**What's aboard?**  
*(Not an exhaustive list)*

**Pressurized Cargo Module**

- Cardinal Muscle evaluation
- Redwire Regolith Print study
- The Flow Boiling and Condensation Experiment
- Kentucky Re-Entry Probe Experiment
- Four Bed CO2 Scrubber
- Blob investigation
- Slingshot Deployment System
- ISS Power Augmentation Mod Kit
- Upgraded acrylic scratch panes for Cupola windows
- Airlock Stowage Platform

**Service Module**

- 2 UltraFlex solar arrays
- Nanoracks External CubeSat Deployer

6.39 meters

3.1 meters

ANTARES

SFI *Space Flight Initiative*





Largest Cygnus cargo ever  
launched to ISS (3,723 kg)



Launch (August 10, 2021, 6:01 pm ET)



Arrival at ISS (August 12, 2021, 6:10 am ET)



# *Predictive Methods*

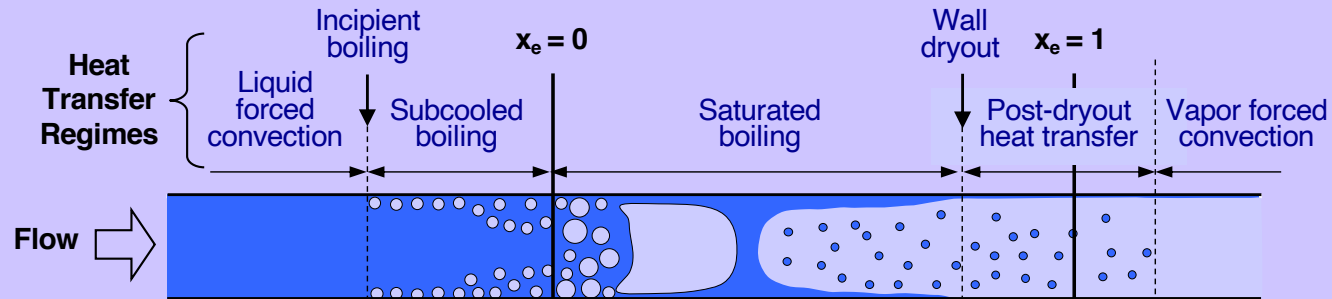


- **Computational methods:** computationally very intensive; predictions very sensitive to numerical and phase change model used
- **Theoretical models:** very few exist!
- **Semi-empirical models:** mechanistic models with specific constants determined from empirical data
- **'Universal' experimental correlations:** based on large databases for many fluids and broad ranges of geometrical or flow parameters; few exist; most desirable in industry
- **Narrow-range experimental correlations:** applicable to one or very few fluids and limited ranges of geometrical or flow parameters; often extrapolated resulting in erroneous predictions
- **Replication:** performance based on specific fluid, geometry, and operating conditions of given device

### Recent Challenges:

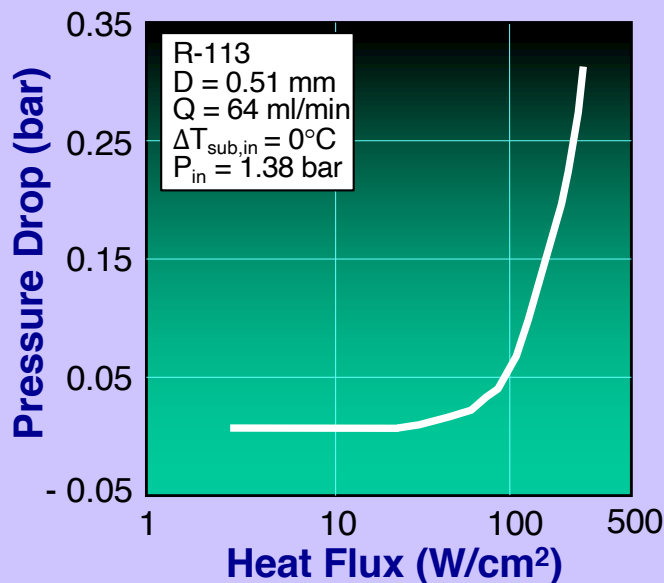
- Reluctance to develop large, complex phase-change facilities
- Studies focused on development of narrow-range correlations or simply data trends
- Findings very incremental
- Avoidance of mechanistic models

## Two-Phase Flow and Heat Transfer Regimes

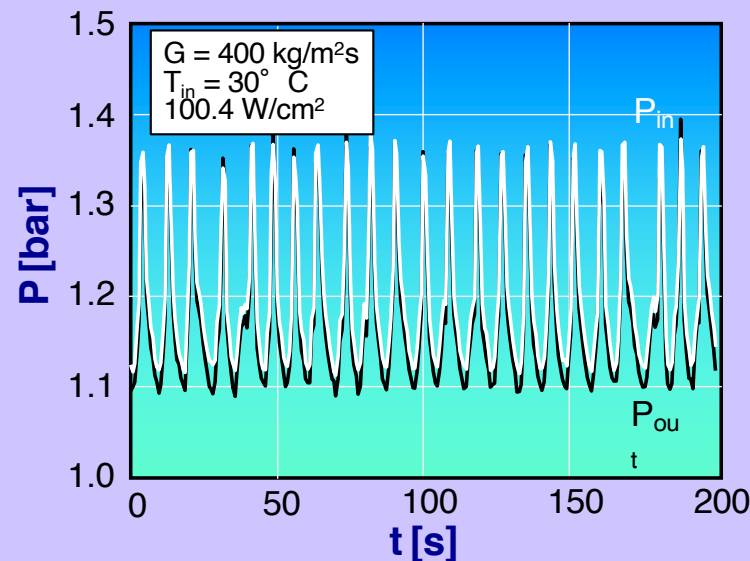


- Heat transfer regimes
- Regime boundaries
- Pressure drop components
- Heat transfer coefficients
- Dryout versus CHF

## Flow Constrains and Instabilities

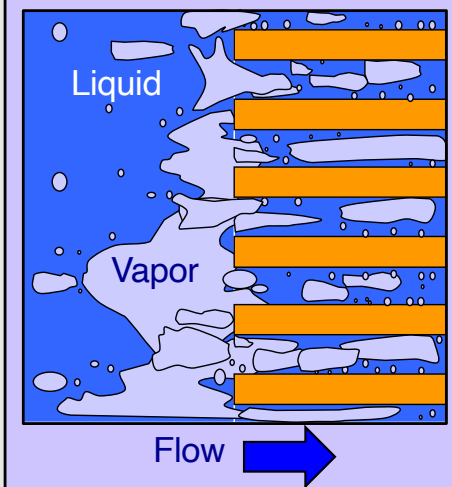


- Compressibility
- Flashing
- Two-phase choking



- Instabilities

## Heat Transfer Limits and Anomalies



- CHF
- Premature CHF





## Universal Heat Transfer Correlation for Saturated Boiling in Small Channel

Consolidated database:  
10,805 saturated boiling heat transfer coefficient data points  
from 37 sources

- FC72, R11, R113, R123, R1234yf, R1234ze, R134a, R152a, R22, R236fa, R245fa, R32, R404A, R407C, R410A, R417A, CO<sub>2</sub>, water
- $0.19 < D_h < 6.5$  mm
- $19 < G < 1608$  kg/m<sup>2</sup>s
- $57 < Re_{fo} = GD_f/\mu_f < 49,820$
- $0 < x < 1$
- $0.005 < \text{Reduced pressure} < 0.69$

$$Bo = \frac{q_H''}{G h_{fg}}$$



$$h_{tp} = (h_{nb}^2 + h_{cb}^2)^{0.5}$$

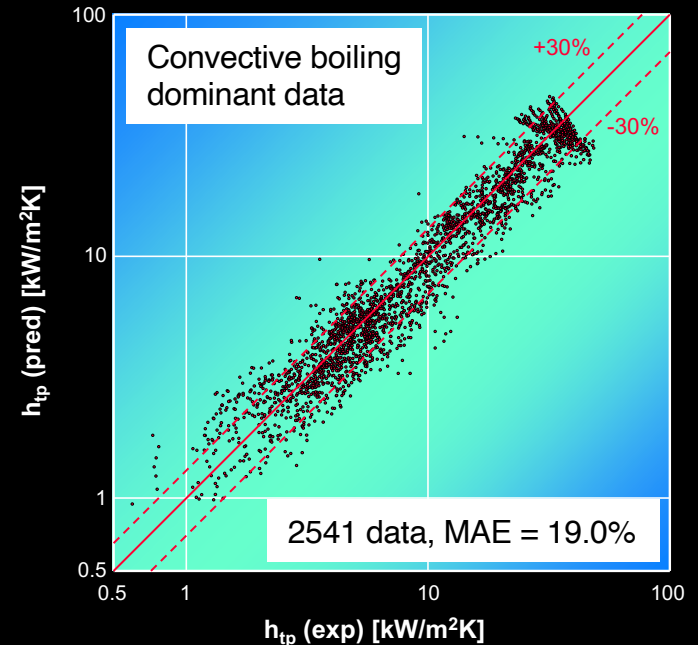
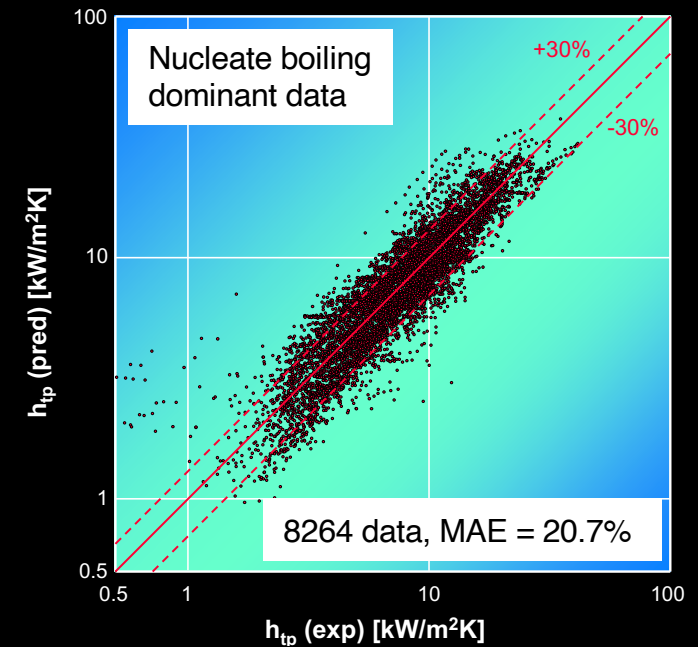
For nucleate boiling dominant regime :

$$h_{nb} = \left[ 2345 \left( Bo \frac{P_H}{P_F} \right)^{0.70} P_R^{0.38} (1-x)^{-0.51} \right] \left( 0.023 Re_f^{0.8} Pr_f^{0.4} \frac{k_f}{D_h} \right)$$

For convective boiling dominant regime :

$$h_{cb} = \left[ 5.2 \left( Bo \frac{P_H}{P_F} \right)^{0.08} We_{fo}^{-0.54} + 3.5 \left( \frac{1}{X_{tt}} \right)^{0.94} \left( \frac{\rho_g}{\rho_f} \right)^{0.25} \right] \left( 0.023 Re_f^{0.8} Pr_f^{0.4} \frac{k_f}{D_h} \right)$$

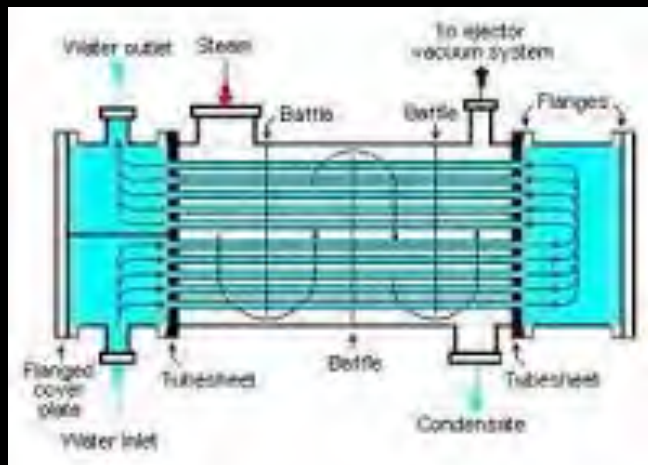
Kim & Mudawar, *Int. J. Heat Mass Transfer*, 2013

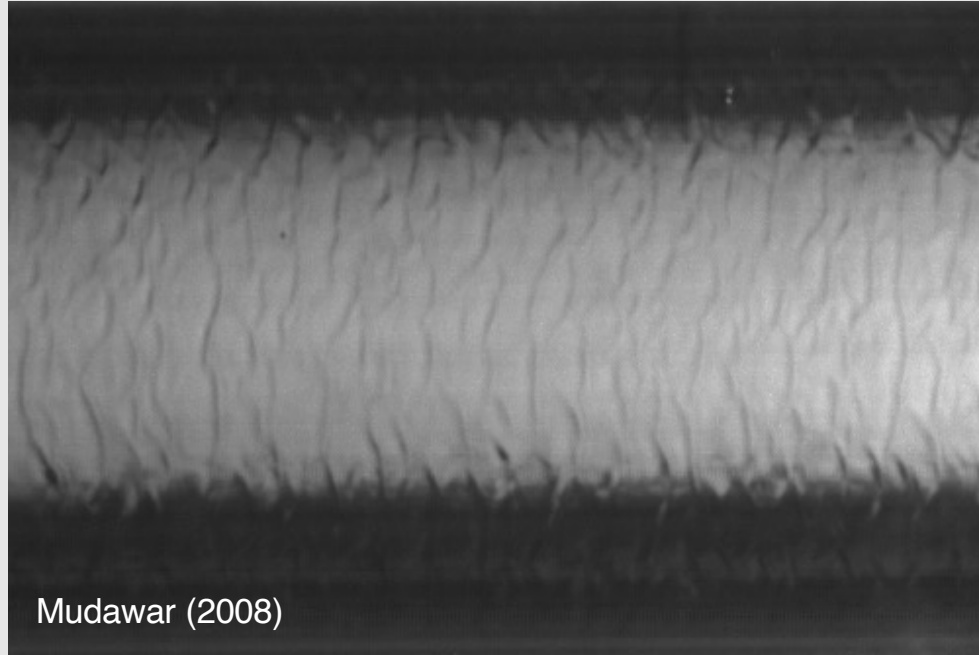


## Air Cooled Condensers for Refrigerant

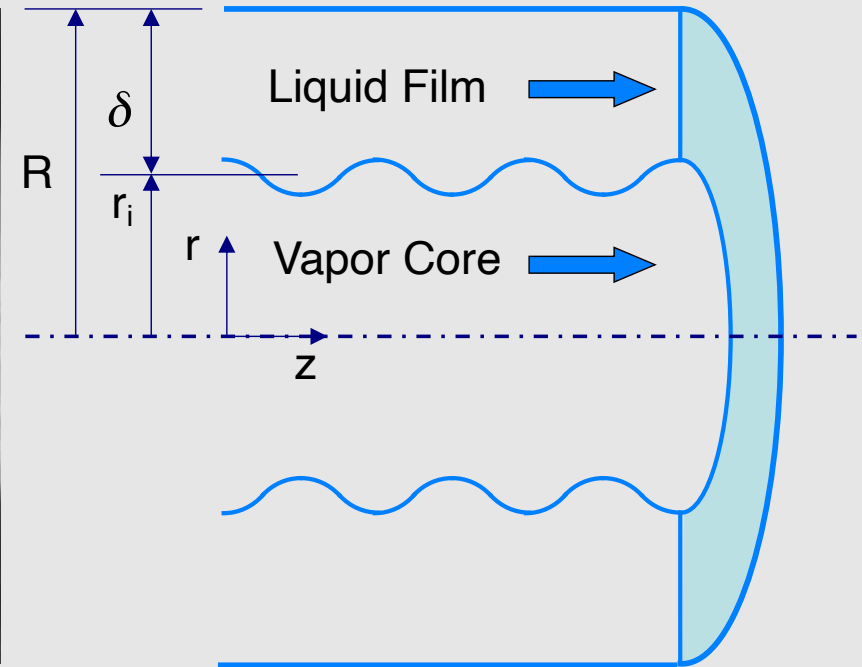


## Shell-and-Tube Condensers





Mudawar (2008)



- Different flow regimes: pure vapor, annular, slug, bubbly and pure liquid
- Annular regime most prevalent ... separated two-phase flow consisting of thin shear-driven liquid film along tube wall and central vapor core
- Thin annular film responsible for high condensation heat transfer coefficients



**Common perception:**

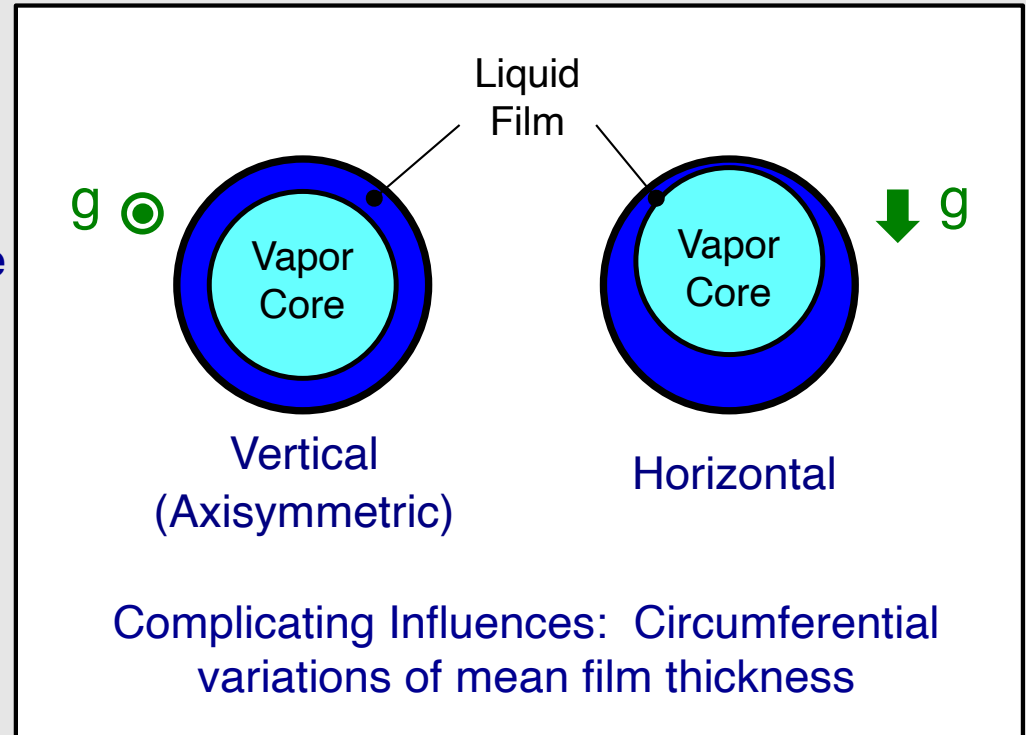
Separated flows presumed convenient to model because they can be represented by two predominantly single-phase flows

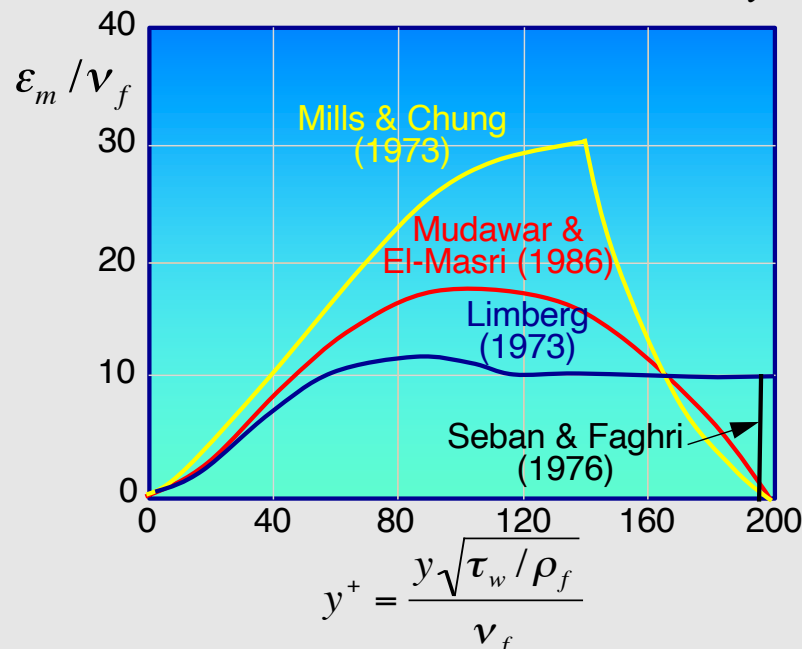
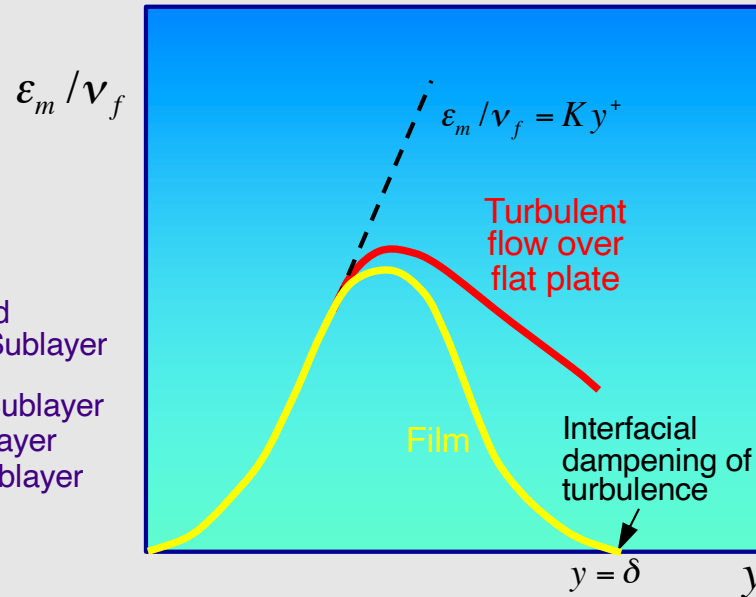
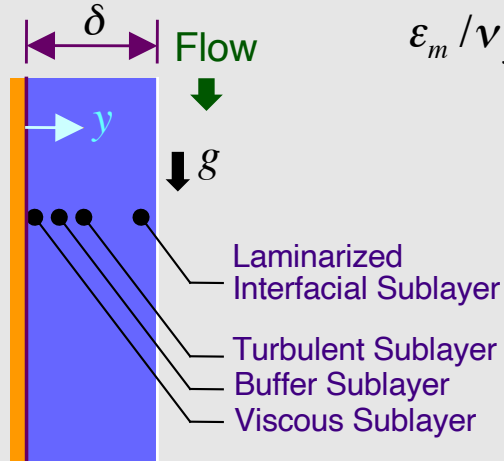
**Complicating Influences (Fluid Physics):**

1. Circumferential variations of mean film thickness due to transverse body force
2. Complex transitional characteristics of condensation films
3. Turbulence dampening near film interface
4. Interfacial waves
5. Droplet entrainment and deposition

**Challenges:**

1. Develop fundamental understanding of all above
2. Incorporate all above influences into robust predictive model

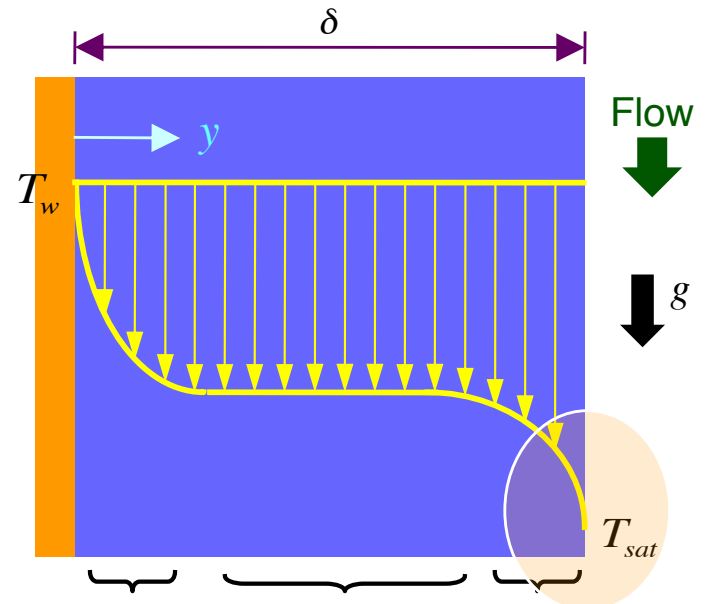




- Near-wall region: Most use Van Driest (1956) function
- Mudawar & El-Masri: Ueda et al. (1977) experimental profile in middle, and a function of Kapitza number near interface:

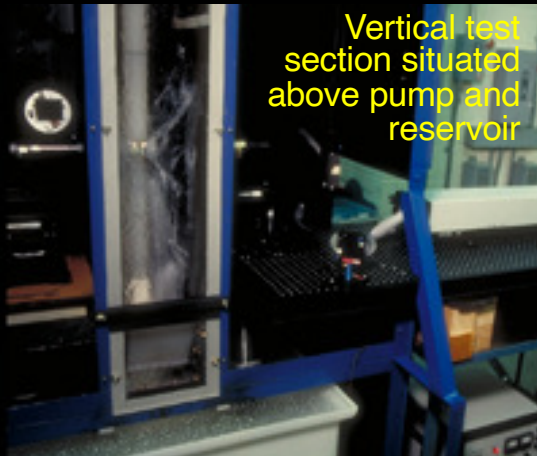
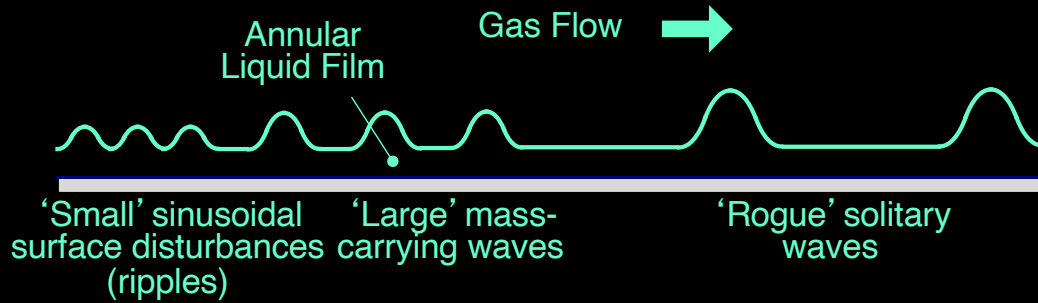
$$Ka = \frac{\mu^4 g}{\rho \sigma^3}$$

$$-q'' = k_f \left( 1 + \frac{\text{Pr}_f}{\text{Pr}_{f,t}} \frac{\epsilon_m}{\nu_f} \right) \frac{d\bar{T}}{dy}$$



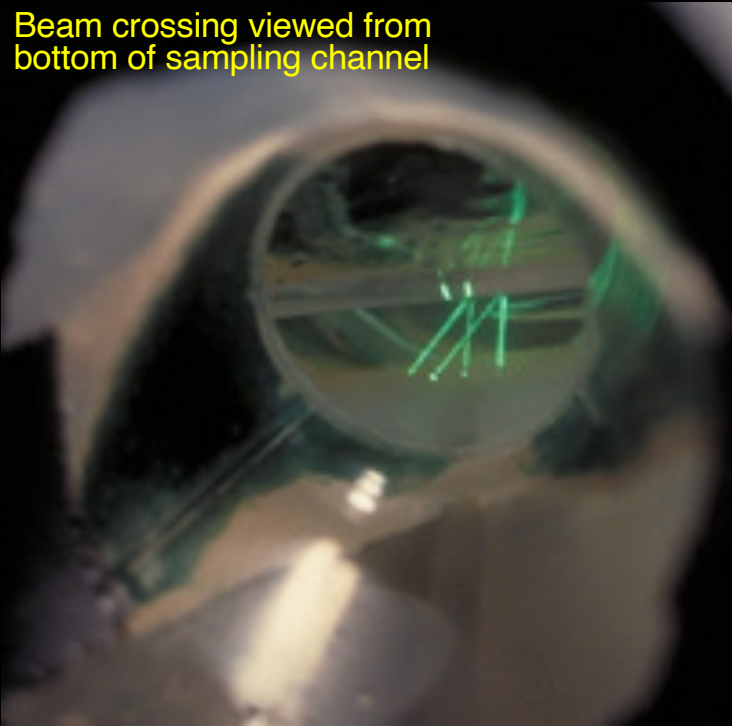
$\frac{\epsilon_m}{\nu_f} \approx 0$	$\frac{\epsilon_m}{\nu_f} \uparrow \uparrow$	$\frac{\epsilon_m}{\nu_f} \approx 0$
$\frac{d\bar{T}}{dy} \uparrow$	$\frac{d\bar{T}}{dy} \downarrow \downarrow$	$\frac{d\bar{T}}{dy} \uparrow$

Failure to account for interfacial dampening of turbulence near the interface and large interfacial temperature gradient has been shown to produce significant errors in predicting heat transfer across films!!

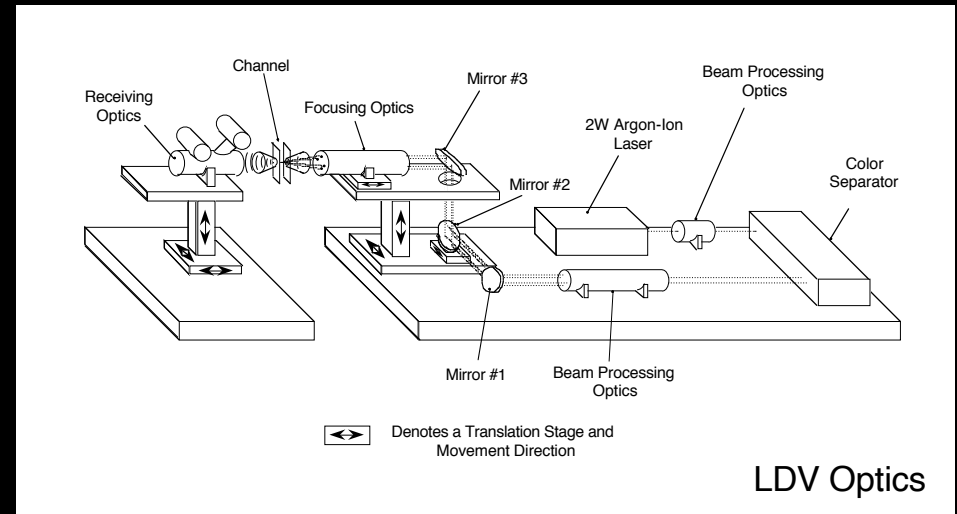
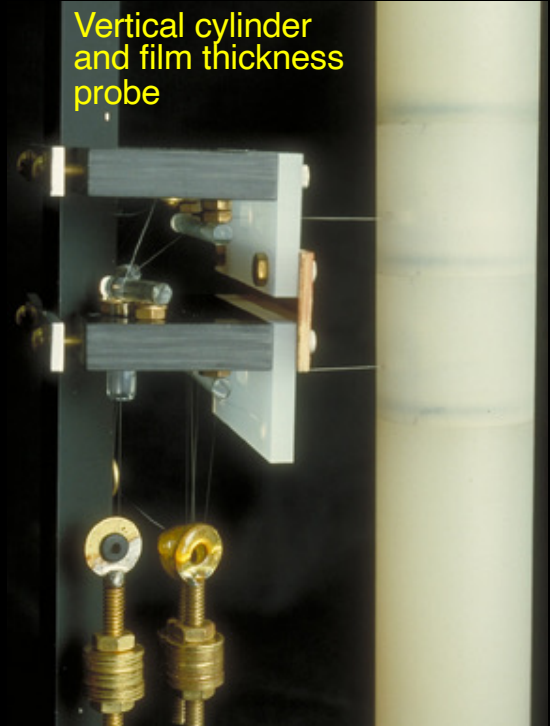


Mudawar & Houpt  
(1993a,b)

Beam crossing viewed from  
bottom of sampling channel

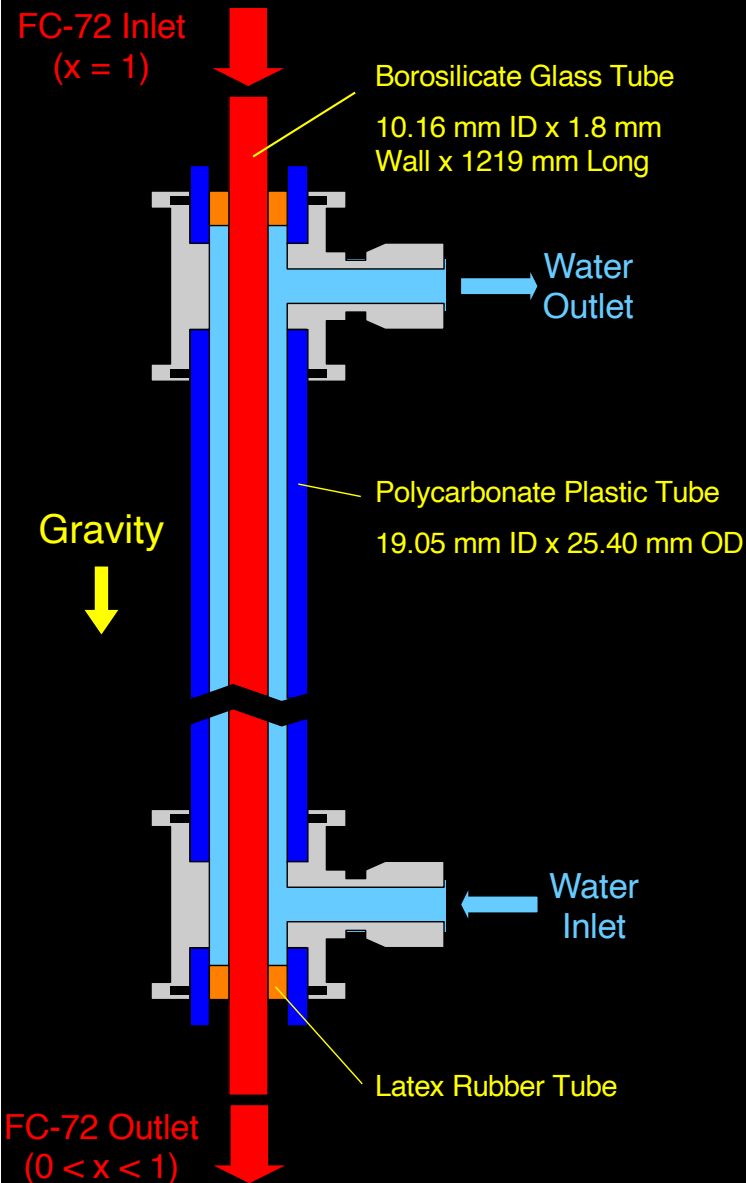


Vertical cylinder  
and film thickness  
probe



LDV Optics

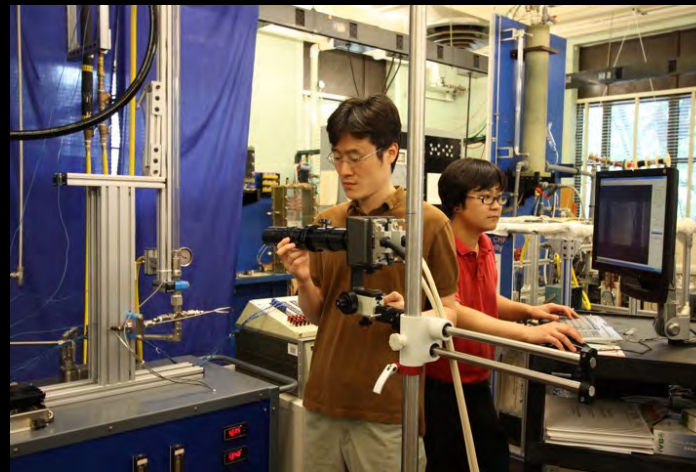




Condensation Module for Flow Visualization



*Purdue  
High Capacity  
Condensation  
Facility*



High-Speed Video Motion  
Analysis System:

- Photron FASTCAM-Ultima camera with shutter speeds up to 1/120,000 s
- Infinity K-2 high magnification lenses
- PerkinElmer Xenon source fitted with Olympus fiber optic cable

Flow

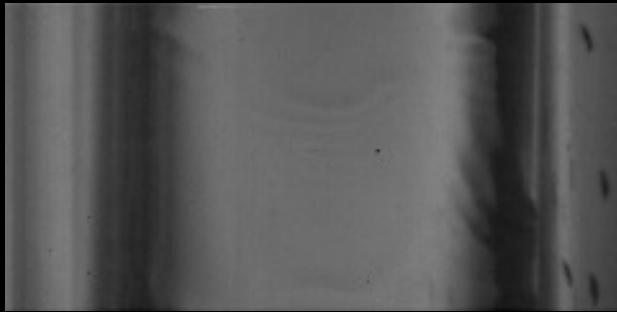


Inlet

$$\dot{m}_{\text{FC-72}} = 19.5 - 19.6 \text{ g/s}$$

1/267 speed

Video:  
4000 frames/s



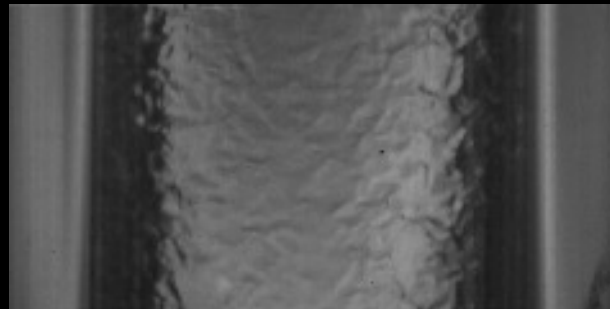
Video:  
4000 frames/s

1/133 speed



**Purdue  
High Capacity  
Condensation  
Facility**

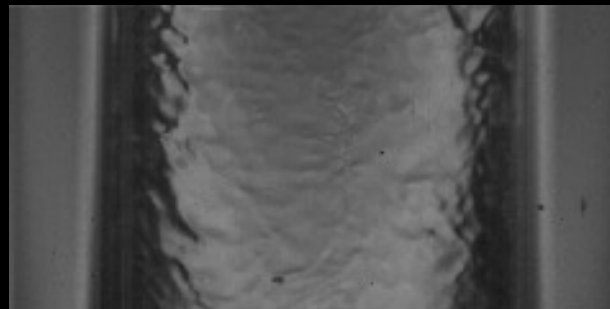
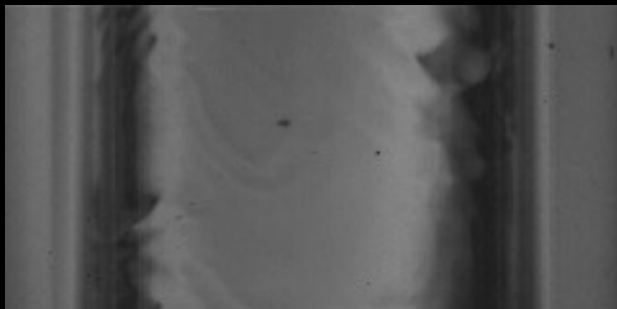
Middle



For  $Re_{f, \text{FC-72, exit}} < 200$

- Smooth Laminar in inlet
- Wavy Laminar in middle and outlet

Outlet



For  $Re_{f, \text{FC-72, exit}} > 200$

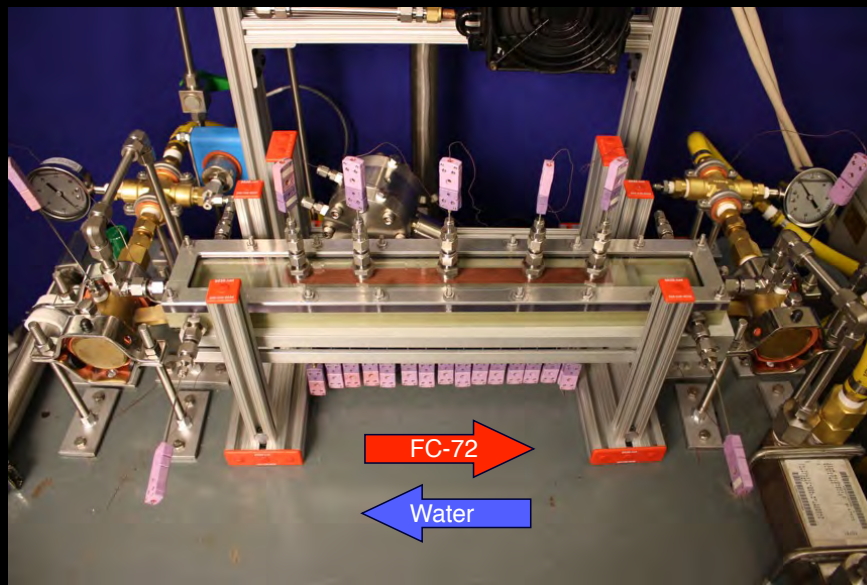
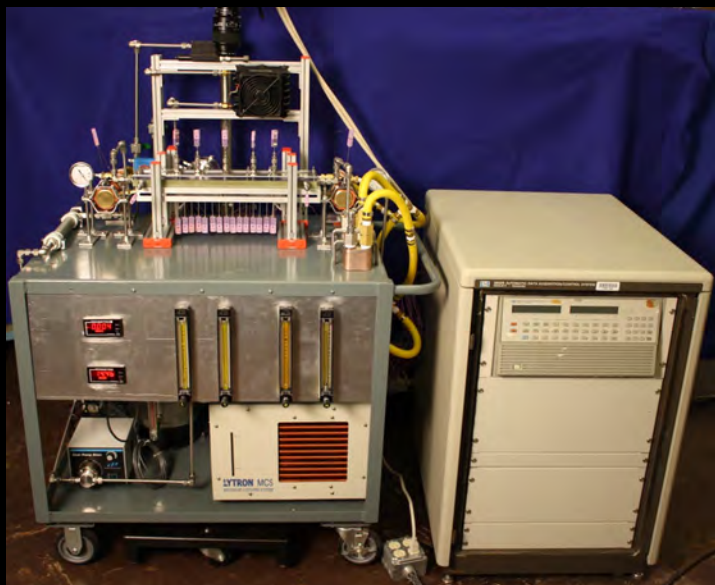
- Flow appears to turn turbulent
- Wave structure more complex

$$Re_{f, \text{FC-72, exit}} = 179 - 181$$

$$q_{\text{out}} [\text{W}] = 77.5 - 78.3$$

$$Re_{f, \text{FC-72, exit}} = 927 - 960$$

$$q_{\text{out}} [\text{W}] = 392 - 406$$



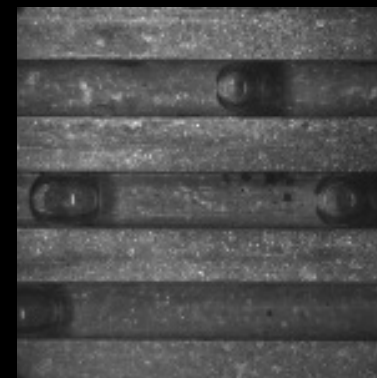
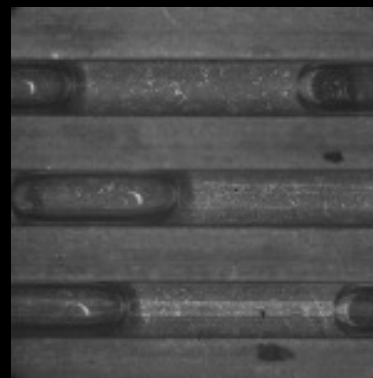
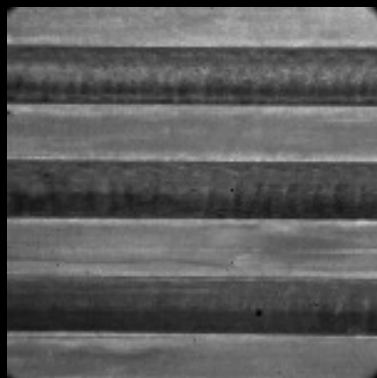
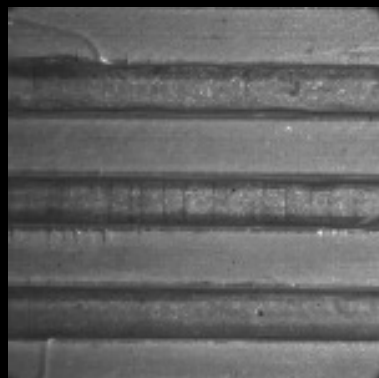
Annular

Wavy-Annular

Slug

Bubbly

Flow



**Purdue**  
**Small Channel**  
**Condensation Facility**

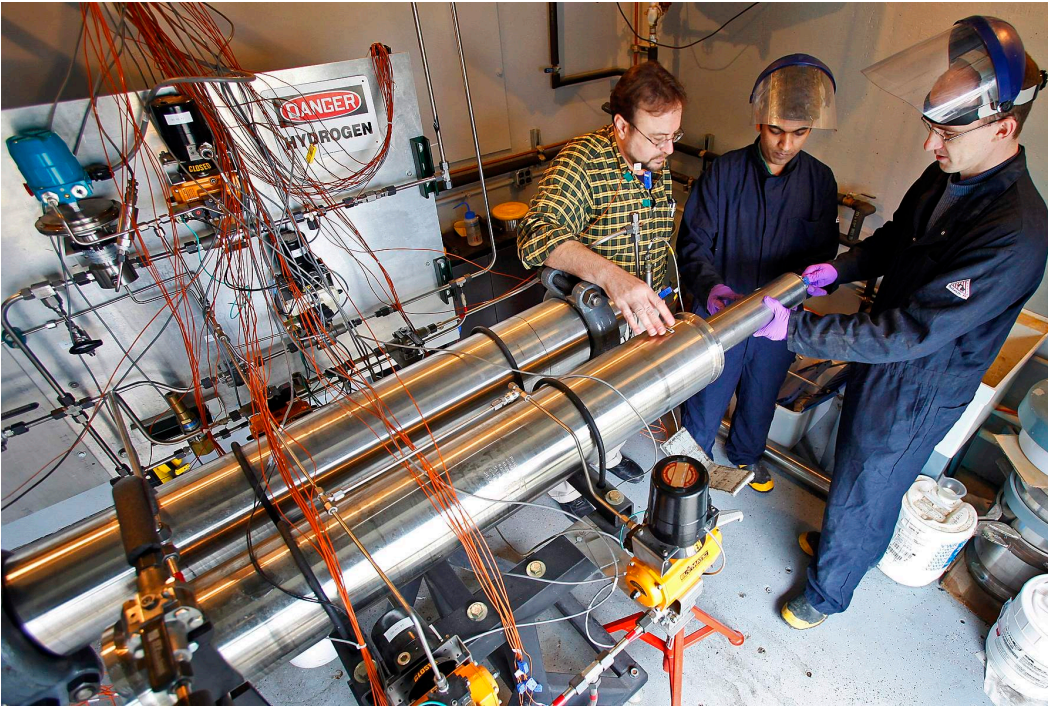
Regime	FC-72 (g/s)	Water (g/s)	Video Rate
Annular	1.51	5.06	1/66.6
Wavy-Annular	1.51	2.97	1/66.4
Slug	0.55	2.97	1/66.4
Bubbly	0.55	5.06	1/66.4



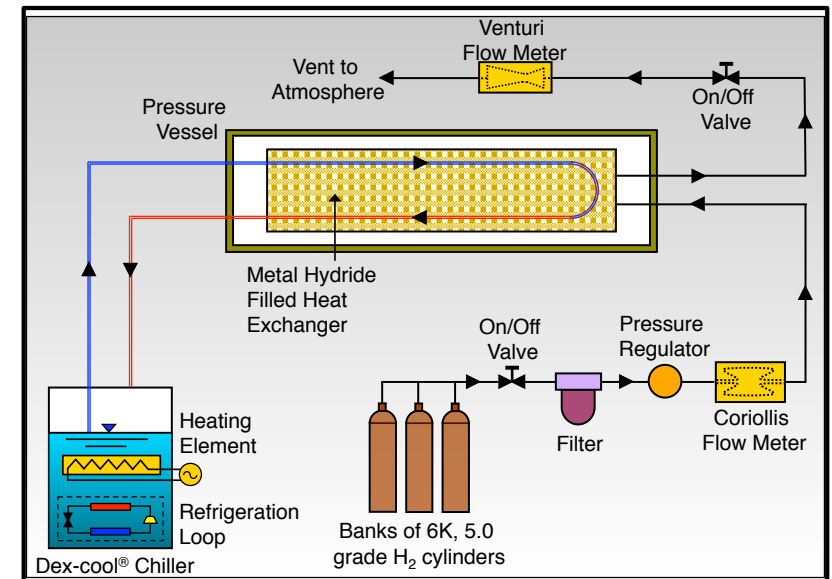
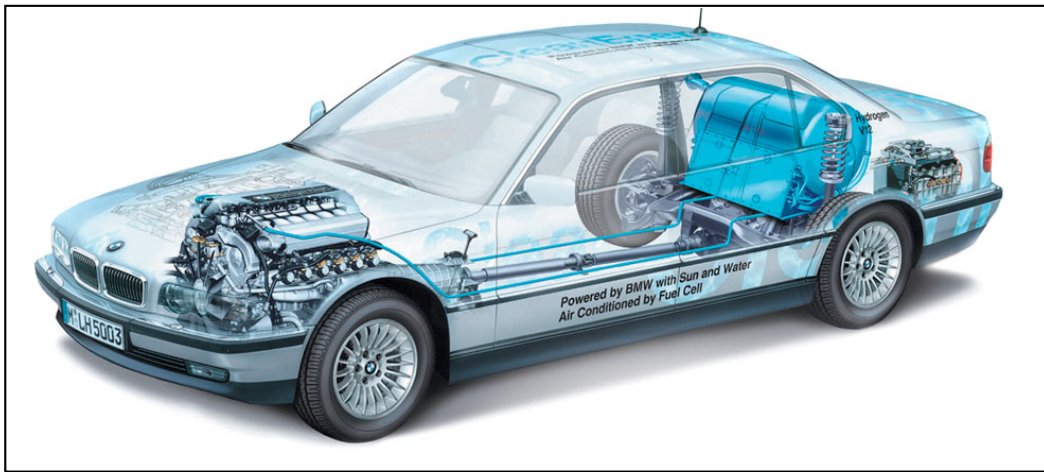
# *In Pursuit of More Efficient, Renewable and Cleaner Energy ...*

- 1. Hydrogen Fuel Cells*
- 2. Hybrid Vehicles*
- 3. Solar Receivers*
- 4. Intelligent Heat Treating of Metal Alloys*

# 1. Heat Exchanger for Hydrogen Storage Fuel Cell Systems

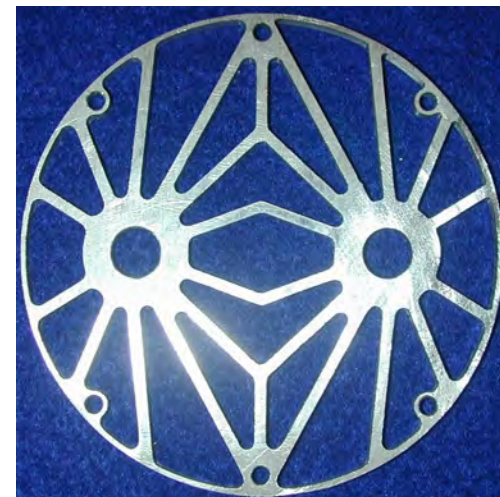
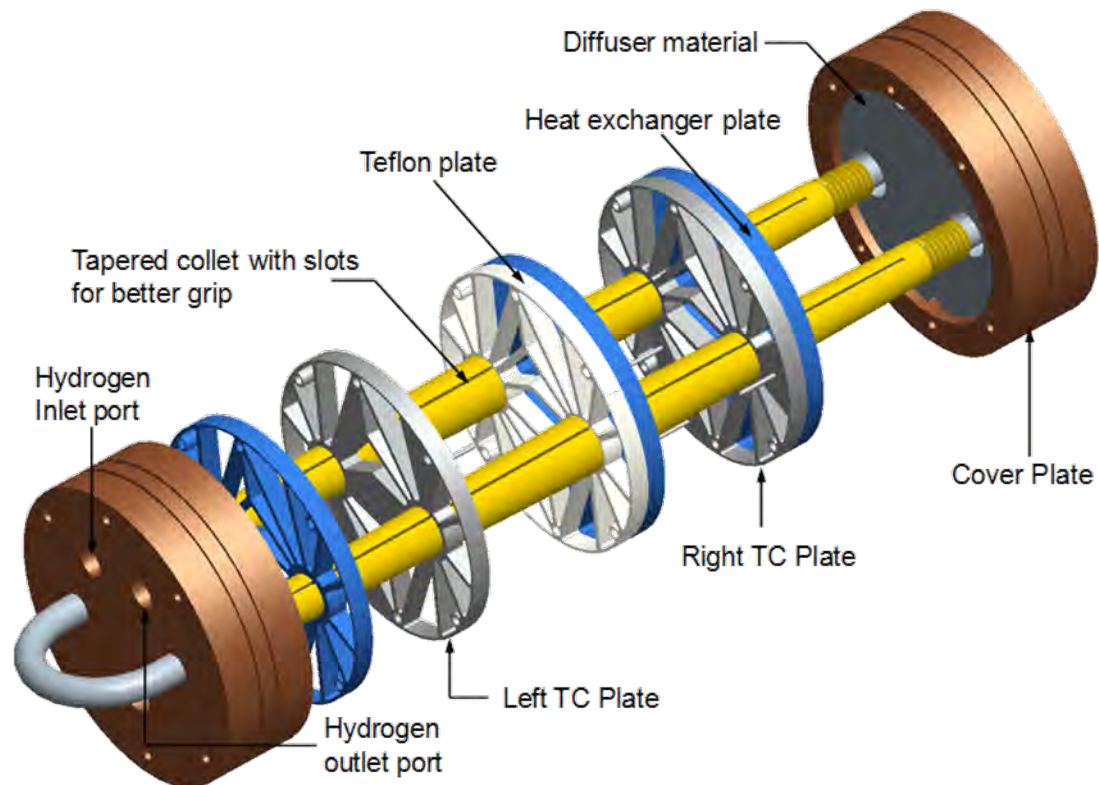


- Developed for high-pressure metal hydride (e.g.,  $\text{Ti}_{1.1}\text{CrMn}$ ) storage systems
- Used to remove large amount of heat dissipated from hydriding reaction when  $\text{H}_2$  is charged into storage vessel



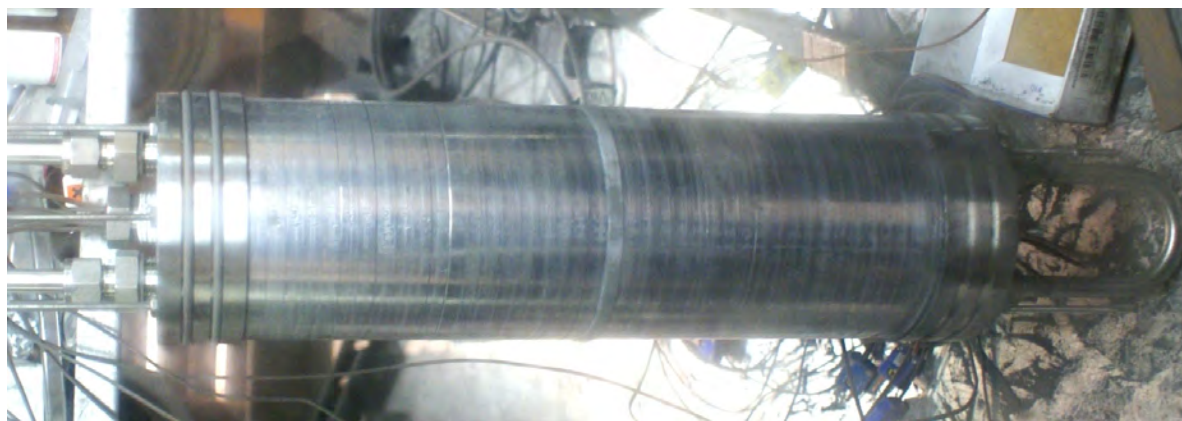


## 1. Thermally-Optimized Heat Exchanger



**US Patent 8,636,836 B2**  
**January 28, 2014**

**Assembled Heat  
Exchanger**





# 1. Transient Contours of Temperature, Fraction of Reaction Completion and Volumetric Heat Generation Rate during Hydriding

*P = 100 bar*

*P = 200 bar*

*P = 300 bar*

*P = 300 bar*

*P = 300 bar*

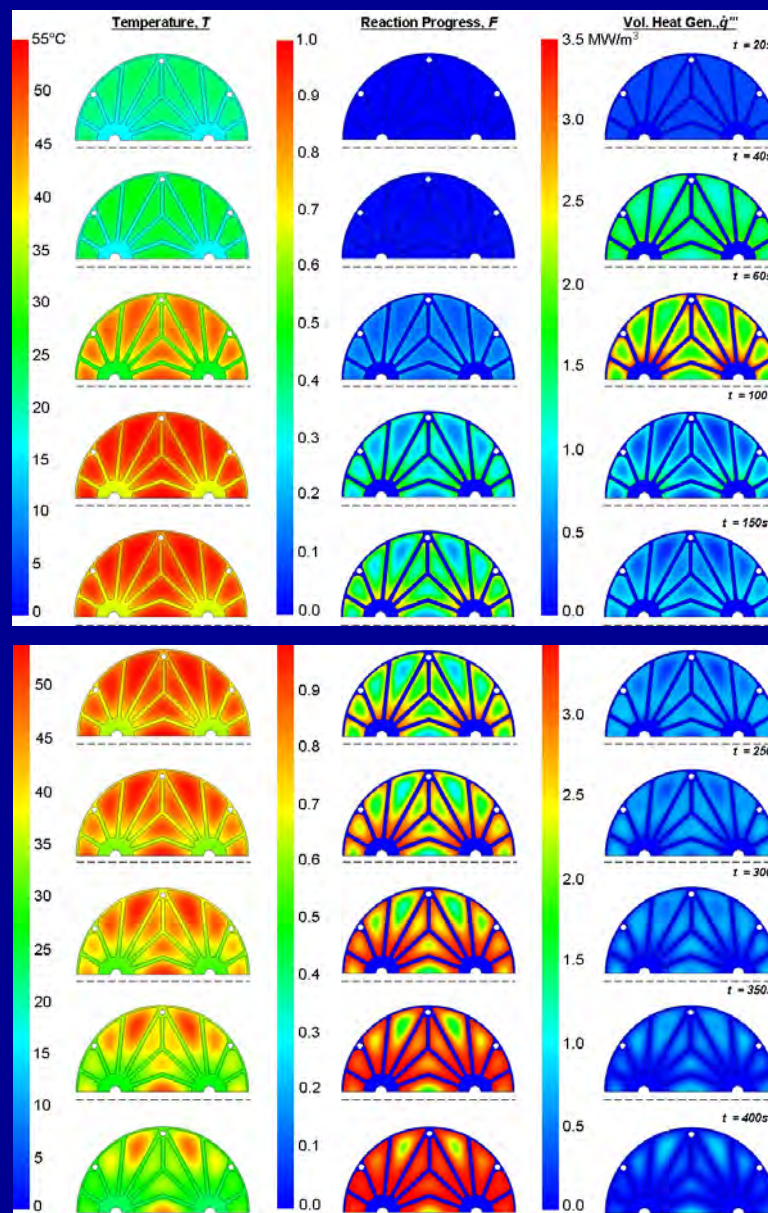
*P = 300 bar*

*P = 300 bar*

*P = 300 bar*

*P = 300 bar*

*P = 300 bar*

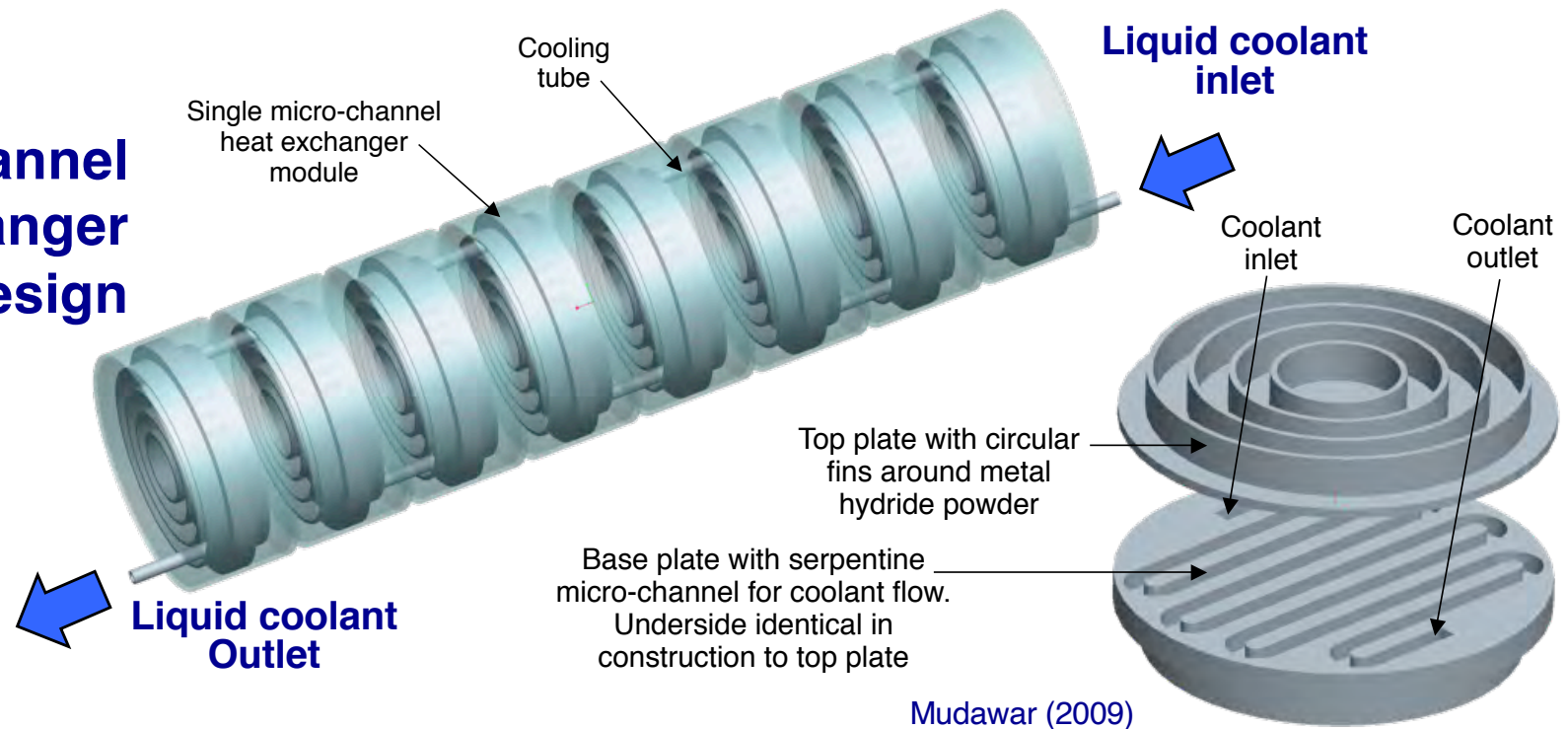




**Coiled  
Heat Exchanger  
Design**

**US Patent 8,778,063 B2  
July 15, 2014**

**Micro-Channel  
Heat Exchanger  
Design**

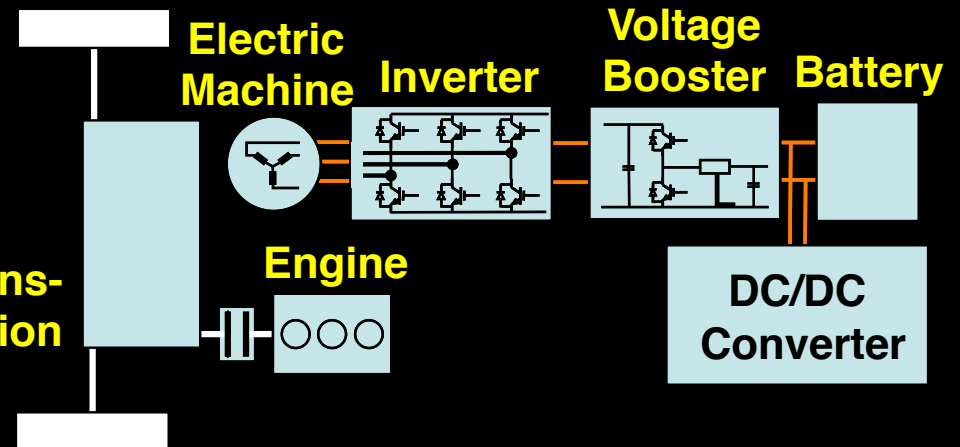
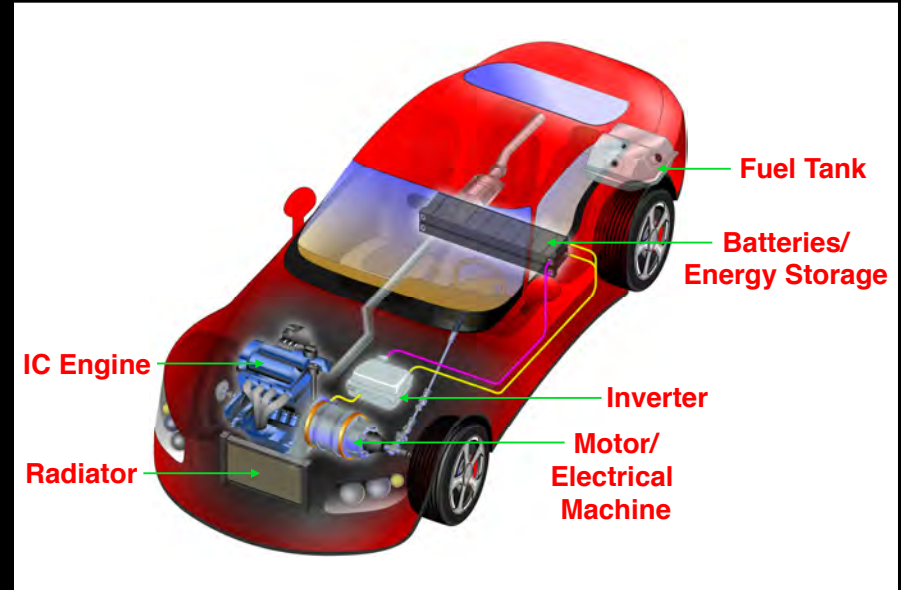
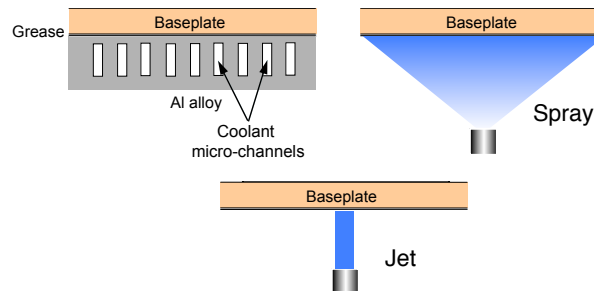
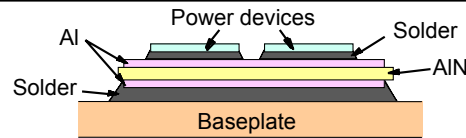
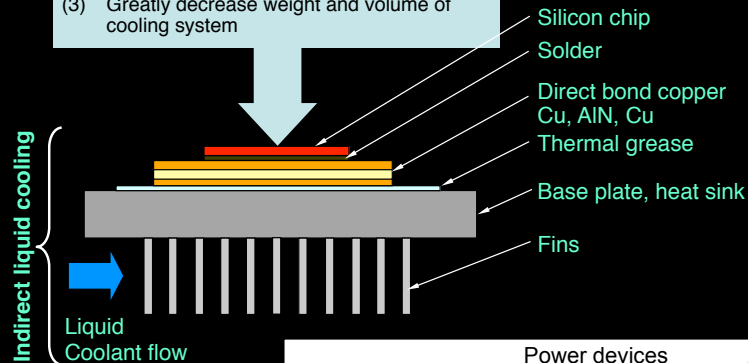


Mudawar (2009)

## 2. Hybrid Vehicle Power Electronics

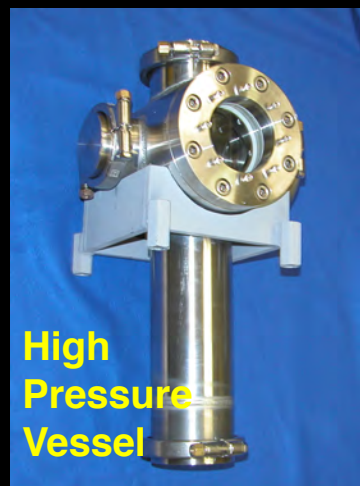
Alternative **direct liquid cooling** of chip to:

- (1) Eliminate conductive resistances of intermediate layers
- (2) Allow closer packaging of high-flux chips
- (3) Greatly decrease weight and volume of cooling system

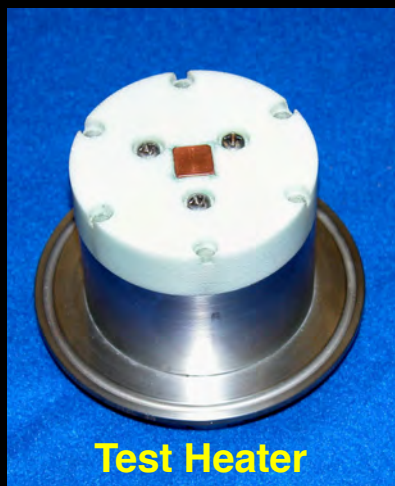


- Goal: dissipate over  $250 \text{ W/cm}^2$  while maintaining device temperature below  $125^\circ \text{ C}$  for silicon devices
- Two-phase cooling options: R134a loop that taps into vehicle's refrigeration loop, or separate HFE 7100 cooling loop

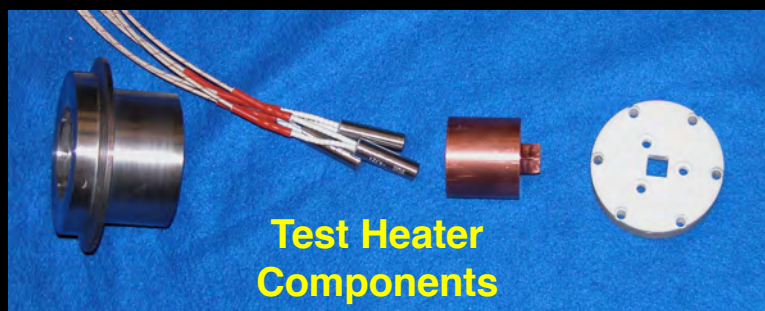




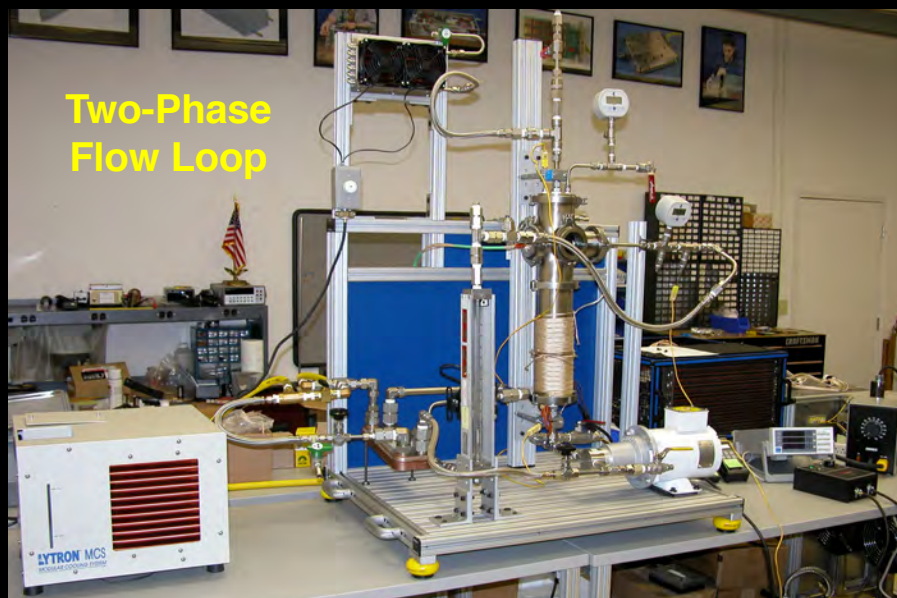
High Pressure Vessel



Test Heater



Test Heater Components



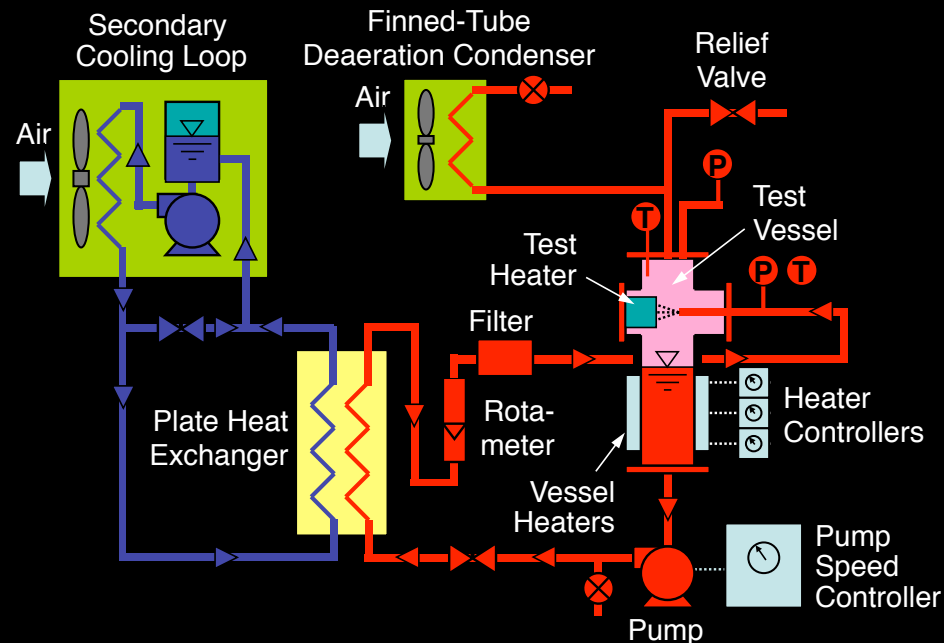
Two-Phase Flow Loop



Mudawar, Bharathan, Kelly & Narumanchi (2009)

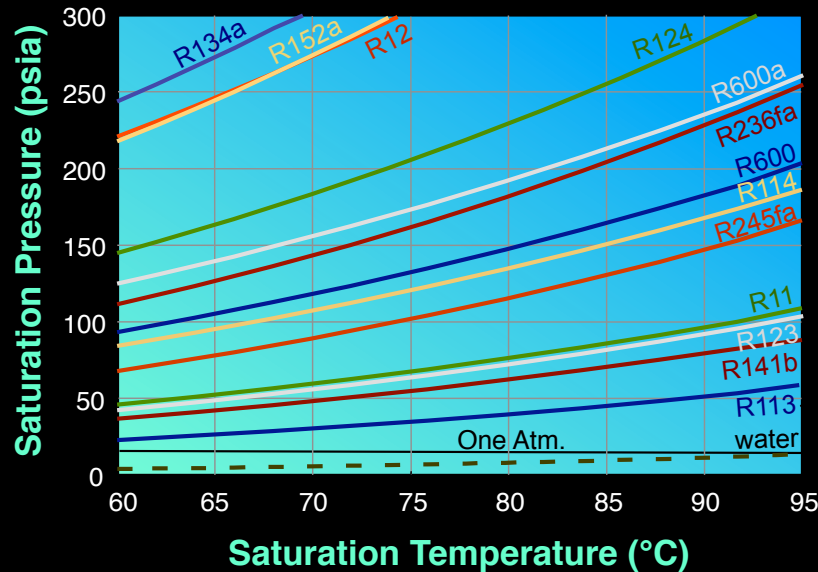


Power Supply and Instrumentation Rack



## 2. Coolant/Refrigerant Thermal/Environmental Study

### Refrigerants



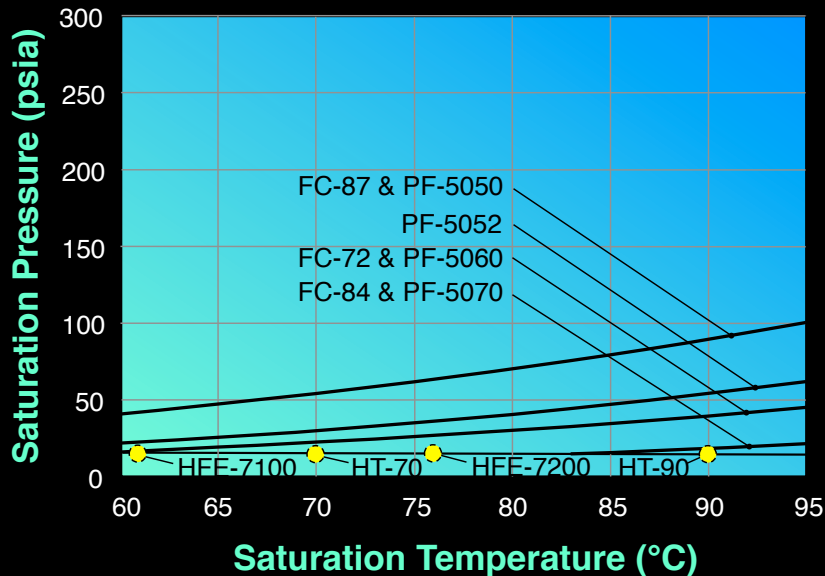
### Refrigerants

**Chlorofluorocarbons (CFCs)** (e.g., R11, R12, R113 and R114) composed of chlorine, fluorine and carbon; they are nontoxic and inert, but highly ozone-depleting and contribute to global warming.

**Hydrochlorofluorocarbons (HCFCs)** (e.g., R123, R124 and R141b) constitute a more recent alternative to CFCs, given their somewhat similar inertness and cooling characteristics but less than 10% of the ozone-depleting effects of CFCs.

**Hydrofluorocarbons (HFCs)** (e.g., R134a and R143a) provide essentially zero ozone depletion and reduced global warming effects; R134a has good ratings in most performance categories.

### Liquid Coolants



### Liquid Coolants

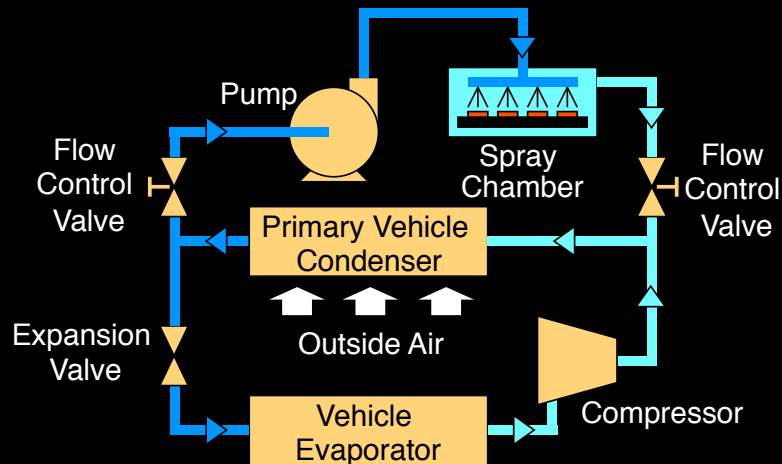
**3M perfluorocarbons (PFCs)** (e.g., Fluorinerts FC-72, FC-87 and FC-84, and Performance Fluids PF-5050, PF-5052, PF-5060 and PF-5070) have *average* environmental ratings because of their relatively high GWP

**3M HFCs** (e.g., Novec fluids HFE-7100 and HFE-7200). HFE-7100 has *good* ratings in all performance categories but HFE-7200 carries a low LEL. HFE-7100 has a freezing point of  $-135^{\circ}\text{C}$ , which is well below any expected automobile application range of temperatures.

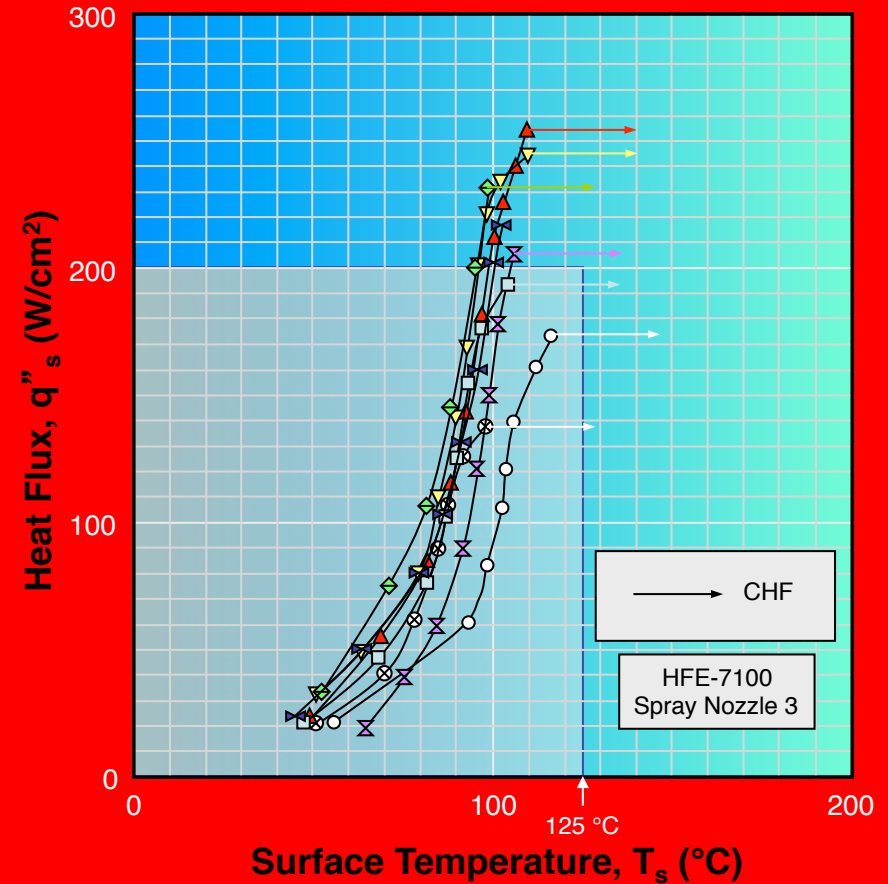
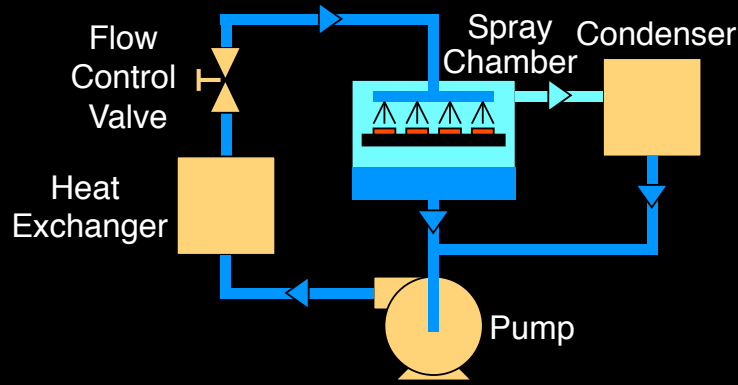
**Most promising: R134a and HFE-7100**

## 2. Cooling Loop Options for Hybrid Vehicle Power Electronics

**Modified R134a Air-Conditioning Refrigeration Loop**



**Separate Cooling Loop with Appropriate Coolant**



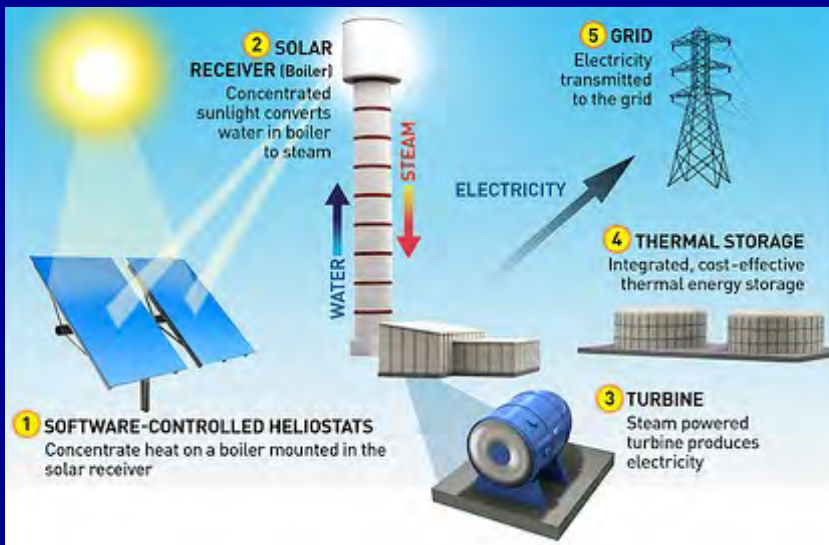
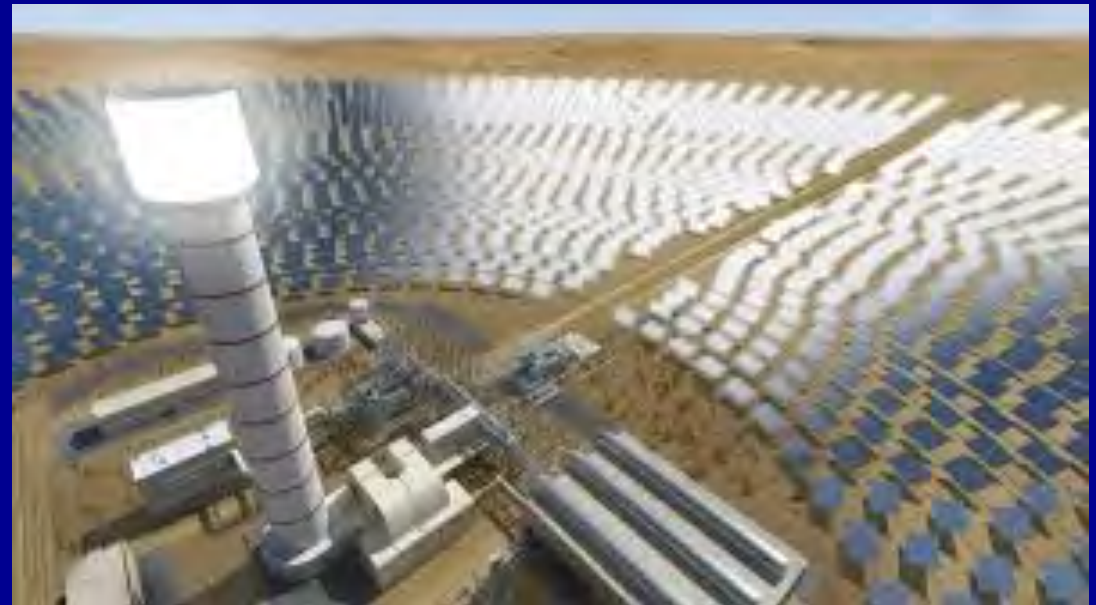
- R134a can yield high CHF values, but cannot maintain low device temperatures.
- HFE-7100 can yield CHF values in excess of 200 W/cm<sup>2</sup> at surface temperatures below 125° C.
- Two-phase HFE-7100 spray cooling is effective at meeting thermal management requirements of hybrid vehicle electronics.



### 3. Solar Receiver Steam Generator

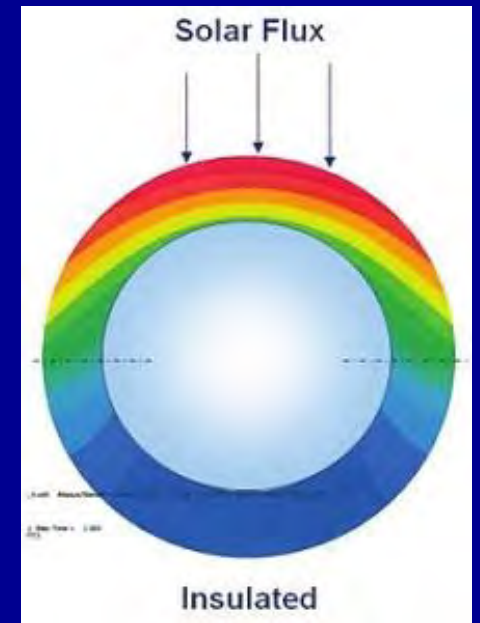
#### Ivanpah Solar Power Facility

- World's largest solar thermal project
- California Mojave Desert
- 3,600 acres (3 units)
- 370 MW power output
- 173,500 heliostats
- Serves 140,000 homes



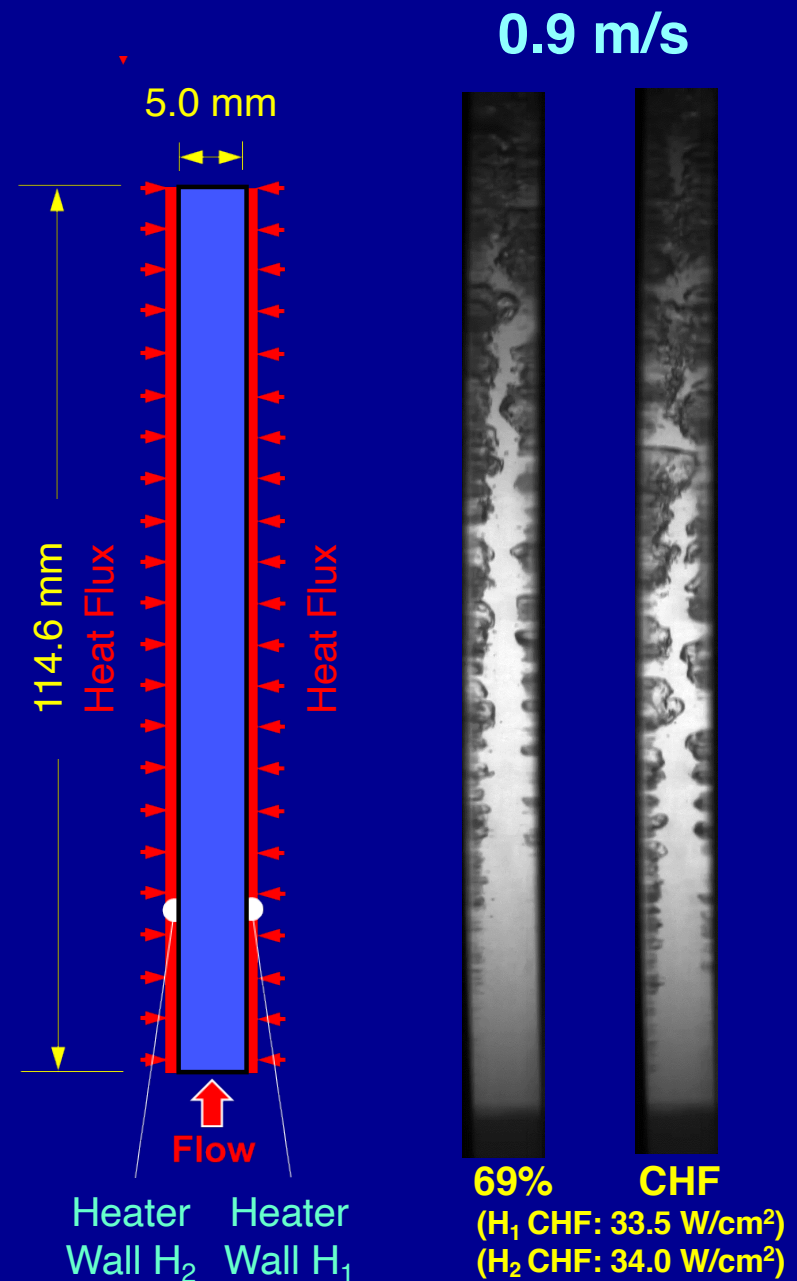
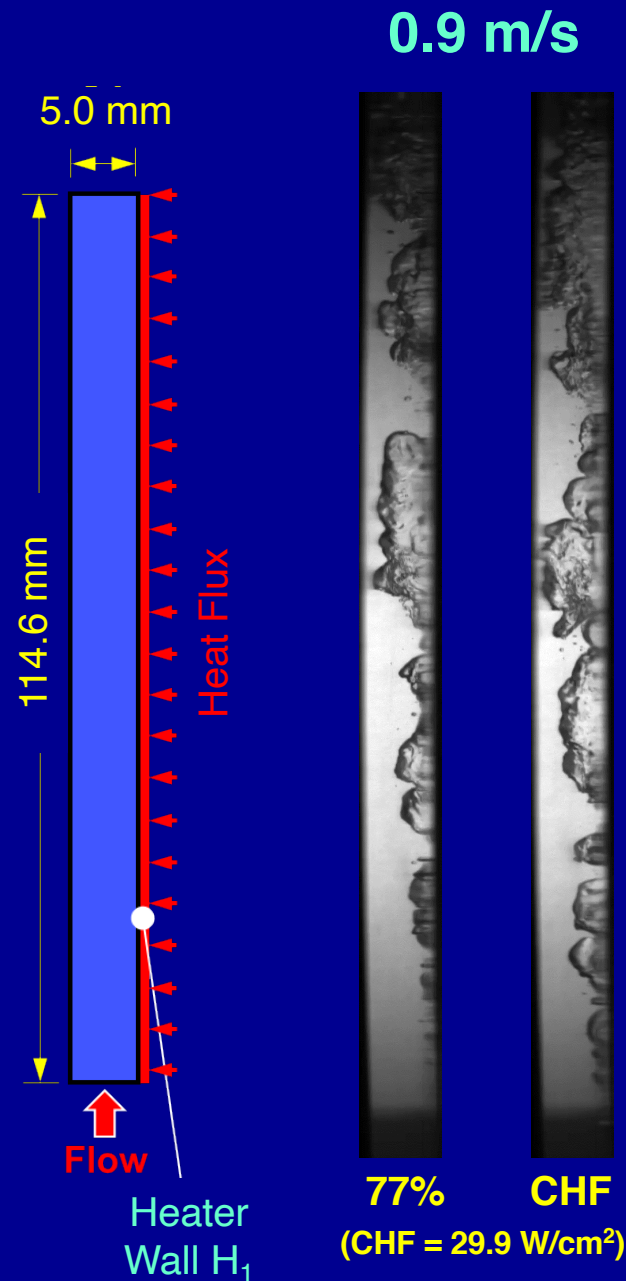
120' Solar Receiver Steam Generator

339' Tall Tower



- Evaporator heats subcooled water to boiling temperature to produce steam-water mixture at high pressure and temperature
- Heat flux one-sided and much higher than in fossil boilers

### 3. Single-Sided versus Double-Sided Heating

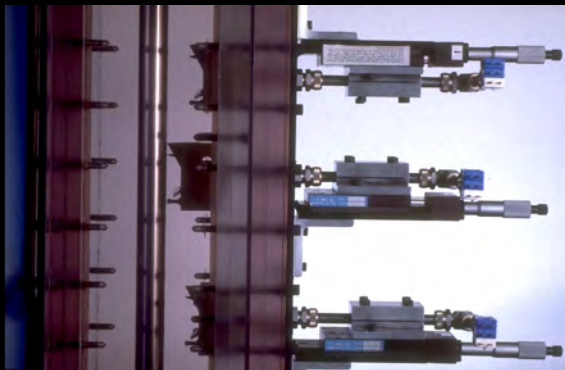




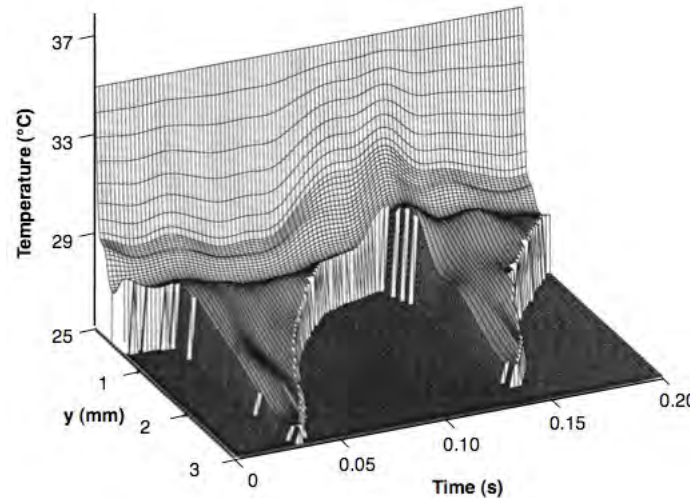
### 3. Measurement of Effects of Waves on Wavy Heated Liquid Layers



**Vertical Evaporation Facility**

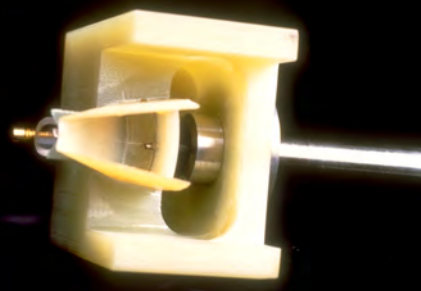


Translation stages for liquid layer probes



Temperature fluctuations across wavy, turbulent water film falling on vertical heated cylinder at  $Re = 5700$  (Lyu & Mudawar, 1991)

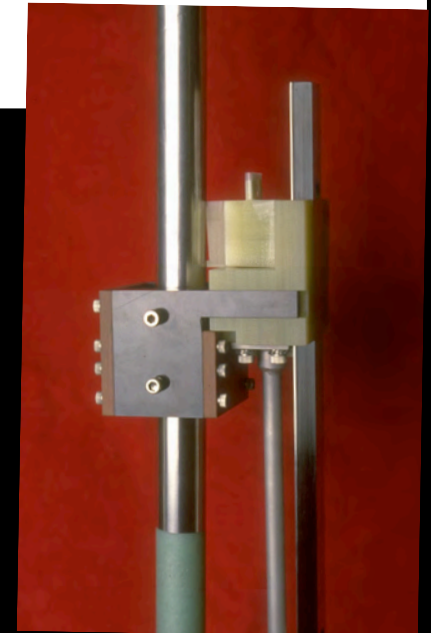
**Thermocouple Knife-edge for Measurement of Temperature Profile across Wavy Liquid Layer**



Scoop for measurement of mean liquid layer temperature

**Probe for simultaneous measurement :**

- Temperature profile across liquid layer
- Liquid layer thickness
- Interfacial wave speed

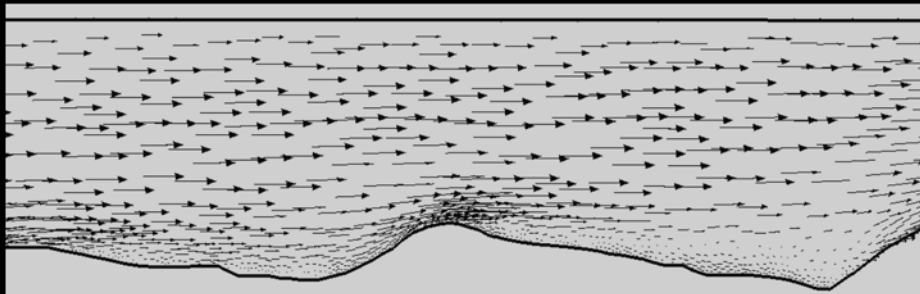




### 3. Computed Nitrogen Velocity Vector and Contour Plots along Stationary Wavy Interface

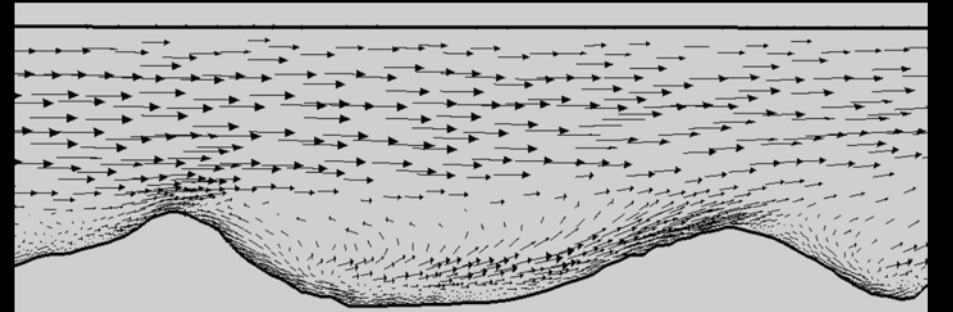
$Re_g = 3450$

→ 24 m/s



$Re_g = 6100$

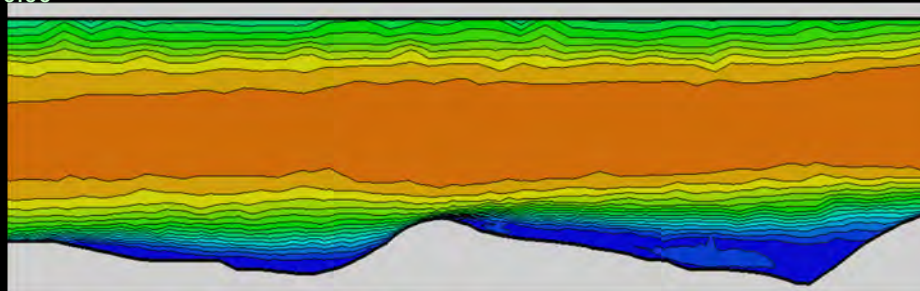
→ 40 m/s



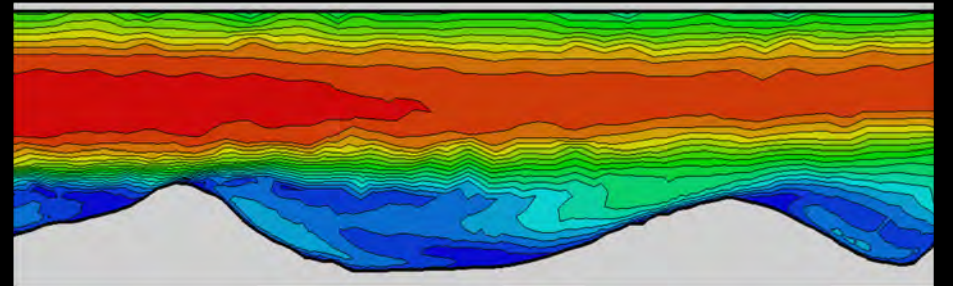
(m/s)

**Nitrogen velocity vector plots at  $z = 0$**

0.00 2.72 5.43 8.14 10.85 13.57 16.28 18.88 (m/s)



0.00 4.78 9.55 14.33 19.1 23.88 28.66 33.44 (m/s)



**Nitrogen velocity contour plot at  $z = 0$**

**FLUENT,  $k - \omega$  model**

## 4. Intelligent Heat Treating of Aluminum Alloy Parts

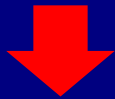
**Enormous  
Energy  
Consumption**

**Poor  
Productivity**

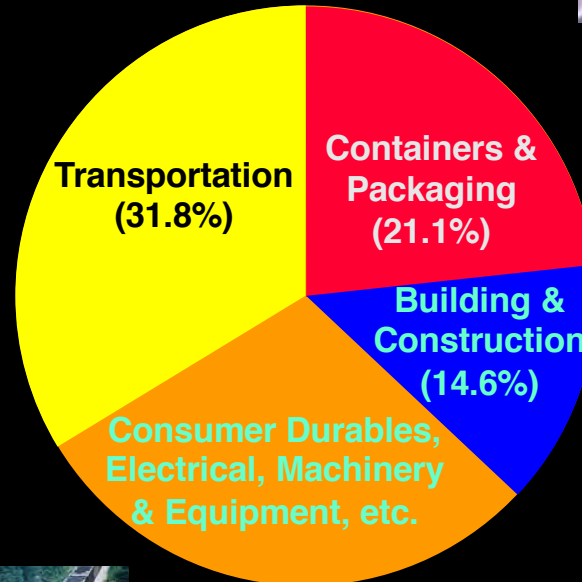
**High Scrap  
Rate**

**High Water  
Consumption**

**High  
Emissions &  
Greenhouse  
Gases**



**Up to 50% of Cost of  
Aluminum Production  
associated with Post-  
Processing of Poorly  
Produced Parts**



**Large  
Residual  
Stresses**

**Warping**

**Cracking**

**Soft Spots**

**Poor  
Corrosion  
Resistance**

**Poor  
Hardness**

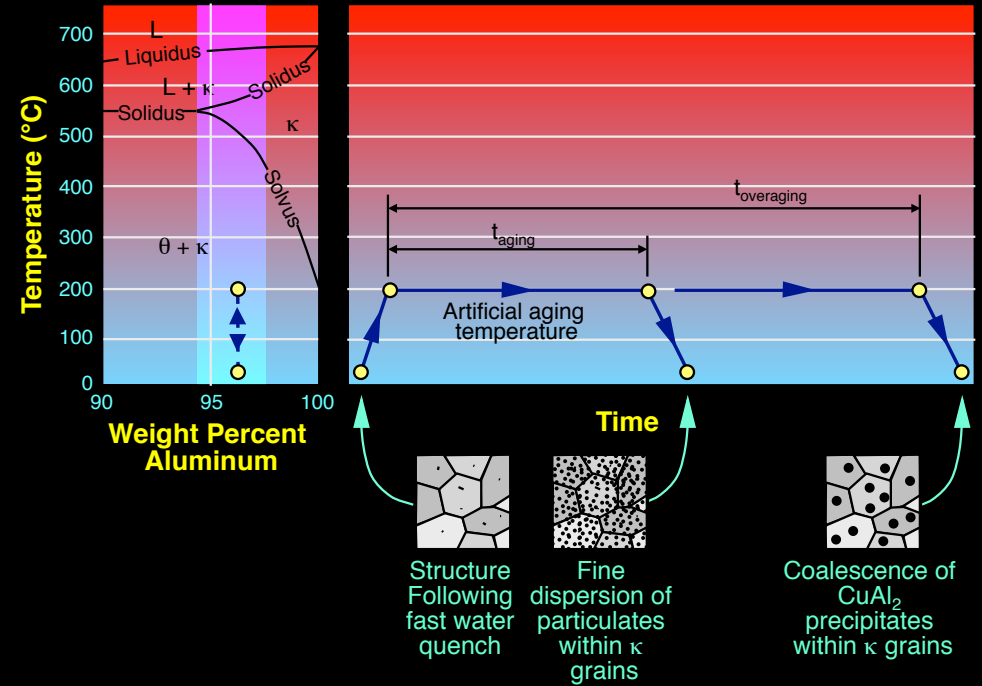
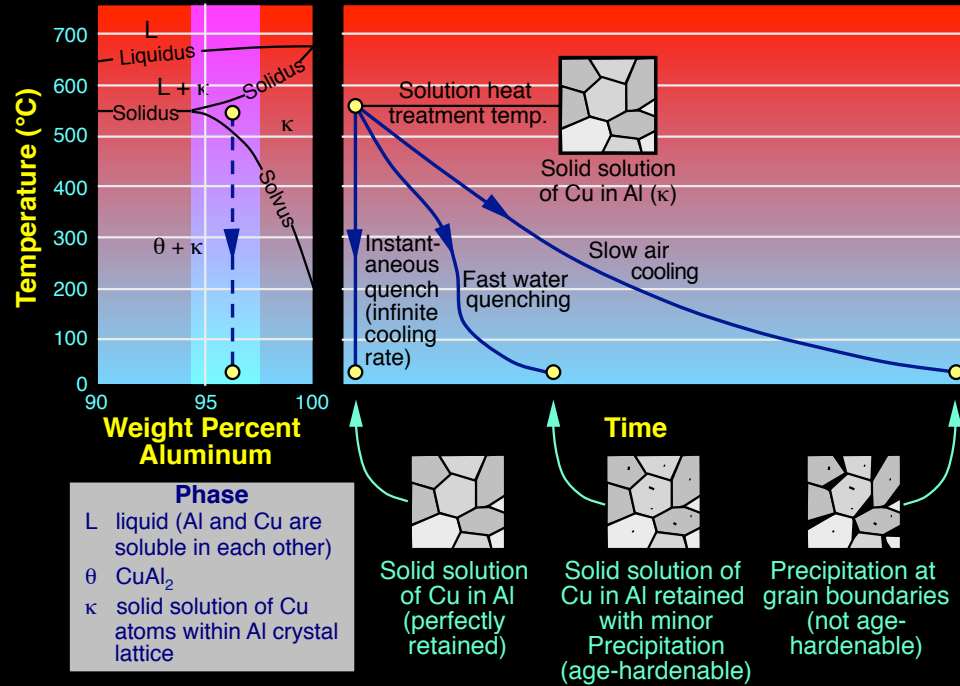
**Poor  
Strength**

### *U.S. Aluminum Industry Profile*

Energy Consumption: 200,000 Trillion BTU  
23.0 B lbs Annual Production Volume  
Generates \$65 Billion (plus \$87 B indirect)  
155,000 Employees (plus 517,000 Indirect)

Source: Aluminum Assoc.

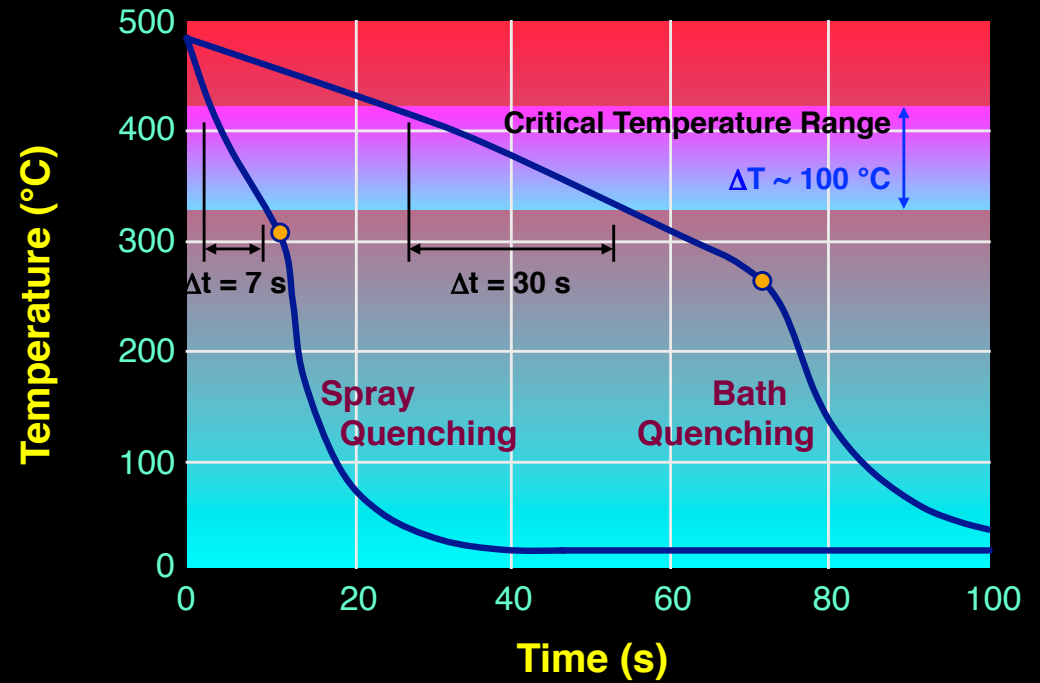
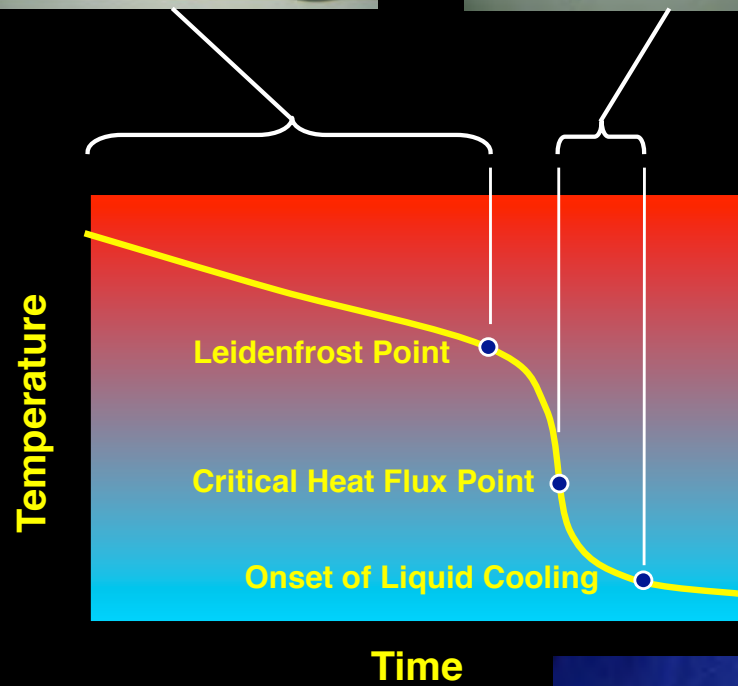
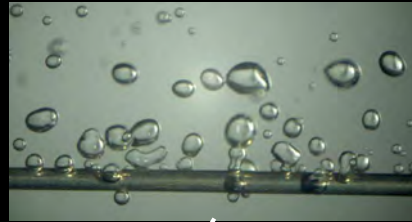
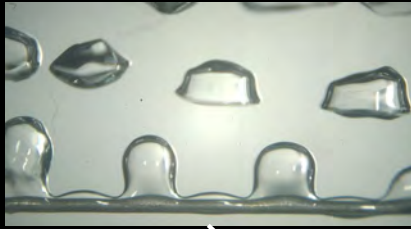
## 4. Microstructural Development of Aluminum Alloy during Quenching Phase of Heat Treating



Heat treating of Al-Cu (4.4 wt %) alloy  
(Hall & Mudawar, 1996)



## 4. Cooling Curve for Liquid Bath Quenching versus Spray Quenching

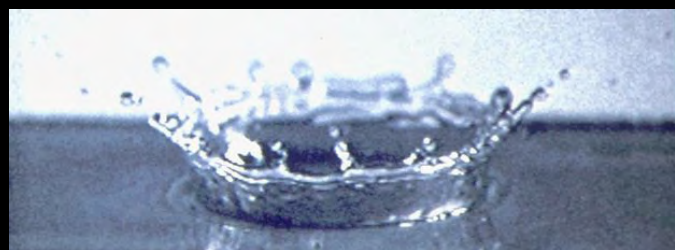


**Distortion induced by Poor Quenching**

## 4. Fundamental Physics of Spray Cooling



Mudawar & Valentine (1989)



### Surface Analysis Facility

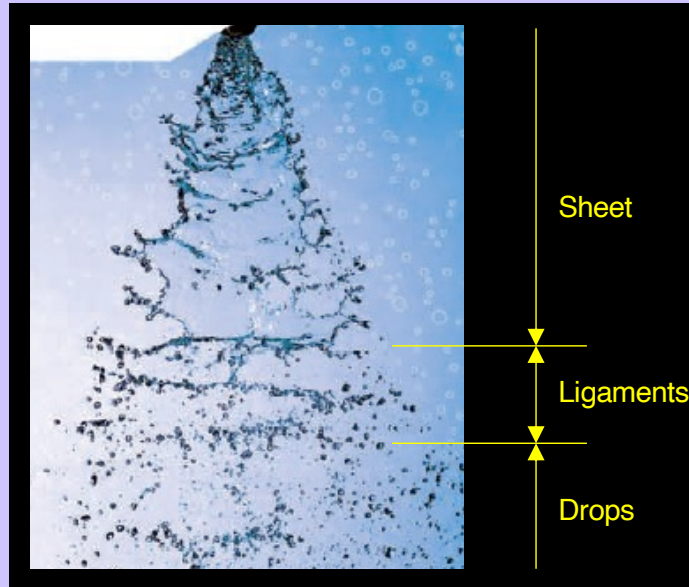
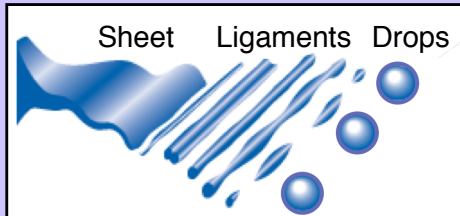


**Krüss  
Tensiometer**

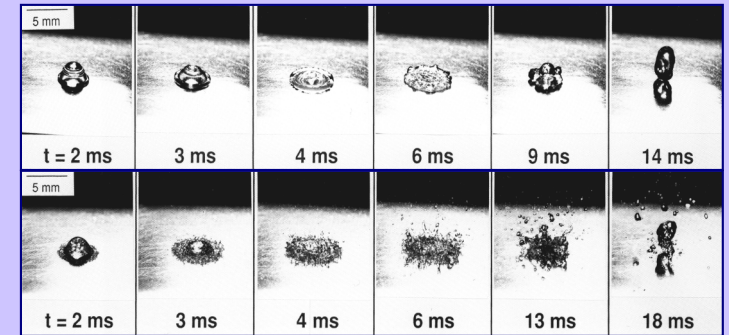


## 4. Fundamental Physics of Spray Cooling

### Nozzle Region: Droplet Breakup Physics



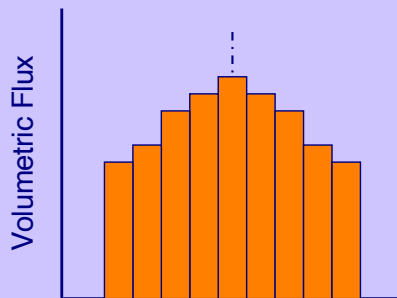
### Surface Region: Local Impact Physics



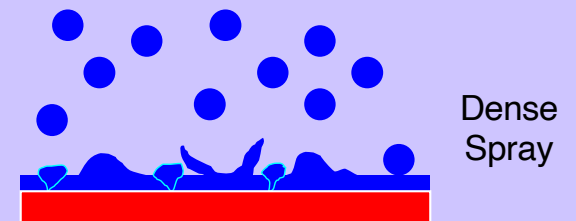
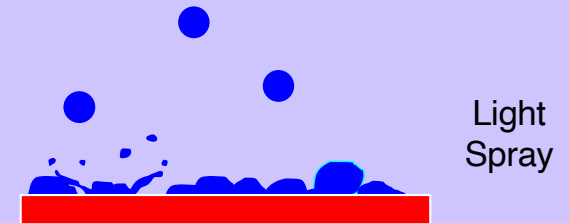
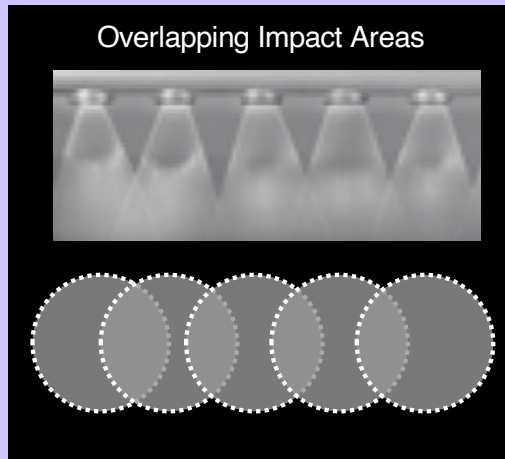
Individual Droplet Impact:

- Weber number
- Surface temperature

### Surface Region: Global Impact Physics



Overlapping Impact Areas

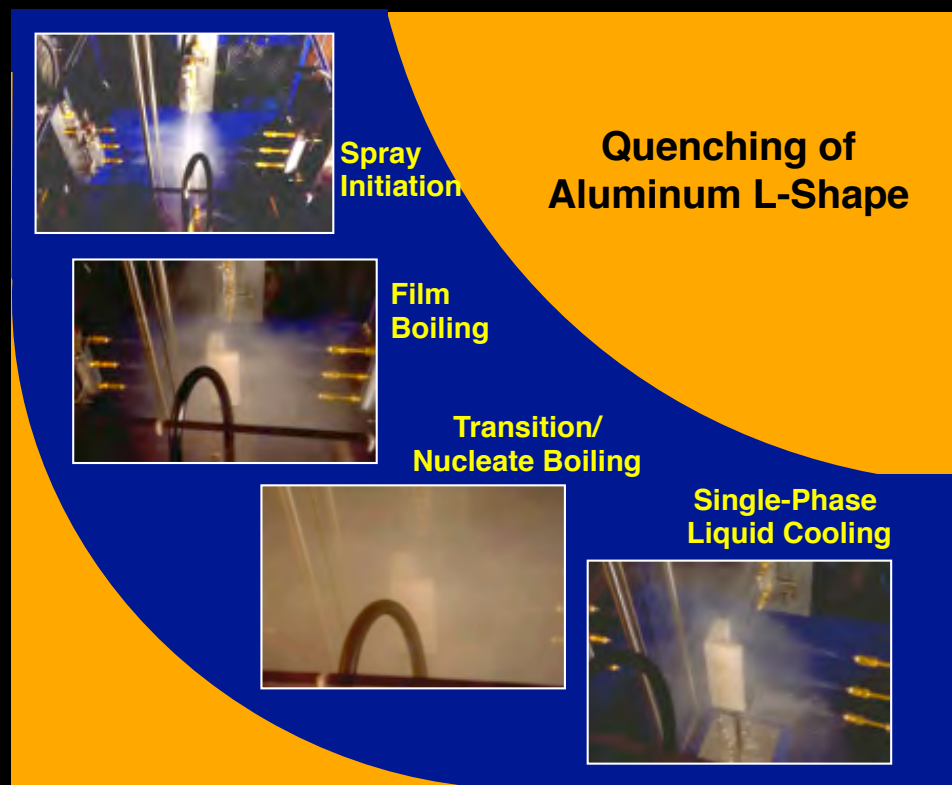


Liquid Film Formation, Evaporation & Boiling:

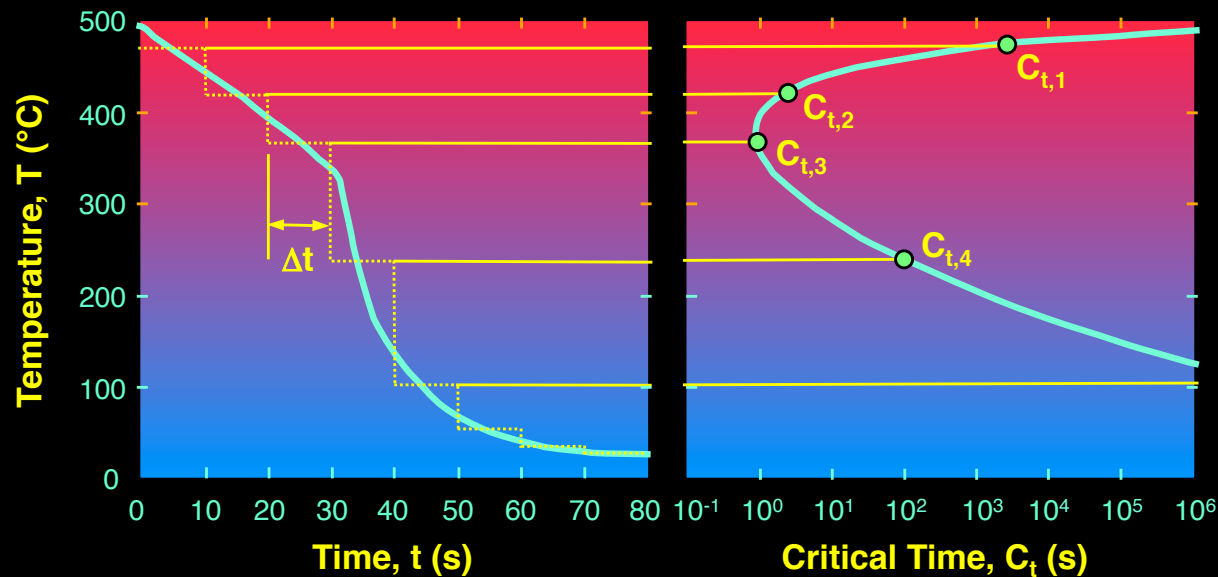
- Volumetric flux



## 4. Purdue Quenching Facility



## 4. Determination of Quench Factor and Hardness



**Quench factor:**

$$\tau = \int_{t_1}^{t_2} \frac{dt}{C_t} \cong \frac{\Delta t_1}{C_{t,1}} + \frac{\Delta t_2}{C_{t,2}} + \dots + \frac{\Delta t_n}{C_{t,n}} = \sum_{i=1}^n \frac{\Delta t_i}{C_{t,i}}$$

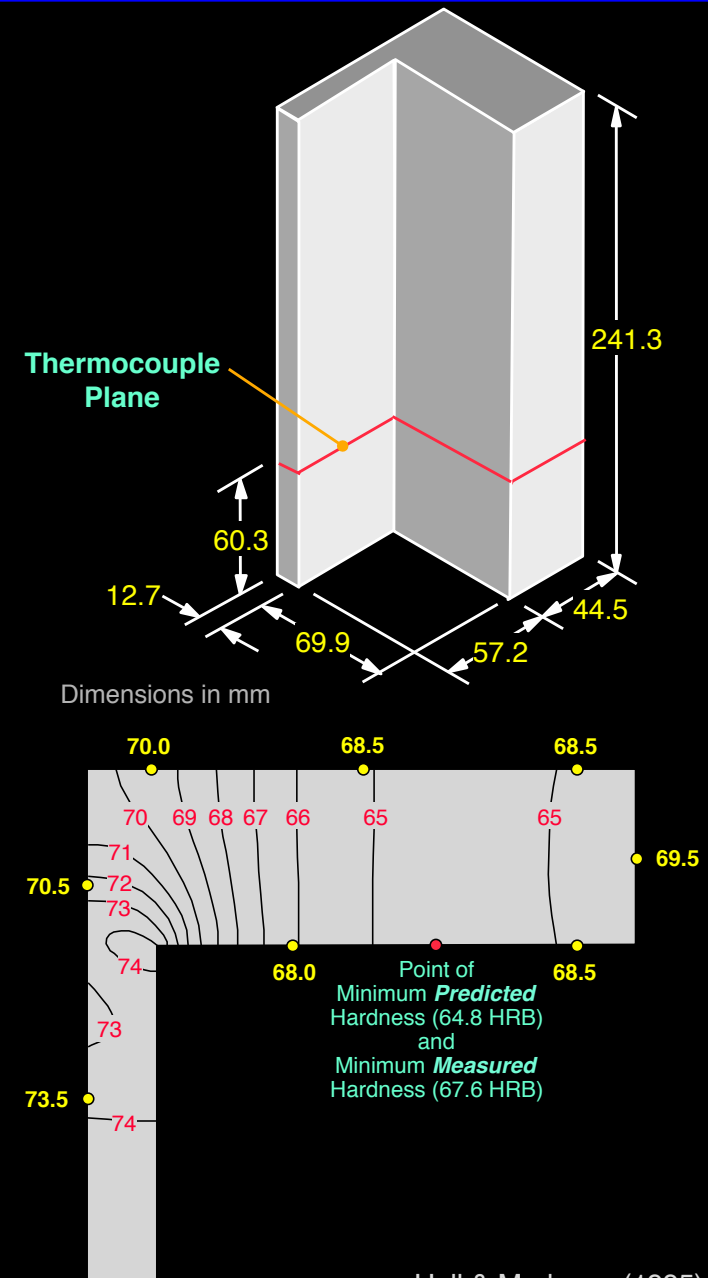
where  $C_t = -k_1 k_2 \exp\left(\frac{k_3 k_4^2}{RT(k_4 - T)^2}\right) \exp\left(\frac{k_5}{RT}\right)$ , T in Kelvins

**Hardness:**

$$H = H_{\min} + (H_{\max} - H_{\min}) \exp(k_1 \tau)$$

**H** ○ Rockwell B hardness (average of three *measurements* near thermocouple plane rounded to nearest 0.5 HRB, all three measurements were within ±1 HRB)

— **H** — Rockwell B hardness contours *predicted* using quench factor technique with *predicted* temperature-time history)



Hall & Mudawar (1995)