

High Heat Flux Interactions and Tritium Removal from Plasma Facing Components by a Scanning Laser*

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Abstract— A new technique for studying high heat flux interactions with plasma facing components is presented. The beam from a continuous wave 300 W neodymium laser was focussed to 80 W/mm² and scanned at high speed over the surface of carbon tiles. These tiles were previously used in the TFTR inner limiter and have a surface layer of amorphous hydrogenated carbon that was codeposited during plasma operations. Laser scanning released up to 84% of the codeposited tritium. The temperature rise of the codeposit on the tiles was significantly higher than that of the manufactured material. In one experiment, the codeposit surface temperature rose to 1,770 °C while for the same conditions, the manufactured surface increased to only 1,080 °C. The peak temperature did not follow the usual square-root dependence on heat pulse duration. Durations of order 100 ms resulted in brittle destruction and material loss from the surface, while a duration of ≈10 ms showed minimal change. A digital microscope imaged the codeposit before, during and after the interaction with the laser and revealed hot spots on a 100-micron scale. These results will be compared to analytic modeling and are relevant to the response of plasma facing components to disruptions and vertical displacement events (VDEs) in next-step magnetic fusion devices.

I. INTRODUCTION

Designing a robust interface between a burning plasma and the material world remains a major issue for magnetic fusion[1]. The survival of plasma facing materials under ELMs, disruptions and vertical displacement events (VDEs), and the control of the tritium inventory are related challenges that must be met for magnetic fusion to achieve its promise as an attractive, environmentally acceptable energy source[2]. Carbon based materials have superior thermomechanical properties and do not melt (they sublime), however they cause high levels of tritium retention by codeposition with eroded carbon that would severely curtail plasma operations[3,4] in a next-step device with carbon plasma facing components.

A novel technique to remove tritium from plasma facing components was proposed at the 17th IEEE/NPSS Symposium for Fusion Engineering[5]. This takes advantage of advances in laser technology to rapidly heat co-deposited layers with a high-power scanning laser beam and thermally desorb tritium. Recent experimental tests have successfully removed 84% of the codeposited tritium on TFTR tiles by this method[6]. The technique is attractive for tritium removal in a next-step DT device since it avoids the use of oxidation, the associated deconditioning of the plasma facing surfaces and expense of processing large quantities of tritium oxide[7].

Under disruption conditions, next-step devices will experience heat loads more than two orders-of-magnitude higher than in present machines [2]. Comprehensive modeling codes[8] have been used simulate the conditions from the transport of the disrupting core plasma to the scrape-off-layer,

the subsequent generation of a vapor shield at the divertor, and the reduced divertor plate lifetime due to melt layer loss and brittle destruction[9]. The thermo-mechanical response of graphite and carbon fiber composite (CFC) to very high heat flux includes sublimation, heating and explosion of gases trapped in the pores, and thermal stresses and fatigue. Pulsed lasers [10], electron beams [10] and plasma guns [11] have been used to reproduce some aspects of the interaction.

Surfaces of plasma facing components in tokamaks undergo ion bombardment and may accumulate deposits. The surface of tungsten is known to blister under high fluence ion bombardment[12, 13] and the surface thermal conductivity will change on a microscopic scale. Although designed for tritium removal, the present experiment offers an opportunity to study in microscopic detail the thermomechanical response of tokamak generated codeposits to high heat fluxes and pulse durations 10 - 300 ms, in the range of disruptions and VDE's. The response is measured without the complications of vapor shielding, which may attenuate the thermal flux in a tokamak. The experimental setup is the same as used for the laser detritiation experiments fully reported in [6]. In the present paper we focus on the material response to the laser heat flux.

II. EXPERIMENTAL SETUP

A 300 W continuous wave Nd laser beam was typically focused to a 2 mm spot of 80 W/mm² intensity and scanned across a tile sample (Fig.1). The tiles were retrieved from TFTR after the tritium campaign and have tritiated codeposited layers of the order of 50 microns thick. The manufactured tile material was graphite (Union Carbide POCO AXF-5Q) and carbon fiber composite (Fiber Materials Inc. 4D coarse weave). The local duration of the heat pulse was varied from 10 ms to 300 ms by changing the scan speed. The change in surface temperature was recorded by an infra red pyrometer with 0.7 mm spatial and 0.3 ms temporal resolution. A digital microscope recorded the surface appearance before, during and after the laser exposure.

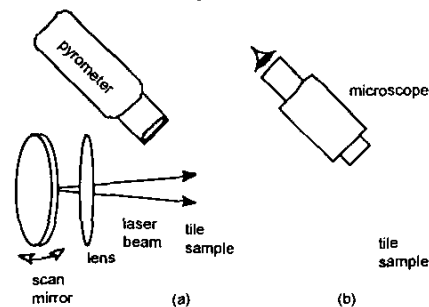


Fig. 1 Experimental setup for (a) laser scan and (b) microscopy.

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III. TEMPERATURE RESPONSE

A brilliant incandescent spot was generated by the focussed laser beam. In some cases the surface temperature rose to peak values over 2,300 °C (the pyrometer range is 500 - 2,300 °C). The laser focus was scanned in a raster pattern and Fig. 2 shows the temperature response as a given position on the surface experienced first the fringe, then the center and then again the fringe of the laser spot. After irradiating the codeposited side, the sample was rotated to expose the manufactured graphite on the cut side with the same laser focal intensity and scan speed. As can be seen in Fig. 2, the maximum temperature rise of the codeposit, 1,770 °C, was much higher than that of the manufactured tile material, 1,080 °C. In a separate experiment, a CFC sample that exhibited both deposition and erosion areas, showed a peak temperature 1,841 °C on the deposition area compared to 1,181 °C on erosion area. The higher temperature is advantageous for tritium release. When the scan speed and hence local heating duration was varied, the peak temperature did not follow the expected square-root dependence on duration. Fig. 3 shows that the peak temperature increased by only 20% (from 1,682 to 2,021 °C) after a 5x decrease in scan speed and corresponding increase in heating duration. Clearly additional heat absorption or surface removal mechanisms become active at longer durations. One can also see precursors in Fig. 3, most likely due to "burn-up" of loosely attached flakes and dust.

IV. MICROSCOPY OF THE SURFACE

A digital microscope recorded the surface appearance before and after the laser irradiation. Before laser irradiation, granulation and irregularities on the codeposited surface were readily apparent. After irradiation at high scan speeds (1,000 mm/s, surface temperature above 500 °C for 10 ms) there is a slight color change but the codeposit appears undisturbed even though the temperature reached 1,770 °C and 18 mCi of tritium was released (Fig. 4(a)) [6]. However, at slow scan speeds substantial surface damage and material loss from the surface occurs. Figs 4(c)-4(d) shows the results of a 25 mm/s scan in which the laser irradiated a sequence of bands on the tile surface at increasing laser power.

Fig. 5 shows a microscope image of the surface *during* laser irradiation obtained by using the digital microscope in a video mode at 30 frames/s with x100,000 attenuation by a neutral density filter. The surface brightness shows discrete "micro hotspots" indicating that the temperature recorded in the 0.7 mm pyrometer viewing spot (Figs 2 & 3) averages over these features. It is clear that the surface granulation seen in Fig. 4 modulates the interaction. Granules that are poorly thermally connected to the underlying material will experience much higher temperatures than average. The range from threshold to saturated of an 8-bit CCD signal corresponds to a factor x2 to x3 temperature change so these are strong inhomogeneities.

Fig. 5 shows clearly that microscopic structures are important in the temperature response of tokamak codeposits to plasma heat flux. An estimate of the temperature response using coefficients for bulk manufactured materials to see, for example, if the temperature rise remains below the threshold for radiation induced sublimation[14], may be unrealistic.

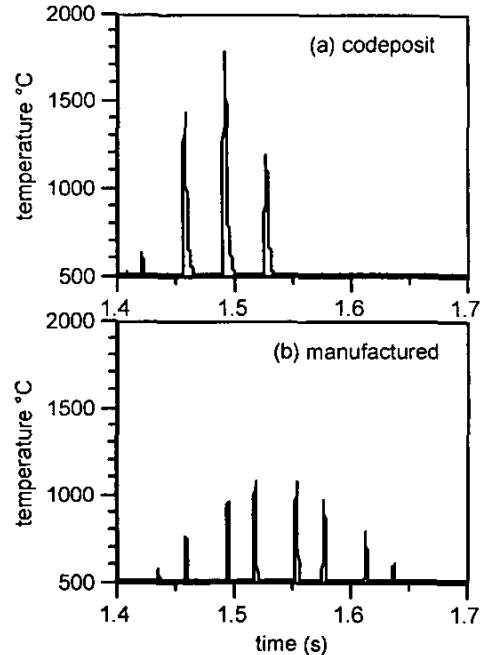


Fig. 2 Comparison of temperature response of (a) codeposit and (b) manufactured surface of graphite sample KC22 6E, both at a laser intensity of 80 W/mm², scan speed 1000 mm/s.

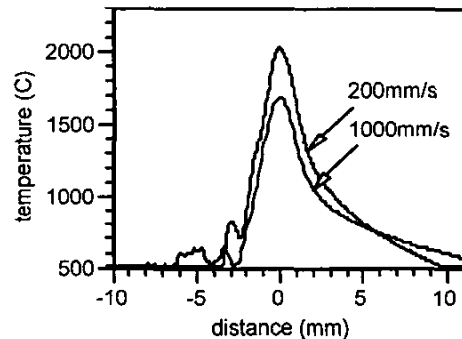


Fig. 3 Temperature response of CFC sample KC17 1C at scan speeds 200 and 1,000 mm/s. Laser intensity 80 W/mm² in both cases. The horizontal axis represents the distance between the pyrometer viewing spot and the scanning laser spot. Note that with 5x longer heat pulse the peak temperature only increases 20%.

V. HEAT TRANSFER

We first consider thermal conduction into the tile sample. Equation (1) describes the temperature, θ , vs. depth, x , of a semi-infinite, homogeneous solid at zero initial temperature under a constant heat flux at $x = 0$ beginning at $t = 0$ [15]:

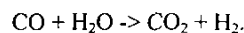
$$\theta = \frac{2F}{K} \left\{ \left(\frac{\kappa t}{\pi} \right)^{1/2} \exp\left(\frac{-x^2}{4\kappa t} \right) - \frac{x}{2} \operatorname{erfc} \frac{x}{2\sqrt{\kappa t}} \right\} \quad (1)$$

Here "erfc" is the complementary error function, K is the thermal conductivity, $\kappa = K / \rho c$ the thermal diffusivity. Note that in practice the carbon based materials are not homogeneous and the conductivity varies on a microscopic scale, is temperature dependent and the local laser intensity is

not constant. Non the less Eqn. 1 predicts that a 10 ms, 80 W/mm² heat flux will cause the surface of a graphite tile to rise from 500 to 1,104 °C, similar to the measured temperature rise of 500 to 1,080 °C. The coefficients used are density: 2.1 g/cm³[16]; thermal conductivity: 0.56 W/cm-K [17]; and heat capacity: 1.9 J/g-K[17]; the latter two at 1,000 °C. Codeposits are known to have a more porous structure[18] and we make an ad hoc reduction by 50% of the density and thermal conductivity. For this case, Eqn. 1 predicts a temperature rise from 500 to 1,708 °C for the same heat pulse, similar to the measured temperature rise of the codeposit of 500 to 1,770 °C. Fig. 6 shows the temperature response for the two cases. It is clear that predictions of surface temperatures of codeposited layers using thermal coefficients measured for manufactured materials may greatly underestimate the temperature rise of a codeposited layer.

The above analysis is approximate and, as evidenced by Fig. 5, the temperature will be strongly modulated by micro inhomogeneities. However it indicates that for the laser conditions found optimal for detritiation (80 W/mm², 10 ms duration) thermal conduction will heat a 50-micron thick codeposited layer to ≈1,500 °C. Significantly higher intensities will cause the surface temperature to exceed the sublimation threshold (2,600 °C) and longer durations will cause the heat to penetrate past the codeposited layer, both causing surface ablation and pore gas explosion that limit the peak temperature.

Another potential energy sink is the endothermic steam reaction[19]



Oxides at an atomic concentration 20-50% have been detected in the codeposited surfaces of tiles retrieved from TFTR, presumably from moisture absorbed while in storage[20]. This reaction of interest as it is a potential source of hydrogen in a loss of coolant accident in a next-step device. More work is needed to ascertain its role in the present experiment.

VI. EROSION MECHANISMS OF CARBON-BASED-MATERIALS

Nonmelting materials such as graphite and carbon based materials (CBMs) have shown large erosion losses in disruption simulation facilities that use electron beams [21], laser [22], plasma guns and other high-power devices [23]. The losses significantly exceed that from surface vaporization. Models have been developed and implemented in the HEIGHTS package[26] to evaluate erosion behavior and the lifetime of carbon based plasma facing and nearby components due to brittle destruction during plasma instabilities [24].

The macroscopic erosion of CBMs depends on the net power flux to the surface, exposure time, and threshold energy required for brittle destruction. This threshold is currently estimated from disruption-simulation experiments and is critical in determining the net erosion rate of CBMs. For MPG-9 graphite it is estimated to be ≈10 kJ/g, or 20 kJ/cm³ [23]. Eqn. 1, with density and conductivity coefficients reduced to 50% of graphite values, predicts that a heat flux of 80 W/mm² for 10 ms or 0.8 J/mm² would result in a surface

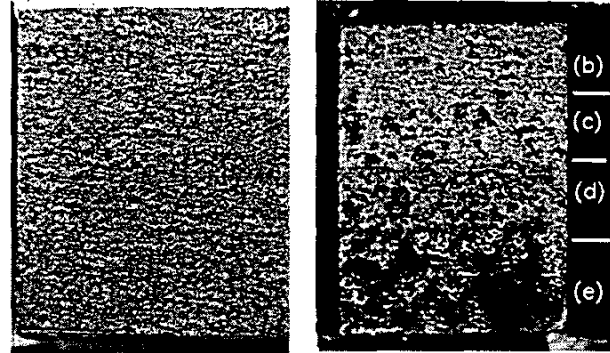


Fig. 4 (a) Codeposit on graphite sample KC22-6E after irradiation at 80 W/mm², scan speed 1000 mm/s. This resulted in a 10 ms temperature excursion to 1,770 °C and the release of 18 mCi of tritium; (b) original codeposit on CFC sample KC17-3C, and after 25 mm/s irradiation at 9 W/mm² (c) 29 W/mm² (d), 77 W/mm²(e). For (e) the temperature excursion > 500 °C lasted 222 ms and peaked at 1,925 °C. The horizontal width of the samples are 24 mm and 15 mm respectively.



Fig. 5 Microscope image of laser interaction with sample KC17-2B at 80 W/mm², 50 mm/s. The scale bar indicates 1 mm.

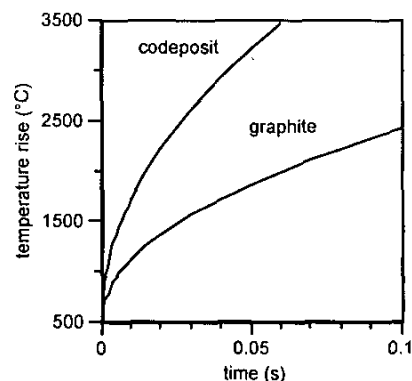


Fig. 6 Temperature rise of POCO AXF-5Q graphite and codeposit for 80 W/mm² (from Eqn. 1 -see text).

temperature of 1,000 °C at 300 microns below the surface. The 20 kJ/cm³ brittle destruction threshold corresponds to 6 J/mm² if absorbed in a codeposited layer 300 microns thick. Thus, the experimental conditions for a 1000 mm/s scan appear to be below the brittle destruction threshold, consistent with the minimal changes seen in Fig. 4(a). For a 100 ms duration heat pulse the surface temperature exceeds the sublimation threshold (Fig. 6) implying the onset of sublimation and pore gas explosion which is consistent with the surface damage observed in Fig. 4(e).

VII. CONCLUSIONS

The thermal response of deposition areas on graphite and CFC plasma facing components is significantly different to that of the manufactured material. The temperature rise is much higher and varies strongly on a 100-micron spatial scale. 1D heat flux calculations for plasma facing components typically use global values of thermal conductivity from manufactured materials to predict the surface temperature under a given heat load. A large influx of carbon during TFTR plasma operations was attributed to radiation induced sublimation, when temperatures of the plasma facing graphite limiter surface exceeded a threshold value of 1,650 °C[25] so it is important that surface temperatures stay below this value during projected heat loads. The above results illustrate that the thermal response of codeposits is markedly different to the manufactured tile materials and has large microscopic spatial variations. Reliable predictions for the thermal behavior of deposition areas need to be based on experiments with tokamak-generated codeposits and relevant pulse durations. This issue also complicates measurements of heat flux at plasma facing surfaces by IR thermography.

Erosion losses from brittle destruction are much higher than that predicted from pure surface vaporization [26]. This could have serious implications during loss of plasma confinement and disruptions in future tokamak devices. Macroscopic erosion of CBMs due to the various mechanisms of brittle destruction depends strongly on the type of carbon material. The erosion lifetime of CBM components could be significantly shorter than the hundreds of disruptions currently assumed for ITER-like devices. More experimental data and additional detailed modeling are urgently needed.

Tungsten avoids tritium codeposition issues but melt layer loss during off normal events may limit the operational lifetime[1]. Blistering has been observed after high ion fluence ion bombardment[27,28] and this is expected to also change the surface thermal conductivity on microscopic scale. Future work will study W samples pre-exposed to high fluence ion bombardment.

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