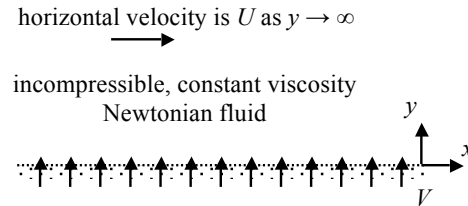


Consider steady flow at horizontal velocity U (at $y \rightarrow \infty$) past an infinitely long and wide plate. The plate is porous and there is uniform flow normal to the surface at a constant velocity, V . Assume there are no pressure gradients and that gravity is negligible.



- Determine the y -velocity at all points in the flow field.
- Determine the x -velocity at all points in the flow field.
- What restriction is there on the velocity V ?
- Quantify how far into the flow the wall effects are felt. Clearly indicate what criterion you are using.

SOLUTION:

Make the following assumptions about the flow:

- | | |
|---|---|
| 1. The flow is planar. | $\Rightarrow \frac{\partial}{\partial z}(\dots) = 0, u_z = 0$ |
| 2. The flow is steady. | $\Rightarrow \frac{\partial}{\partial t}(\dots) = 0$ |
| 3. The flow is fully-developed in the x -direction. | $\Rightarrow \frac{\partial u_x}{\partial x} = \frac{\partial u_y}{\partial x} = 0$ |
| 4. Neglect gravity. | $\Rightarrow g_x = g_y = g_z = 0$ |
| 5. No pressure gradients. | $\Rightarrow \frac{\partial p}{\partial x} = \frac{\partial p}{\partial y} = \frac{\partial p}{\partial z} = 0$ |

The continuity equation for an incompressible, planar flow is:

$$\underbrace{\frac{\partial u_x}{\partial x}}_{=0(\#3)} + \frac{\partial u_y}{\partial y} = 0 \Rightarrow \frac{\partial u_y}{\partial y} = 0 \quad (\text{call this condition \#6})$$

Since u_y is not a function of x (#3), z (#1), or y (#6), then $u_y = \text{constant}$. Since the flow at the wall has vertical velocity, V :

$$\boxed{u_y = V} \quad (1)$$

Now examine the x -momentum equation:

$$\rho \left(\underbrace{\frac{\partial u_x}{\partial t}}_{=0(\#2)} + u_x \underbrace{\frac{\partial u_x}{\partial x}}_{=0(\#3)} + \underbrace{u_y}_{=V(\#6)} \frac{\partial u_x}{\partial y} \right) = - \underbrace{\frac{\partial p}{\partial x}}_{=0(\#5)} + \mu \left(\underbrace{\frac{\partial^2 u_x}{\partial x^2}}_{=0(\#3)} + \frac{\partial^2 u_x}{\partial y^2} \right) + \rho \underbrace{g_x}_{=0(\#4)}$$

$$V \frac{du_x}{dy} = \nu \frac{d^2 u_x}{dy^2} \quad (2)$$

where the partial derivatives have been replaced by ordinary derivatives since u_x is not a function of x (#3) or z (#1). Note also that $\nu = \mu/\rho$.

Solving Eqn. (2):

$$\frac{V}{\nu} = \frac{d}{dy} \left(\ln \frac{du_x}{dy} \right)$$

$$\frac{V}{\nu} y + c_1 = \ln \frac{du_x}{dy}$$

$$\exp \left(\frac{V}{\nu} y + c_1 \right) = c_2 \exp \left(\frac{V}{\nu} y \right) = \frac{du_x}{dy}$$

$$u_x = c_3 \exp \left(\frac{V}{\nu} y \right) + c_4 \quad (3)$$

Apply the following boundary conditions:

$$\text{no-slip at } y = 0 \quad \Rightarrow \quad u_x(y = 0) = 0 \quad (4)$$

$$\text{horz. velocity is } U \text{ as } y \rightarrow \infty \quad \Rightarrow \quad u_x(y \rightarrow \infty) = U \quad (5)$$

$$u_x(y = 0) = c_3 + c_4 = 0 \quad (6)$$

$$u_x(y \rightarrow \infty) = c_3 \lim_{y \rightarrow \infty} \left[\exp \left(\frac{V}{\nu} y \right) \right] + c_4 = U \quad (7)$$

Note that in order to have u_x remain finite as $y \rightarrow \infty$, we must have $V < 0$. Hence, Eqn. (7) implies that $c_4 = U$. Substituting and simplifying gives:

$$u_x = U \left[1 - \exp \left(\frac{V}{\nu} y \right) \right] \quad (8)$$

To quantify the distance into the flow that the wall effects are felt, use the 99% boundary layer thickness, δ , *i.e.*:

$$\frac{u_x}{U} = 0.99 = 1 - \exp \left(\frac{V}{\nu} \delta \right)$$

$$\delta = \frac{\nu}{V} \ln 0.01 \quad (9)$$